ELEC-E8101 Digital and Optimal Control Exercise 9 - solution Autumn 2022

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For rotational motion, the sum of torques acting on the system is equal to the product of the moment of inertia *I* and angular acceleration $\ddot{\theta}$. For this system the torques are caused by the external force *F* and the gravity of the end of the pendulum (assume that the stick is massless).

Thus,

$$egin{aligned} &I\ddot{ heta} = \sum M \ &ml^2\ddot{ heta} = -mgl\sin heta + Fl\cos heta \Rightarrow \ &\ddot{ heta} = -rac{g}{l}\sin heta + Frac{1}{ml}\cos heta \end{aligned}$$

Using state variables $x_1= heta, x_2=\dot{ heta}, u=F, \ y= heta=x_1$ gives the state-space form $\dot{x}_1=x_2$

$$egin{aligned} \dot{x}_2 &= -rac{g}{l} \sin x_1 + rac{1}{ml} u \cos x_1 \ y &= x_1 \end{aligned}$$

Linearization of a function f(x) around a point x_0 is equivalent to approximating the function around the point with a first-order Taylor series, given by

$$f(x)pprox f(x_0)+f'(x_0)(x-x_0)$$
 .

The nonlinear functions in the state equations are the trigonometric functions, which only appear in the terms for \dot{x}_2 . Their linearizations around zero are then

$$egin{aligned} \sin x &pprox \sin 0 + \left(rac{\mathrm{d}\sin x}{\mathrm{d}x} \,\Big|\, x=0
ight) imes x \ &pprox 0 + \ \cos 0 \ imes x \ &pprox 0 + \left(rac{\mathrm{d}\cos x}{\mathrm{d}x} \,\Big|\, x=0
ight) imes x \ &pprox 1 + (-\sin 0) imes x \ &pprox 1 \end{aligned}$$

The linearization can then be written

$$egin{aligned} \dot{x}_2 &= -rac{g}{l} \sin x_1 + rac{1}{ml} u \cos x_1 \ &pprox -rac{g}{l} x_1 + rac{1}{ml} u \end{aligned}$$

with all other parts of the model remaining original.

In the standard form this gives

$$egin{aligned} \dot{x}_1\ \dot{x}_2\end{pmatrix} &= Ainom{x_1}{x_2} + Bu = inom{0}{1-rac{g}{l}} 0inom{1}{x_2} + inom{0}{rac{1}{ml}} + inom{0}{rac{1}{ml}} u \ y &= Cinom{x_1}{x_2} = (1 \ 0)inom{x_1}{x_2} \end{aligned}$$

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A state space model

 $\dot{x} = Ax + Bu, \, y = Cx + Du$

can be discretized with ZOH with sample time T_s as

$$x[k+1]=A_dx[k]+B_du[k],\,y[k]=Cx[k]+Du[k]$$

where

$$A_d=e^{AT_s}, B_d=A^{-1}(A_d-I)B_{\perp}$$

The matrices can be calculated using Matlab symbolic toolbox as

syms g l Ts m positive A=[0 1;-g/l 0] B=[0 ; 1/(m*l)] Ad=simplify(expm(A*Ts)) Bd=simplify(inv(A)*(Ad-eye(2))*B)

giving

$$A_d = egin{pmatrix} \cos\left(T_s\sqrt{rac{g}{l}}
ight) & \sqrt{rac{l}{g}}\sin\left(T_s\sqrt{rac{g}{l}}
ight) \ -\sqrt{rac{g}{l}}\sin\left(T_s\sqrt{rac{g}{l}}
ight) & \cos\left(T_s\sqrt{rac{g}{l}}
ight) \end{pmatrix} \ B_d = egin{pmatrix} \left(\left(1-\cos\left(T_s\sqrt{rac{g}{l}}
ight)
ight)/(mg) \ \sin\left(T_s\sqrt{rac{g}{l}}
ight)
ight)/(m\sqrt{gl}) \end{pmatrix} \end{pmatrix}$$

Simplifying this by substituting

$$\omega = \sqrt{rac{g}{l}}$$

gives

$$egin{aligned} A_d &= egin{pmatrix} \cos{(T_s\omega)} & \omega^{-1}\sin{(T_s\omega)} \ -\omega\sin{(T_s\omega)} & \cos{(T_s\omega)} \end{pmatrix} \ B_d &= egin{pmatrix} (1-\cos{(T_s\omega)})/(mg) \ \sin{(T_s\omega)}/(ml\omega) \end{pmatrix} \end{aligned}$$

The quantity ω is the natural frequency of the pendulum.

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Assume $g = 9.81, \, l = 9.81, \, m = 1/9.81$. Thus $\omega = 1$.

The discretized system is stable if the eigenvalues of A_d are in the unit circle. Finding them using Matlab symbolic toolbox

eig(Ad)

gives

$$z=\cos{(T_s\omega)}\pm j\sin{(T_s\omega)}$$

for which |z| = 1. The system (without a controller) is thus marginally stable (oscillation continues forever). 5 *

Simulator model



Matlab code for the non-linear plant

```
function dxdt = fcn(x,u)
theta=x(1);
dtheta=x(2);
F=u;
g=9.81;
1=9.81;
m=1/9.81;
dxdt=[dtheta;...
    -g/l*sin(theta)+1/(m*1)*cos(theta)*F];
```

and its linearized version

function dxdt = fcn(x,u)
theta=x(1);
dtheta=x(2);
F=u;
g=9.81;
l=9.81;
m=1/9.81;
dxdt=[dtheta;...
 -g/l*theta+1/(m*1)*F];



Using the PID parameters given, the response for $heta_{t\,\mathrm{arg}\,et}=0.4$ is shown below.

Looking at the figure, both the true non-linear system and its linear approximation behave in a similar fashion, with small differences.



Now the response for $heta_{t\,\mathrm{arg}\,et}=1.5$ is plotted below.

The linearized system behaves similarly to the earlier response, as expected for a linear system, where the set-point does not affect the system performance characteristics. However, the true

non-linear system behaves in a very different way—it oscillates heavily and its steady state error appears to remain high. Also, unlike a linear system, the frequency of oscillation varies over time.

This experiment demonstrates a few important points: First, that the linearization is reasonably accurate only around the point in which it was performed. Thus, analysis and design of control based on a linearized model is valid only when the linearization error is small. Despite the shortcomings, linearization is an important tool for analyzing and designing controllers also for non-linear systems.