


A?

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E4230

Microwave EO Instrumentation

A satellite in orbit over Earth, emitting a beam of light towards the ground. The satellite is a rectangular box with various instruments and antennas. The Earth's surface is visible below, showing green land and blue oceans. The satellite is positioned in the upper right quadrant of the image, with a beam of light extending from it towards the bottom left.

(5 cr)

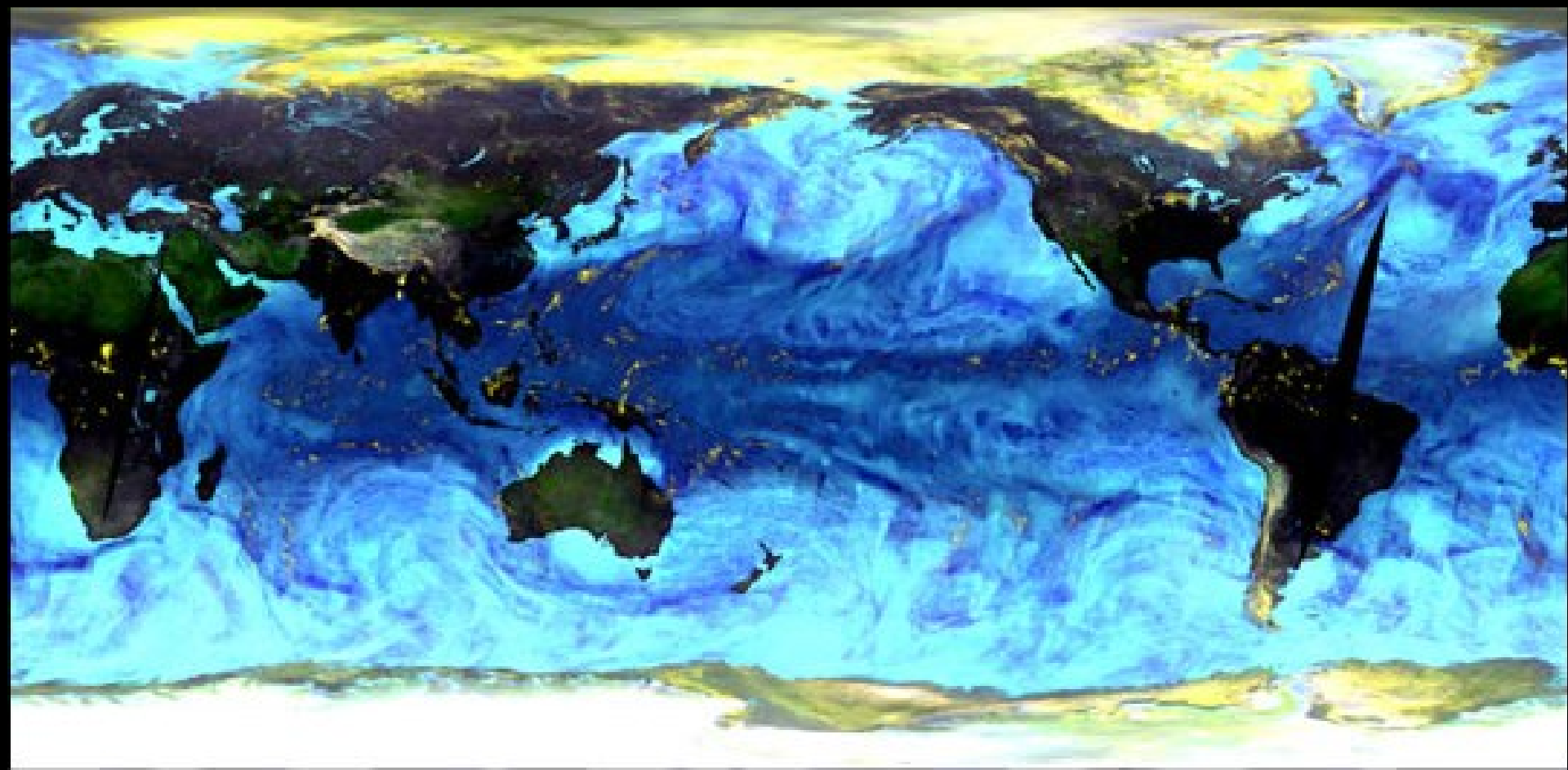
Jaan Praks

Aalto University

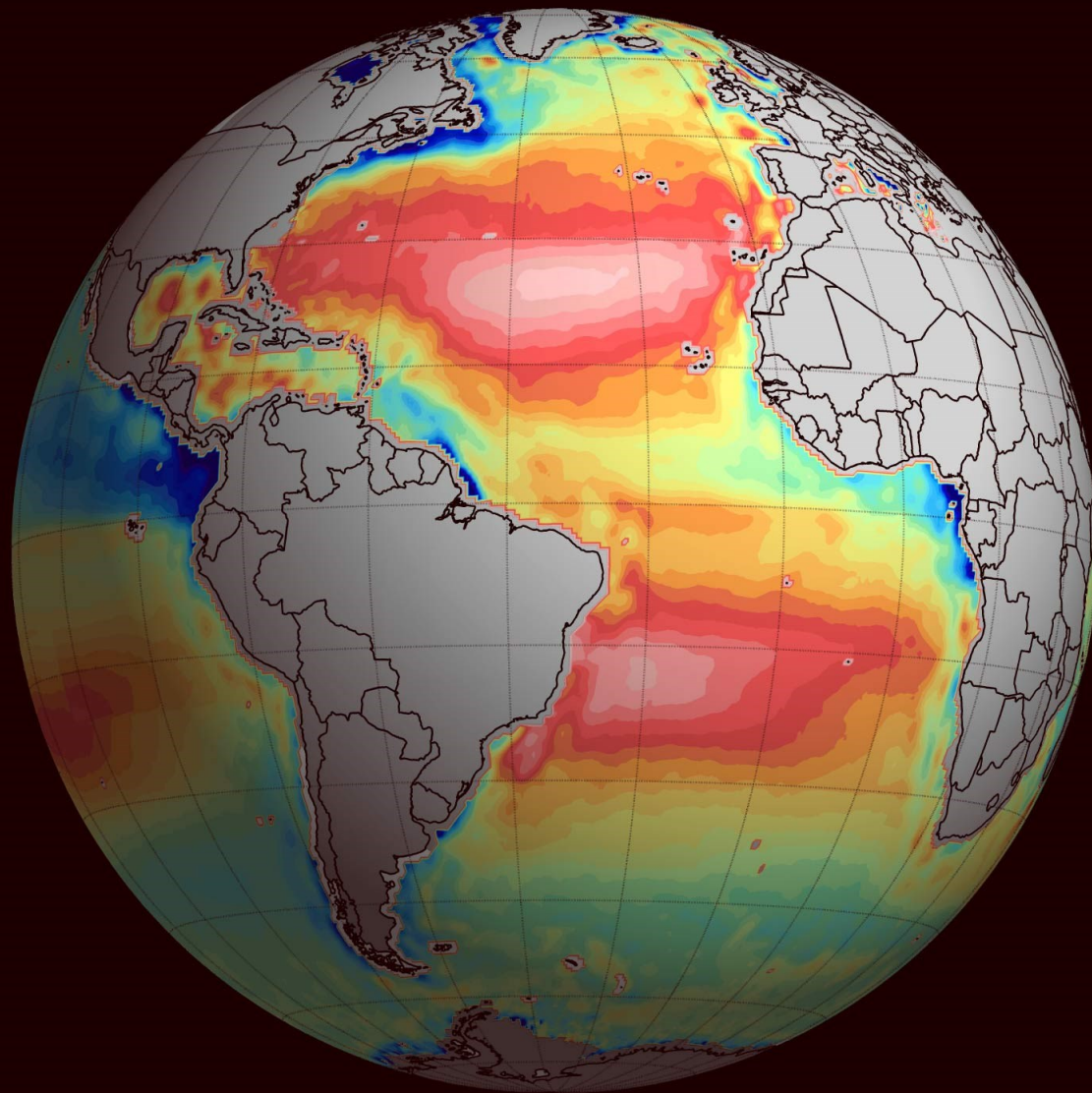


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What is common for next images?



Brightness Temperature



$\mu\text{mol.kg}^{-1}$

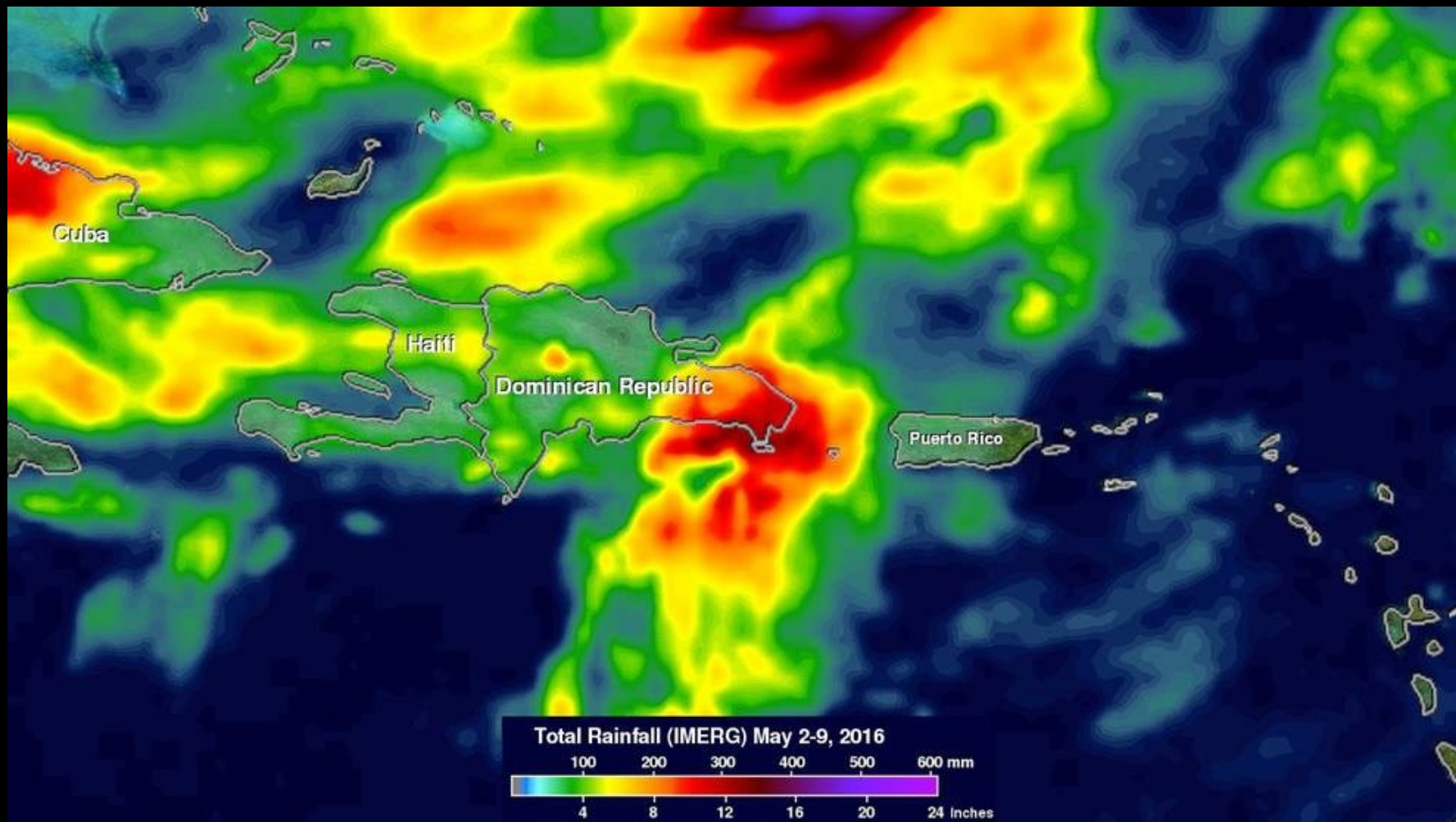
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2400

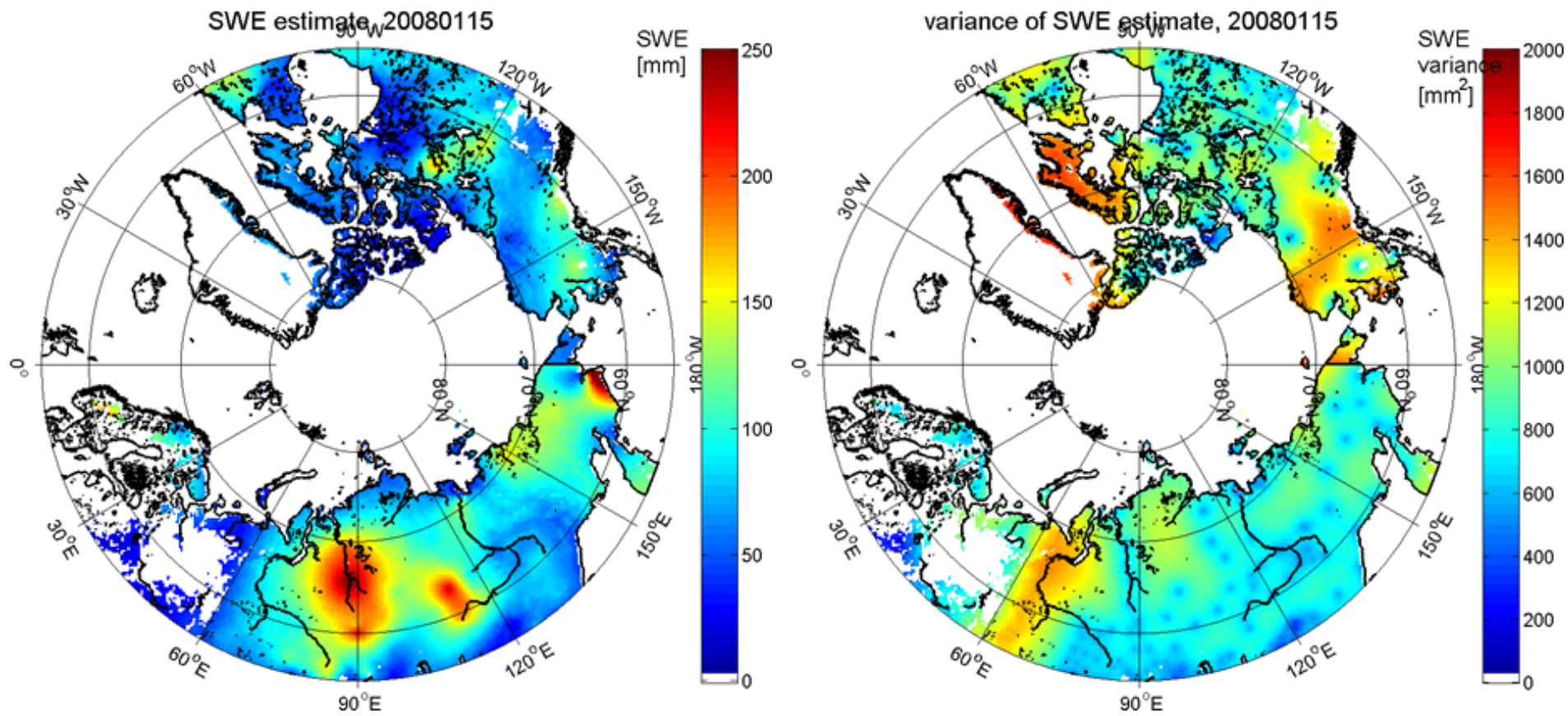
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2200

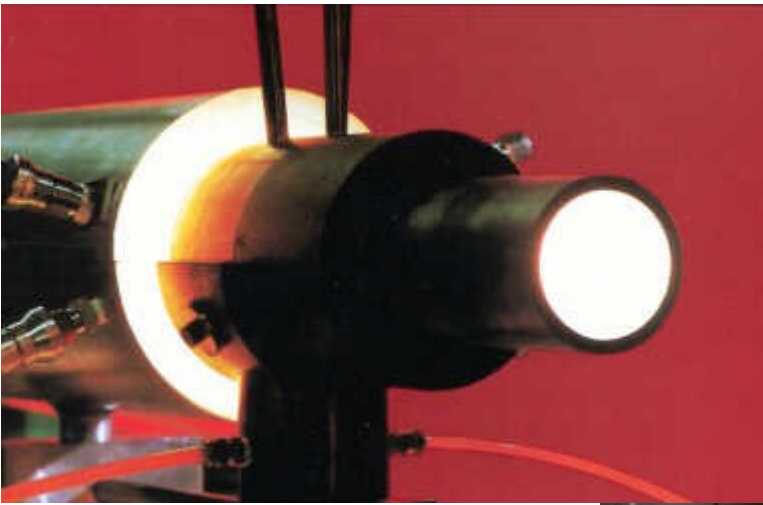
2100



Snow water equivalent and the estimation uncertainty for 15 January 2008



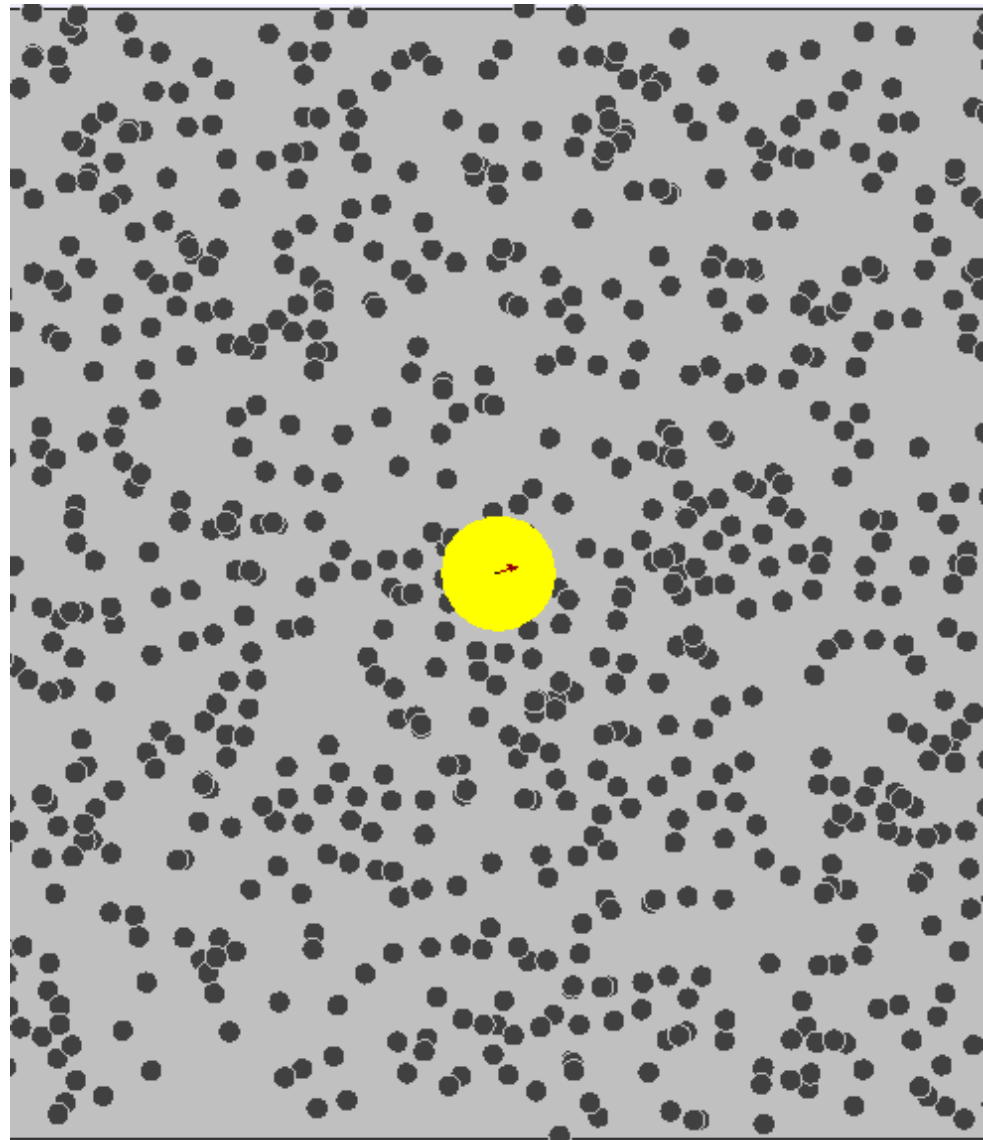
Blackbody



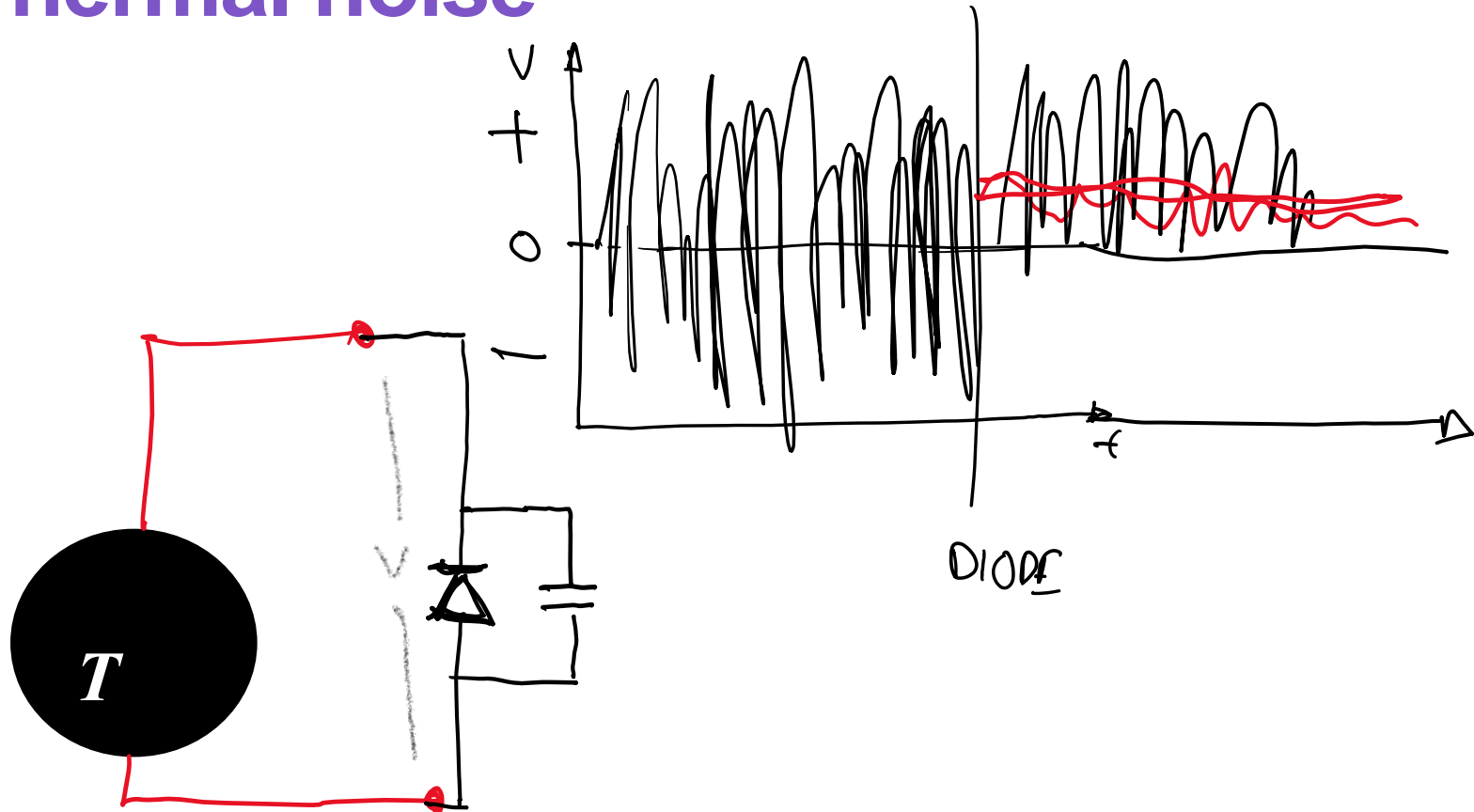


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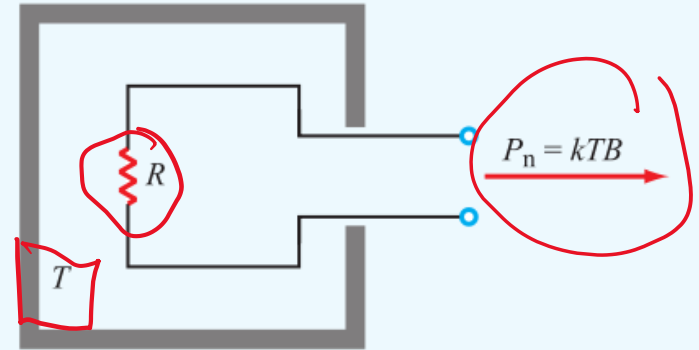
Measuring thermal noise



Thermal noise

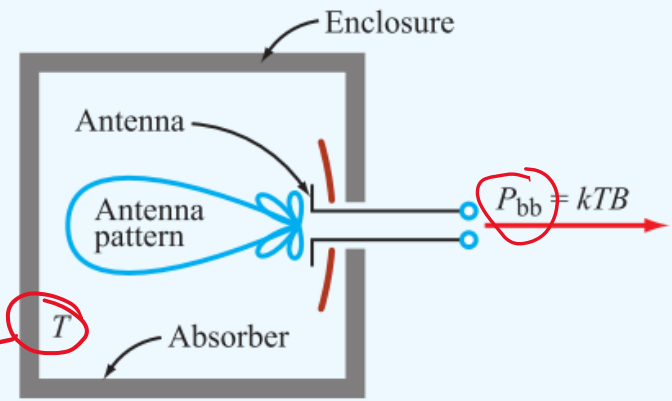


Power and temperature equivalence

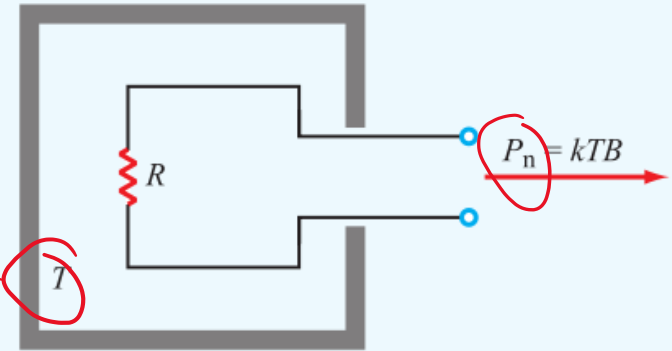


(b) Resistor at temperature T

Power and temperature equivalence -- also for system with an antenna



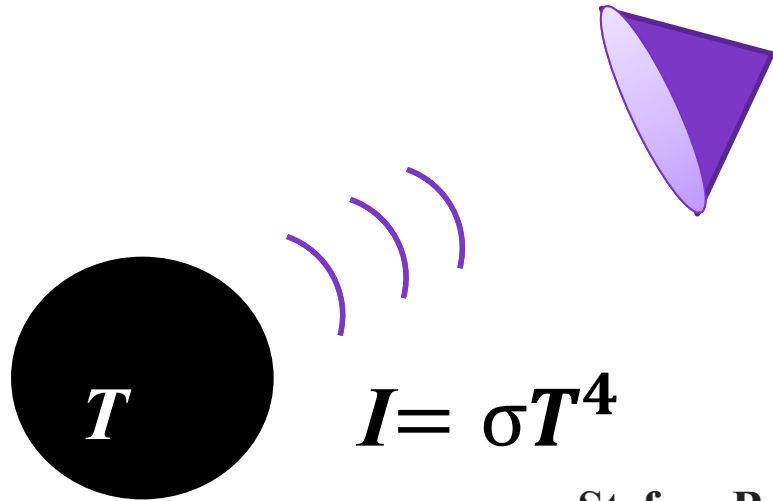
(a) Antenna inside a blackbody enclosure



(b) Resistor at temperature T

Figure 6-5: The power delivered by (a) an antenna placed inside a blackbody enclosure of temperature T is equal to the power delivered by (b) a resistor maintained at the same temperature.

Measuring the radiation

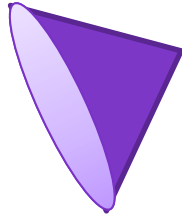
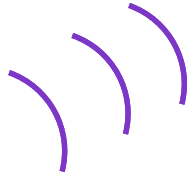
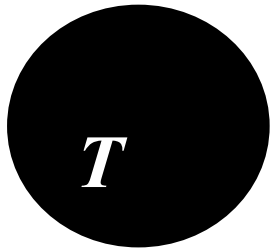


Stefan–Boltzmann law

We add an antenna:

Radiated energy is directly related to temperature T^4

$$\sigma = \frac{2\pi^5 k_B^4}{15h^3 c^2}$$



$$I_f = \frac{2hf^3}{c^2} \frac{1}{e^{\frac{hf}{kT}} - 1}$$

Planck's law

However, we can only measure a certain frequency with antenna.

Radiated energy as a function of frequency.

h - Planck constant

k - Boltzmann constant



Planck's law

$$I_f = \frac{2hf^3}{c^2} \frac{1}{e^{\frac{hf}{kT}} - 1}$$

A **blackbody** is defined as an idealized, perfectly opaque material that absorbs all the incident radiation at all frequencies, reflecting none.

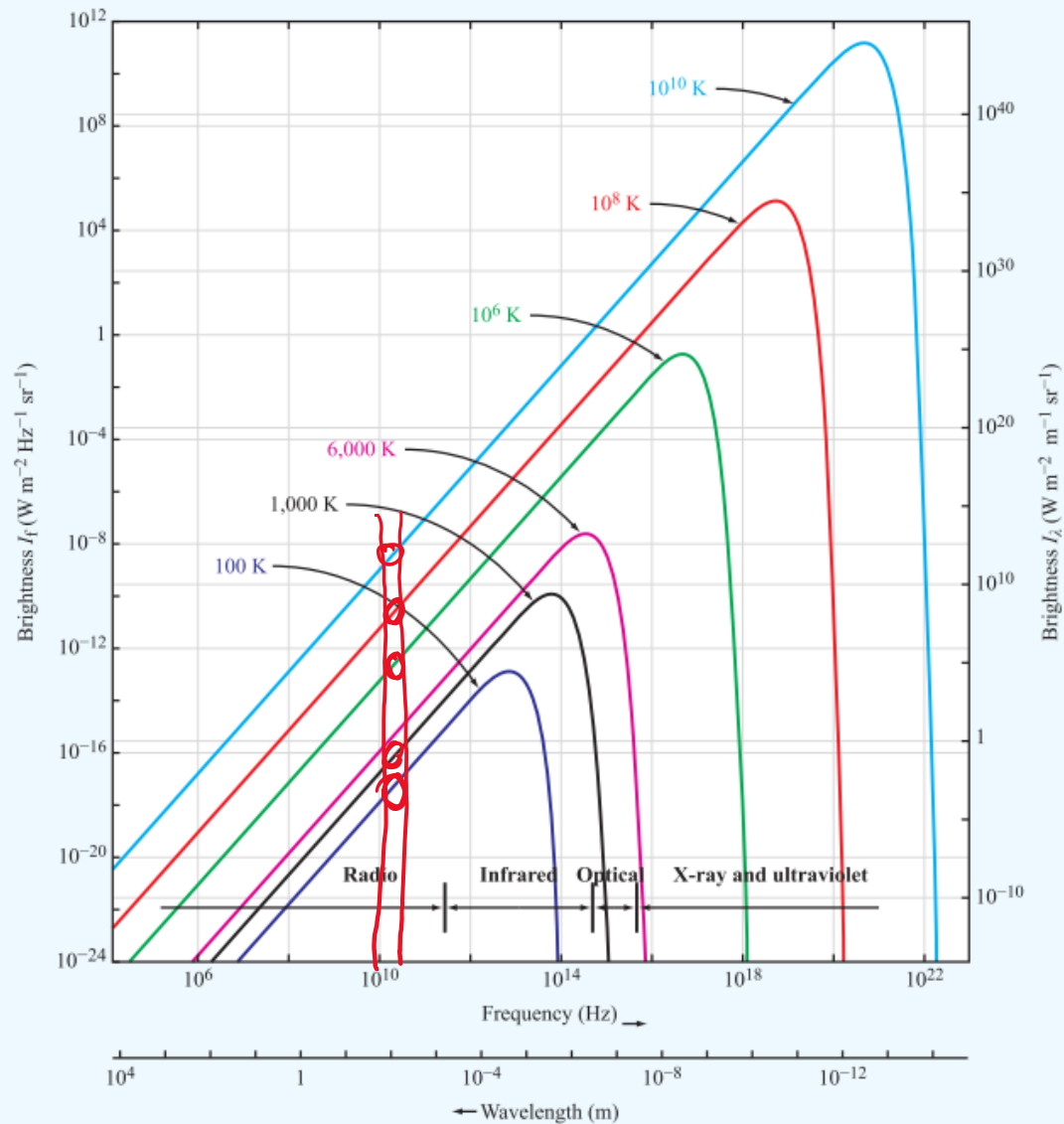


Figure 6-1: Planck's radiation law [adapted from Kraus, 1966].

Rayleigh–Jeans's law

The Rayleigh–Jeans approximation is very useful in the microwave region: it is mathematically simpler than Planck's law and yet its fractional deviation from Planck's exact expression is less than 1% if $\lambda T > 0.77 \text{ m K}$, or equivalently, $f/T < 3.9 \times 10^8 \text{ Hz K}^{-1}$.

$$I_f \approx \frac{2kT}{\lambda^2}$$

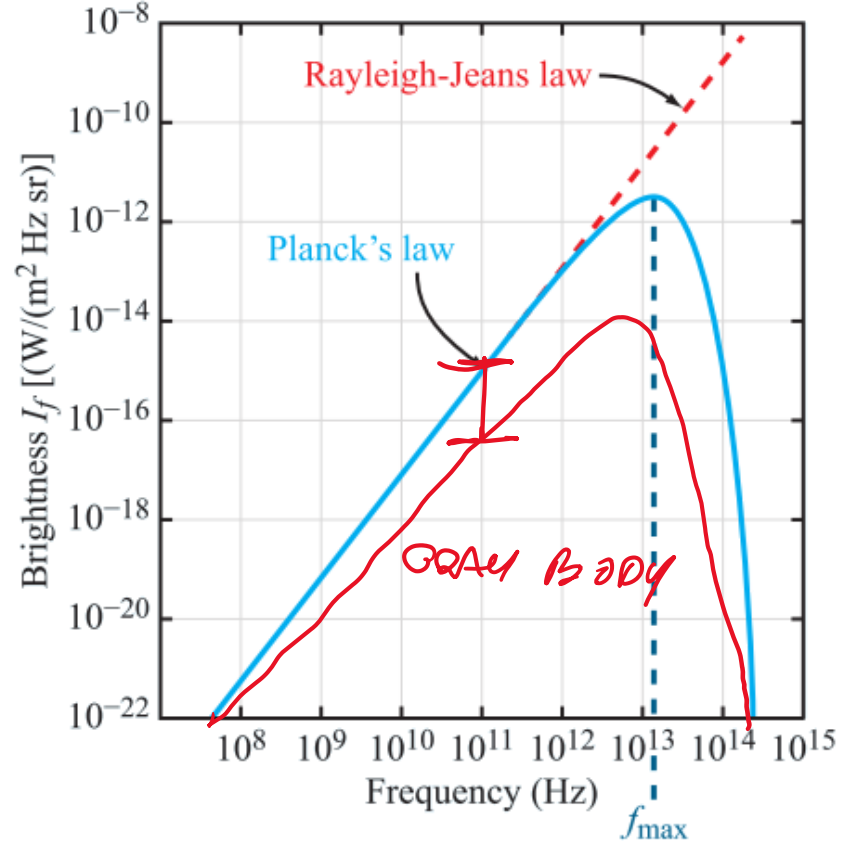


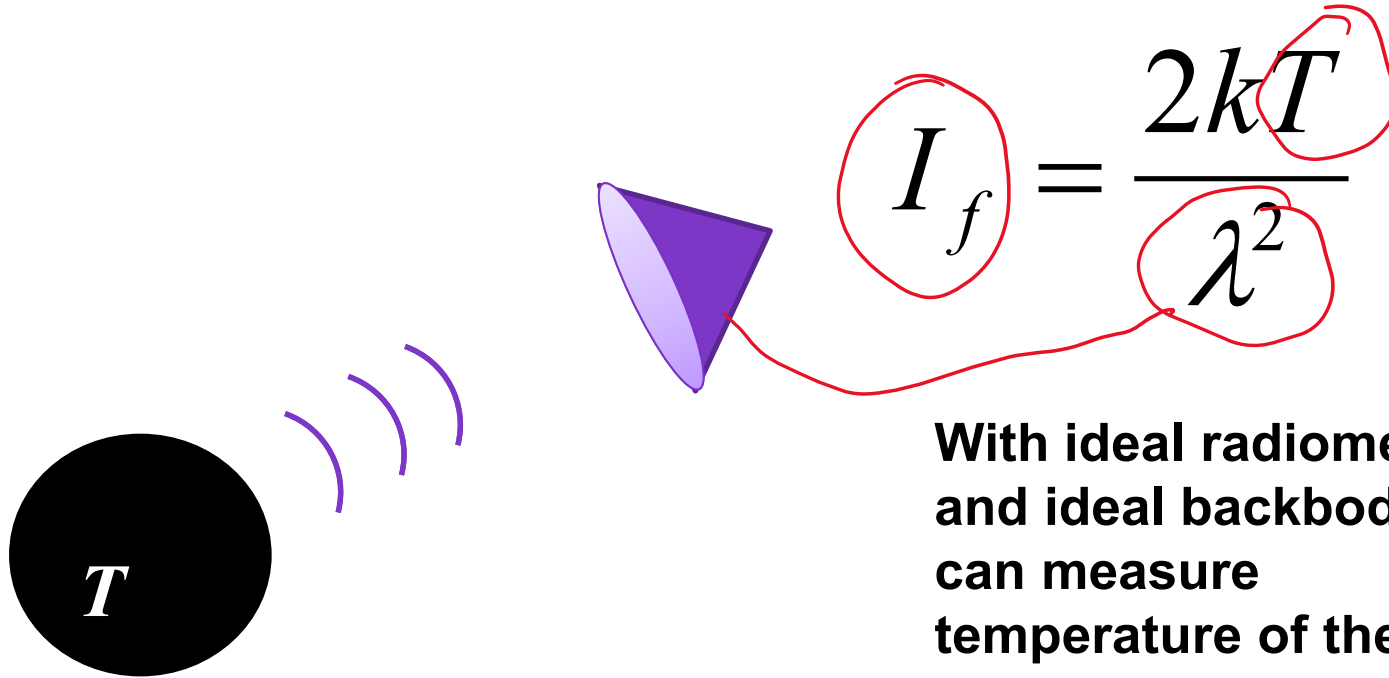
Figure 6-3: Comparison of Planck's law with its low-frequency approximation (Rayleigh–Jeans law) at 300 K.

Blackbody concept

A **blackbody** is defined as an idealized, perfectly opaque material that **absorbs all the incident radiation** at all frequencies, **reflecting none**.

A body in thermodynamic equilibrium emits to its environment the same amount of energy it absorbs from its environment. Hence, in addition to being a perfect absorber, a **blackbody also is a perfect emitter**.

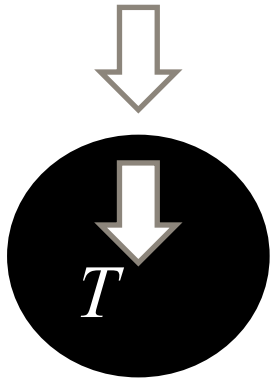
Measuring Blackbody temperature



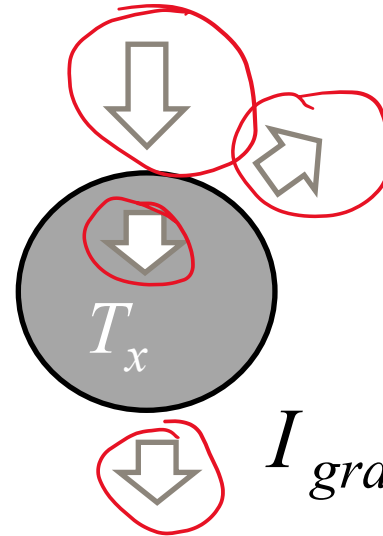
**With ideal radiometer
and ideal backbody we
can measure
temperature of the body.**

Gray body, perfect absorber is not realistic

Real body cannot absorb everything, because they also reflect and transmit something! If absorbing is not 100%, also emitting cannot be 100%.



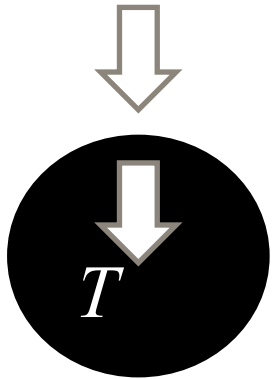
$$I_{black} = \sigma T^4$$



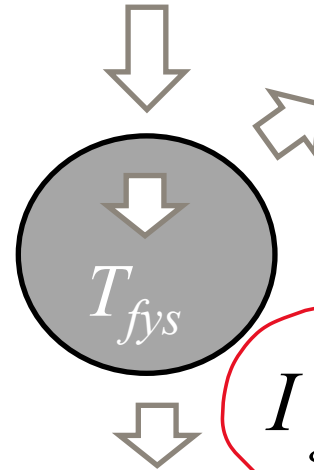
$$I_{gray} = e\sigma T_{real}^4$$

Gray body

Simplest approximation for real body is gray body, a body which emits and absorbs less than black body, scaled by a simple parameter ε emittance. This allows reflectance and transmittance!



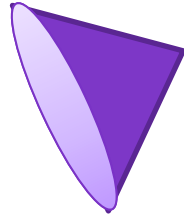
$$I_{black} = \sigma T^4$$



$$I_{gray} = \varepsilon \sigma T_{real}^4$$

Graybody brightness temperature

Appears like a
blackbody with
temperature T_B



$$I_f = \frac{2k \boxed{T_B}}{\lambda^2} = \frac{2k \boxed{e T_{real}}}{\lambda^2}$$

With ideal radiometer we can
measure apparent
temperature of the body and
emissivity, if we know the real
temperature!

$$T_{blackbody} = e T_{real}$$

$$e = \frac{T_{black}}{T_{real}}$$

Brightness temperature – the temperature of an imagined black body would emit to match the measurement.

Often noted as T_B or **TB**. Sometimes also pronounced TB.

It is easier to use temperature than brightness. Therefore, radiometer community talks about brightness temperature while they are meaning irradiance.

How emissivity is connected to electromagnetic properties??

Because emissivity is dependent on reflectivity and transmissivity, it depends on dielectric properties of the medium, as well as on many other parameters

$$\eta = \sqrt{(\mu/\epsilon)}$$

$$T_B(\theta) = e(\theta, \epsilon, \mu, \lambda, T) T_{real}$$

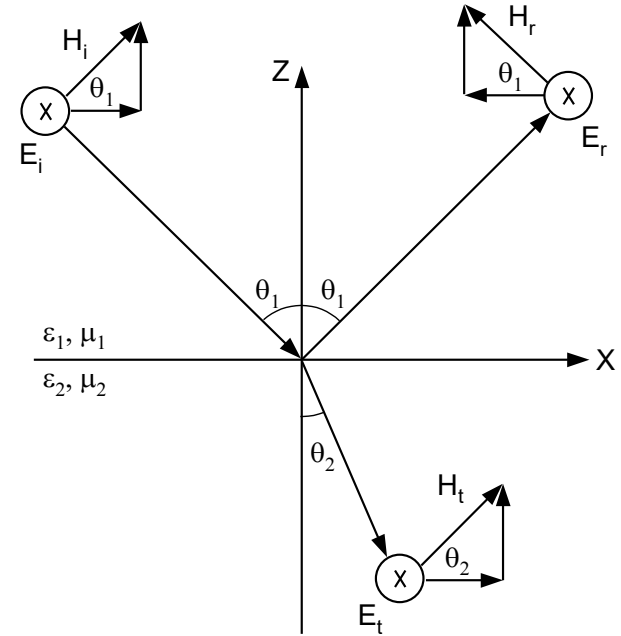
Reminder, reflectivity depends on dielectric properties

Reflection coefficient for H and V polarizations:

$$\rho_h = \frac{\eta_2 \cos \theta_1 - \eta_1 \cos \theta_2}{\eta_2 \cos \theta_1 + \eta_1 \cos \theta_2}$$

$$\rho_v = \frac{\eta_1 \cos \theta_1 - \eta_2 \cos \theta_2}{\eta_1 \cos \theta_1 + \eta_2 \cos \theta_2}$$

For power quantities: $R_v = |\rho_v|^2$

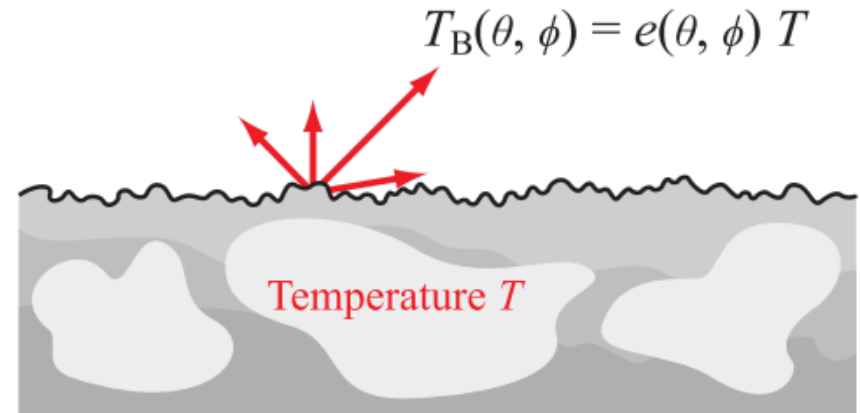


Emissivity

The ratio of the brightness intensity $I(\theta, \phi)$ of the material to that of a blackbody at the same temperature is defined as the emissivity $e(\theta, \phi)$:

$$e(\theta, \phi) = \frac{I(\theta, \phi)}{I_{\text{bb}}} = \frac{T_{\text{B}}(\theta, \phi)}{T}$$

$$0 < e < 1$$





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Radiometric Quantities

Chapter 6 in course book

Table 6-1: Standard units, symbols, and defining equations for fundamental radiometric quantities.

Microwave terminology	Optical terminology	Symbol	Defining equation	Unit	Abbreviation
Energy	Radiant energy	\mathcal{E}		joule	J
Power	Radiant flux	P	$P = \partial\mathcal{E}/\partial t$	watt	W
Power (or flux) density	Radiant flux	\mathcal{S}	$\mathcal{S} = \partial P/\partial A$	watt per square meter	Wm^{-2}
Brightness intensity	Radiance	I	$I = \partial^2 P/\partial\Omega \partial A$	watt per steradian per square meter	$\text{W sr}^{-1}\text{m}^{-2}$
Emissivity	Emissivity	e	$e = I/I_{\text{blackbody}}$	(unitless)	
Reflectivity	Reflectance	Γ	$\Gamma = P^r/P^i$	(unitless)	
Absorptivity	Absorptance	a	$a = P^a/P^i$	(unitless)	
Transmissivity	Transmittance	\mathbb{T}	$\mathbb{T} = P^t/P^i$	(unitless)	

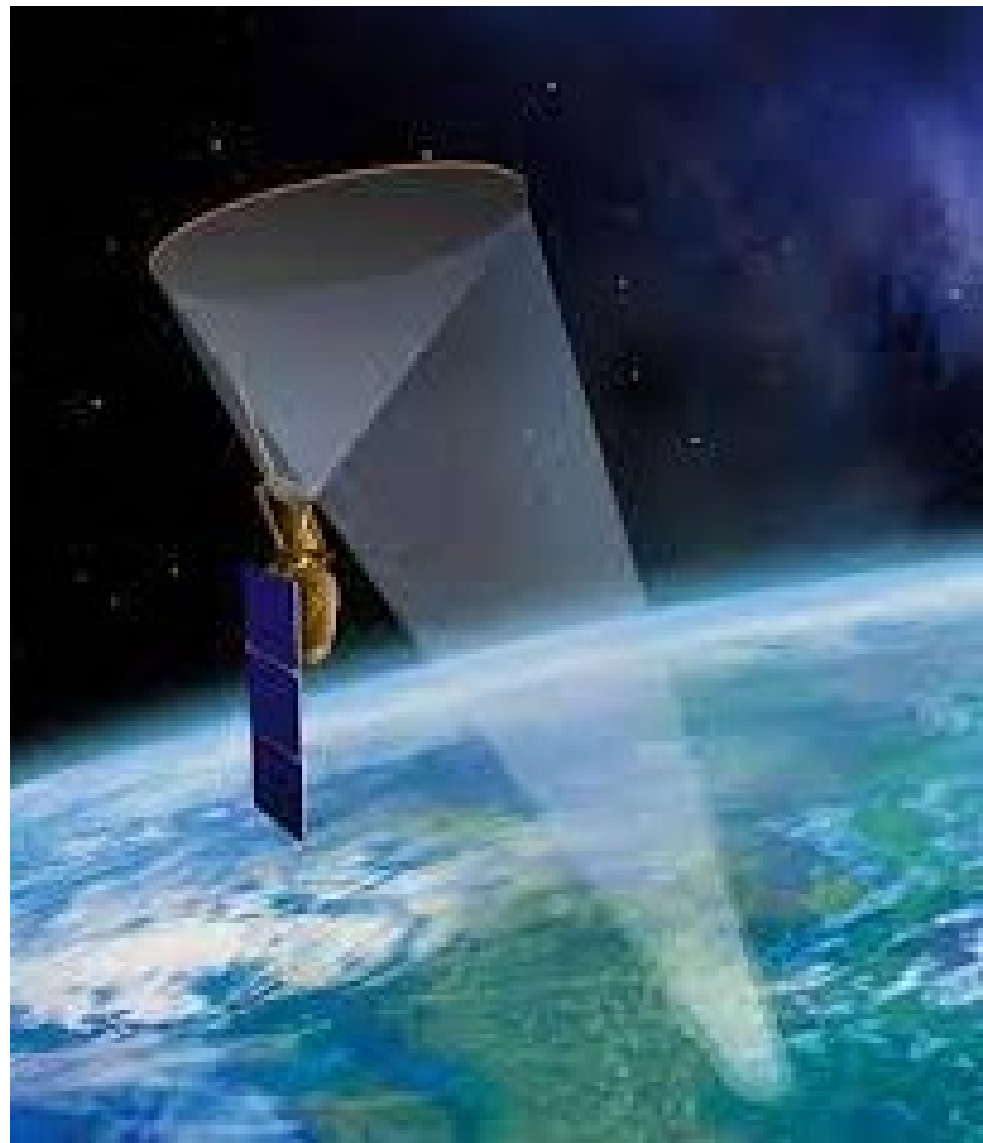
Superscripts: i = incident, r = reflected, a = absorbed, and t = transmitted.



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Antennas and dependence on direction

Chapter 6 in course book



Antenna temp

The antenna temperature T' is the brightness temperature reduced to the antenna location.

An ideal and lossless antenna would deliver this temperature to receiver.

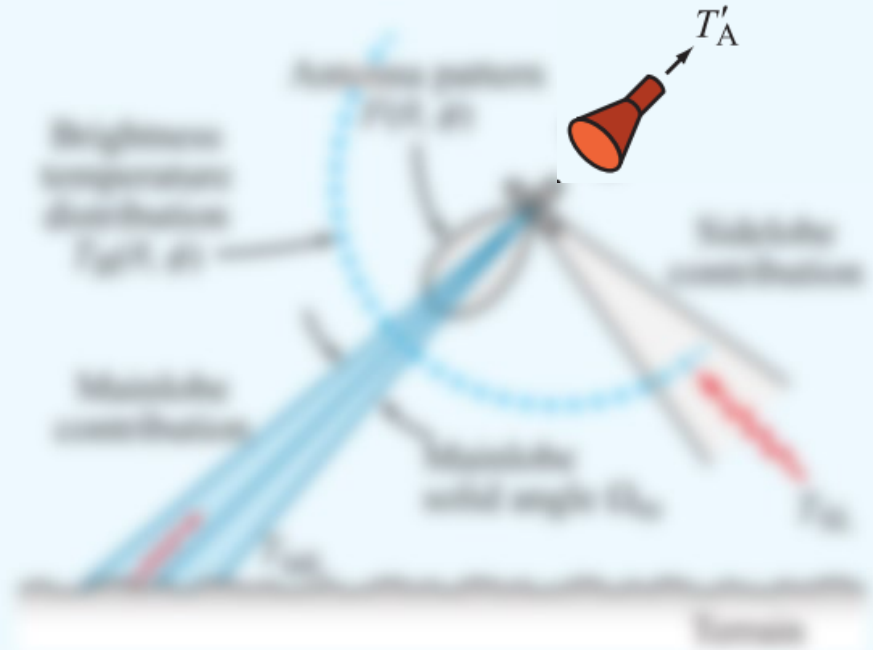


Figure 6.9 T_{gal} and T_{cwb} are the galactic and cosmic contributions to antenna temperature T'_A .

Generic Radiometer system

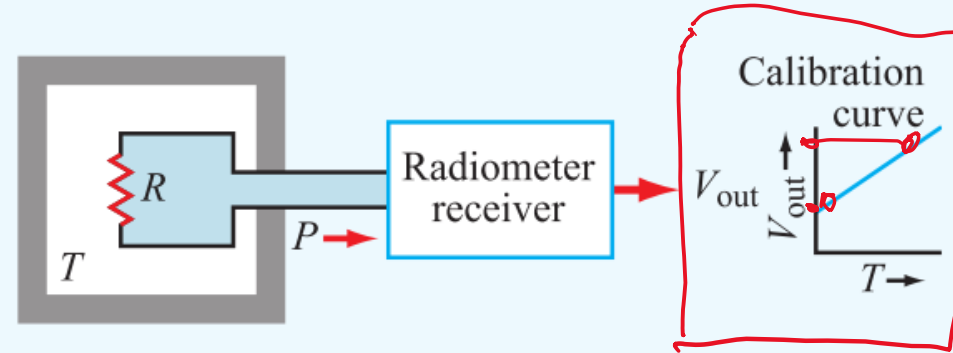
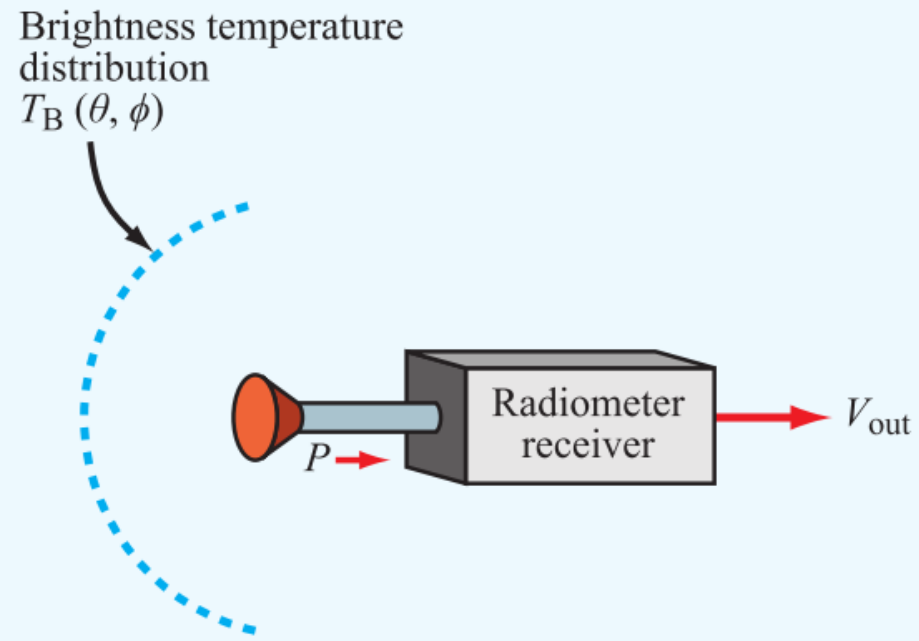


Figure 6-8: The power received by an antenna is equivalent to the noise power delivered by a matched resistor.

Antenna temperature: the brightness temperature a lossless antenna would deliver to receiver.

Beam efficiency and sidelobes

$$\eta_b = \frac{\Omega_m}{\Omega_p}.$$

$$T'_A = \eta_b T_{ML} + (1 - \eta_b) T_{SL}.$$

(6.36)

(6.37)

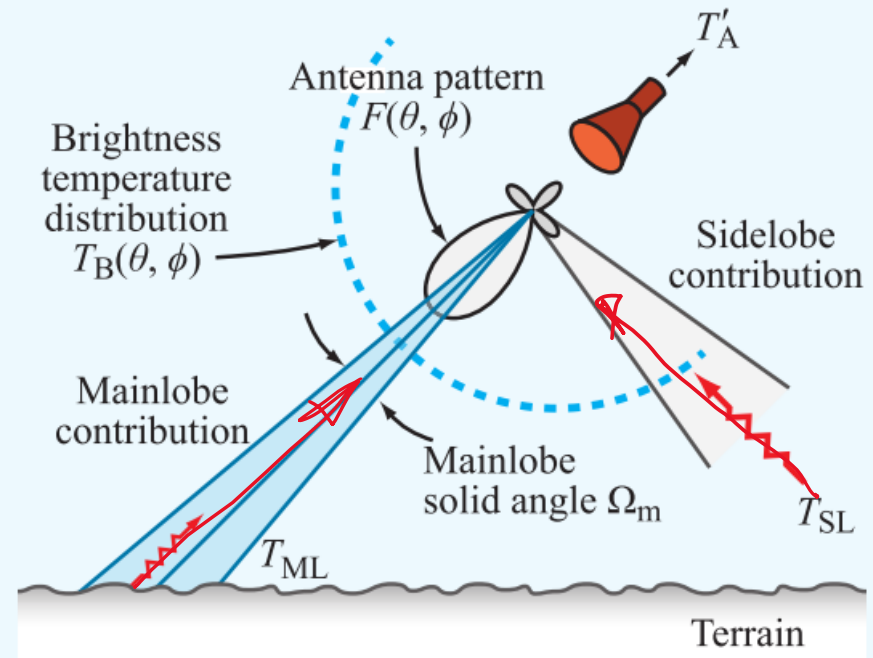


Figure 6-9: T_{ML} and T_{SL} are the mainlobe and sidelobe contributions to antenna temperature T'_A .

$$P = \int_{f_1}^{f_2} P_f df = A_r \Omega_s \int_{f_1}^{f_2} I_f df. \quad (6.11)$$

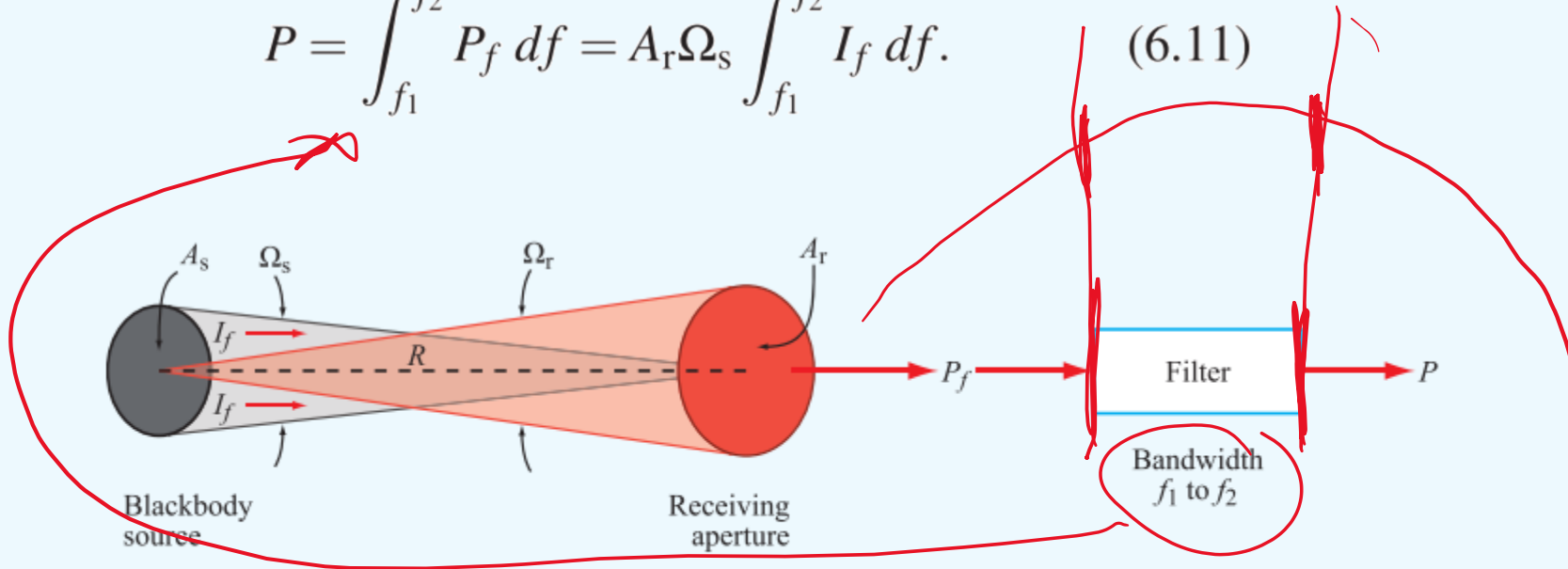
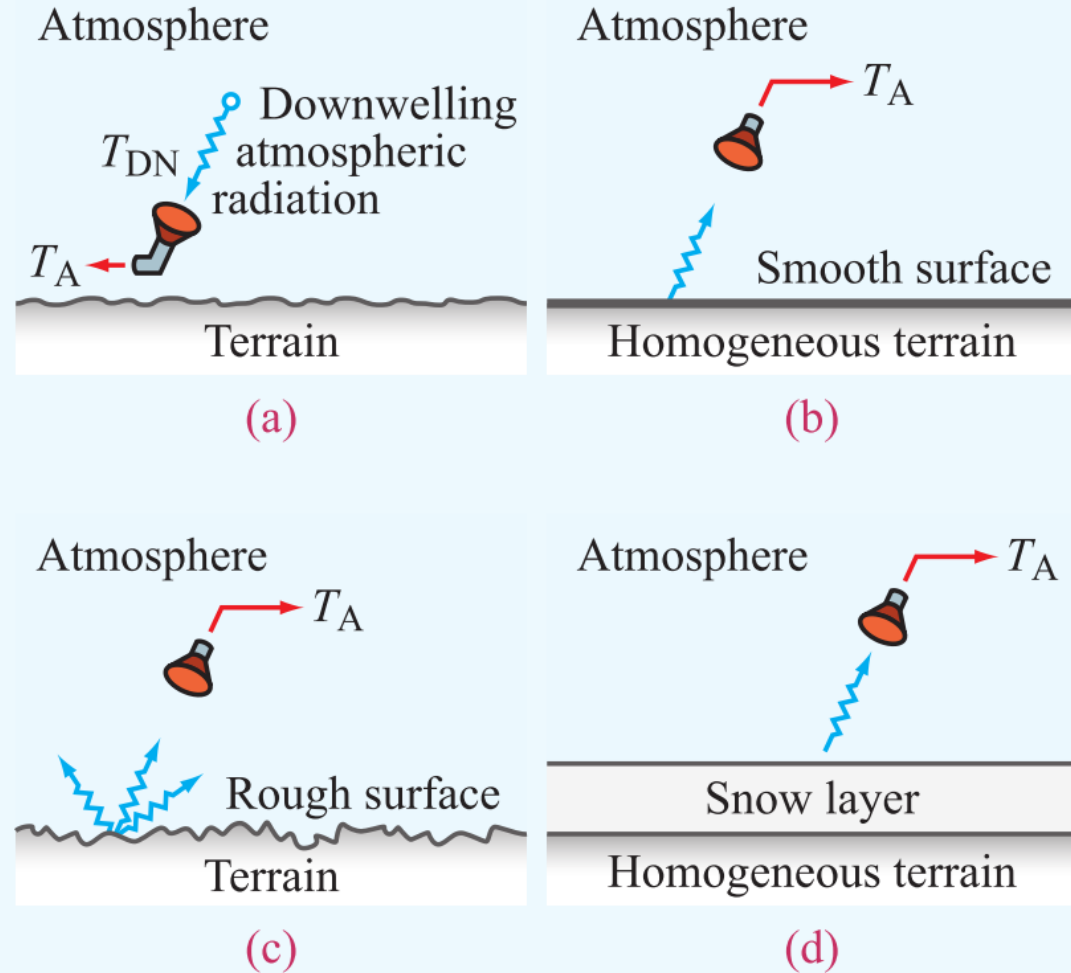
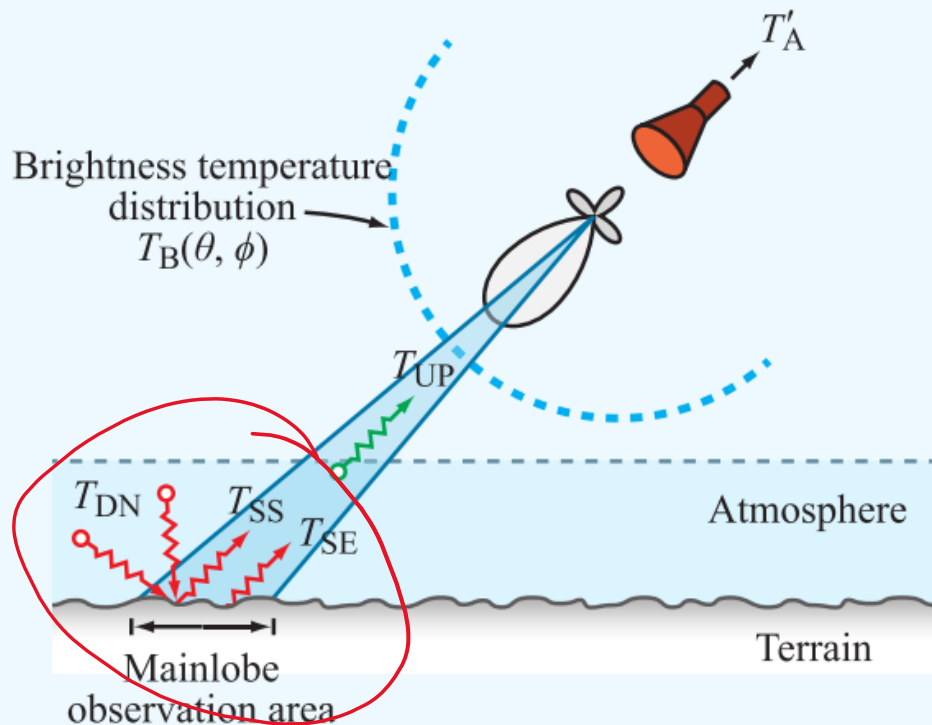


Figure 6-2: Geometry for power received from a blackbody source.

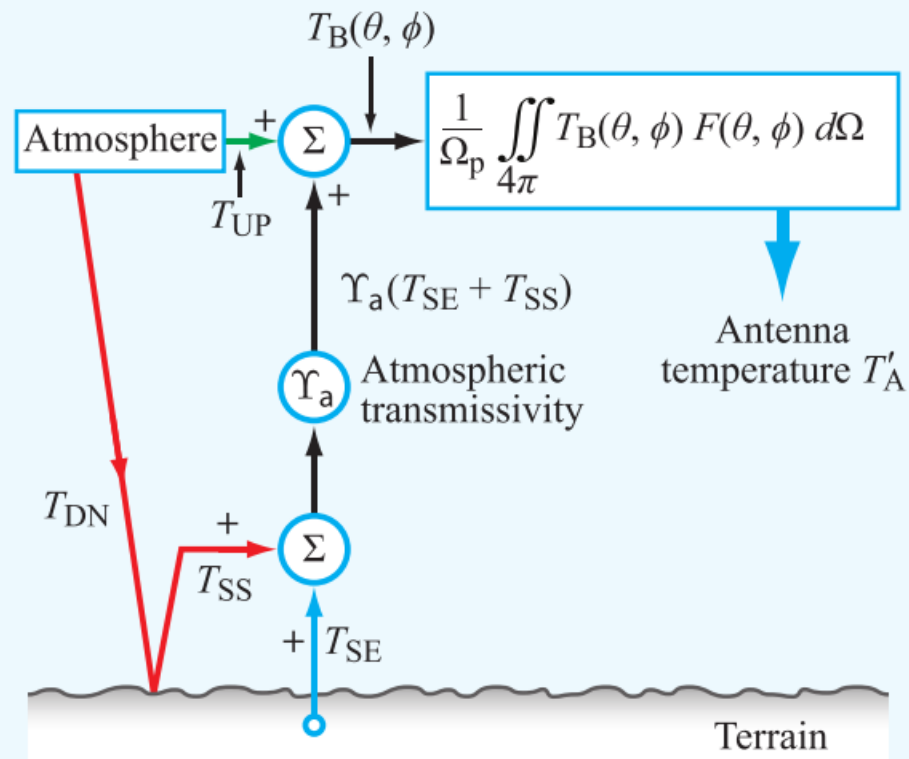
Figure 6-11: Examples of configurations of interest in radiometric remote sensing: (a) upward-looking radiometer, and downward-looking radiometer with (b) smooth-surface boundary, (c) rough-surface boundary, and (d) two-layer terrain.



T_{SE} = Surface emission
 T_{DN} = Atmospheric downward emission
 T_{UP} = Atmospheric upward emission
 T_{SS} = Surface scattered radiation



(a) Emission by surface and atmosphere



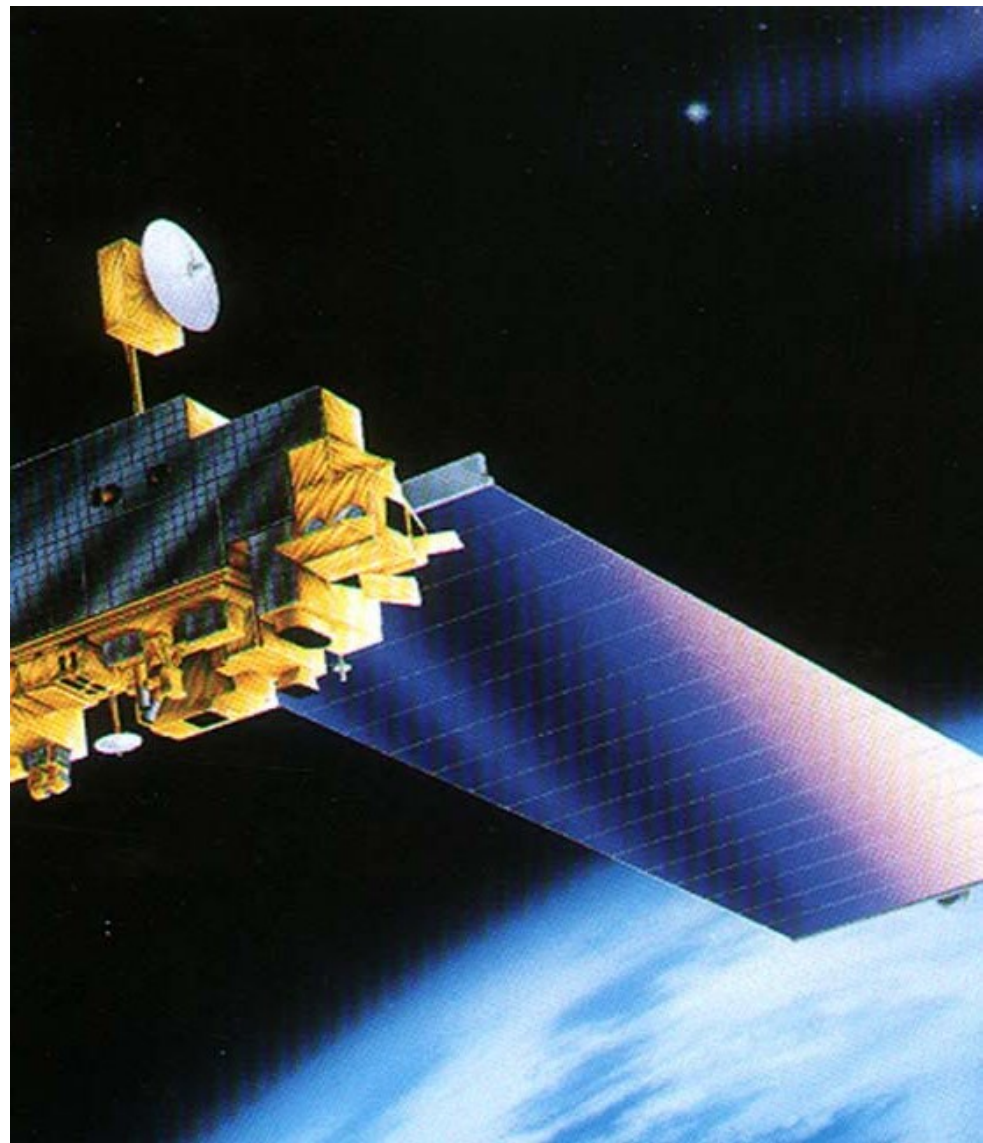
(b) Block diagram

TB brightness temperature distribution



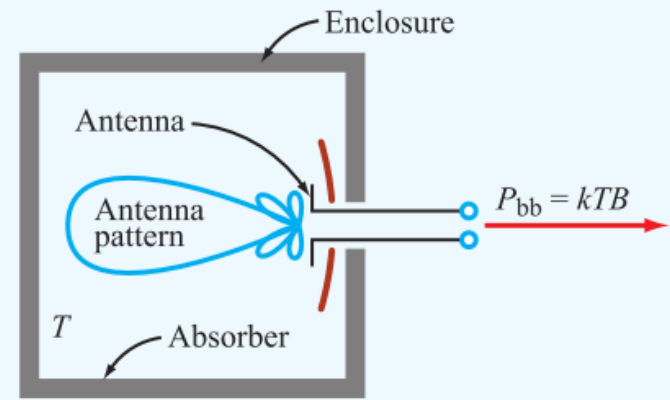
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Radiometer system

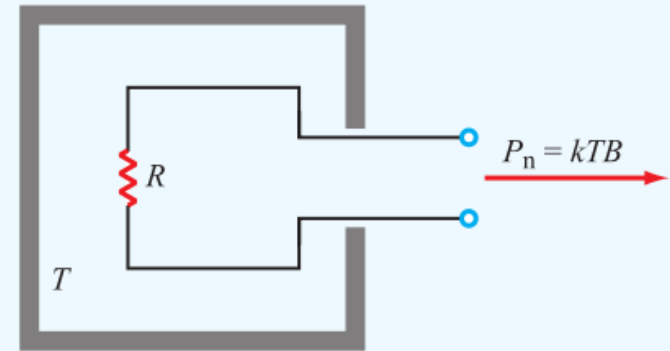


Power and temperature equivalence

From the standpoint of an ideal receiver of bandwidth B , the antenna connected to its input terminals is equivalent to a resistance R_{rad} , called the antenna **radiation resistance**.



(a) Antenna inside a blackbody enclosure



(b) Resistor at temperature T

Figure 6-5: The power delivered by (a) an antenna placed inside a blackbody enclosure of temperature T is equal to the power delivered by (b) a resistor maintained at the same temperature.

Power received by an antenna

effective aperture A_r *antenna pattern* $F(\theta, \phi)$

$$dP_f = I_f A_r F(\theta, \phi) d\Omega. \quad (6.15)$$

$$P = A_r \int_{f_1}^{f_2} \iint I_f F(\theta, \phi) d\Omega df, \quad (6.16)$$

$$P = \frac{1}{2} A_r \int_{f_1}^{f_2} \iint_{4\pi} I_f F(\theta, \phi) d\Omega df. \quad (6.17)$$

$B = (f_2 - f_1)$

(polarized antenna)

$$P_{bb} = kTB \frac{A_r}{\lambda^2} \iint_{4\pi} F(\theta, \phi) d\Omega. \quad (6.19)$$

$\Omega_p = \frac{\lambda^2}{A_r}$

$$P_{bb} = kTB. \quad (6.22)$$

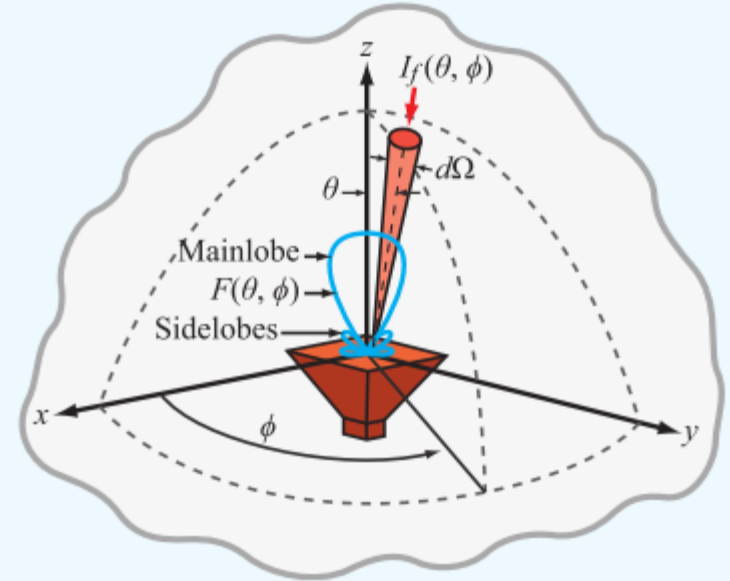


Figure 6-4: Blackbody spectral brightness I_f incident on an antenna with effective aperture A_r and radiation pattern $F(\theta, \phi)$.

Brightness temp and brightness intensity

Because of the one-to-one correspondence between the brightness temperature

$T_B(\theta, \varphi) \sim I(\theta, \varphi)$ often T_B is used instead of intensity I .

T_B means usually intensity!

$T_B(\theta, \varphi)$ is called brightness temperature distribution.

Simple Radiometer system

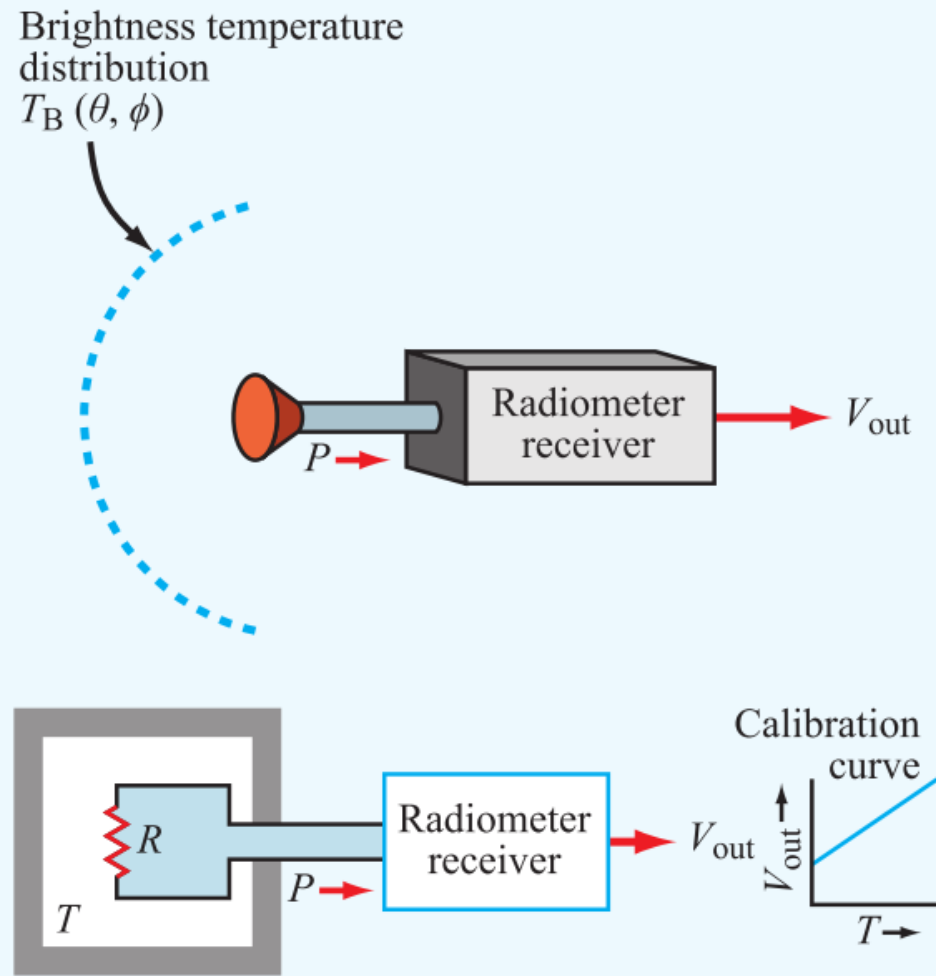
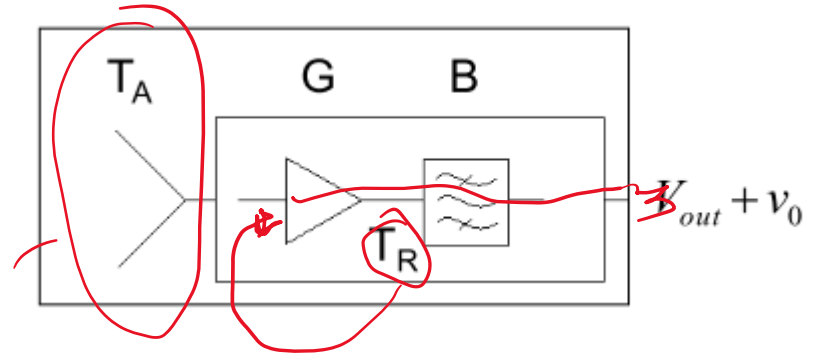


Figure 6-8: The power received by an antenna is equivalent to the noise power delivered by a matched resistor.

Generic radiometer system

The radiometer -

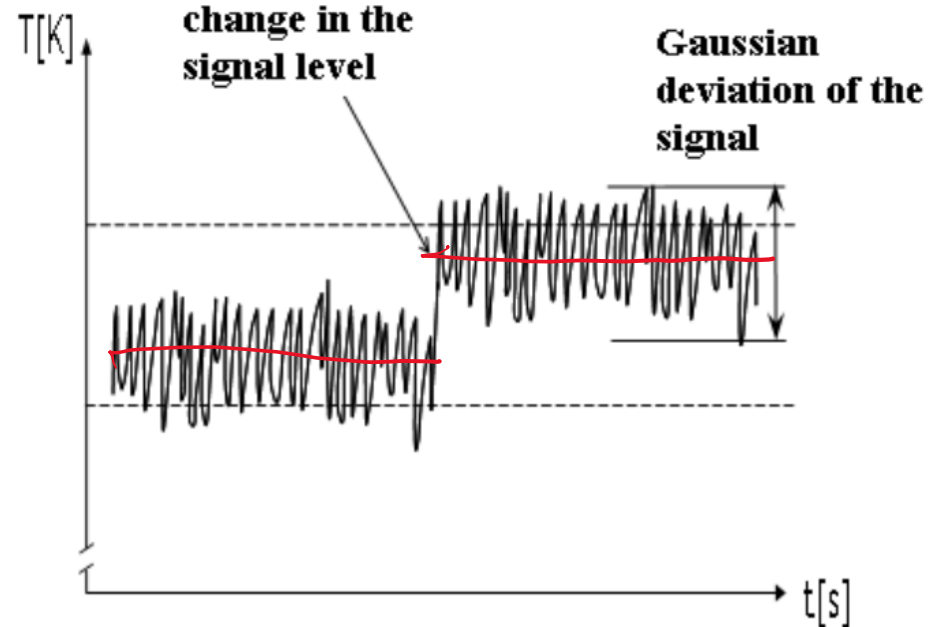
- antenna collects radiation from the target
- amplifies (G =gain) and filters (B =bandwidth) the collected signal
- detects the signal intensity, i.e. the noise power coming from the antenna.



$$V_{out} = k \cdot G \cdot T_{SYS} \cdot B + v_0 = k \cdot G \cdot (T_A + T_R) \cdot B + v_0$$

Measured signal is **thermal noise**

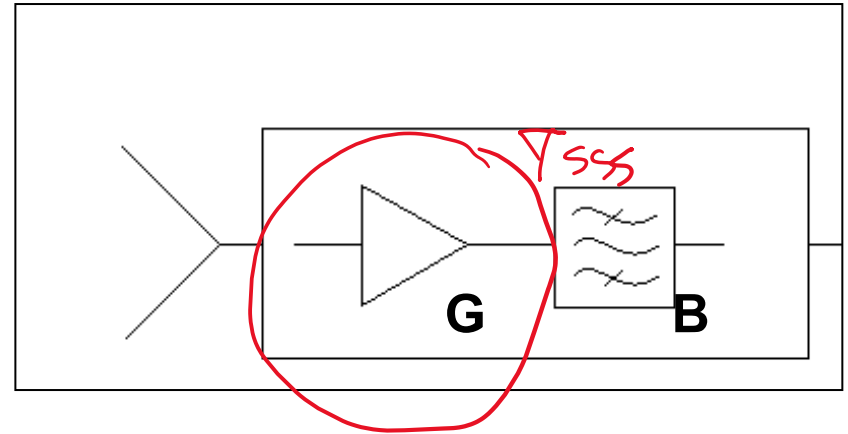
- ✓ The measured signal is electromagnetic noise, i.e. it's a composition of all the possible polarizations and phase states.
- ✓ Measurement of a noisy signal is noisy, i.e. inaccurate. Averaging of the result helps.



Signal is weak

- As predicted by Planck's law, level of the measured signal is **LOW** (10^{-13} – 10^{-14} Watts. Also, bandwidths reserved for passive use are reasonably narrow.
- In order to detect the incoming power high gain is required.
- High gain causes problems like crosstalk, coupling, thermal instability etc..

$$V_{out} = k \cdot G \cdot T_{SYS} \cdot B$$

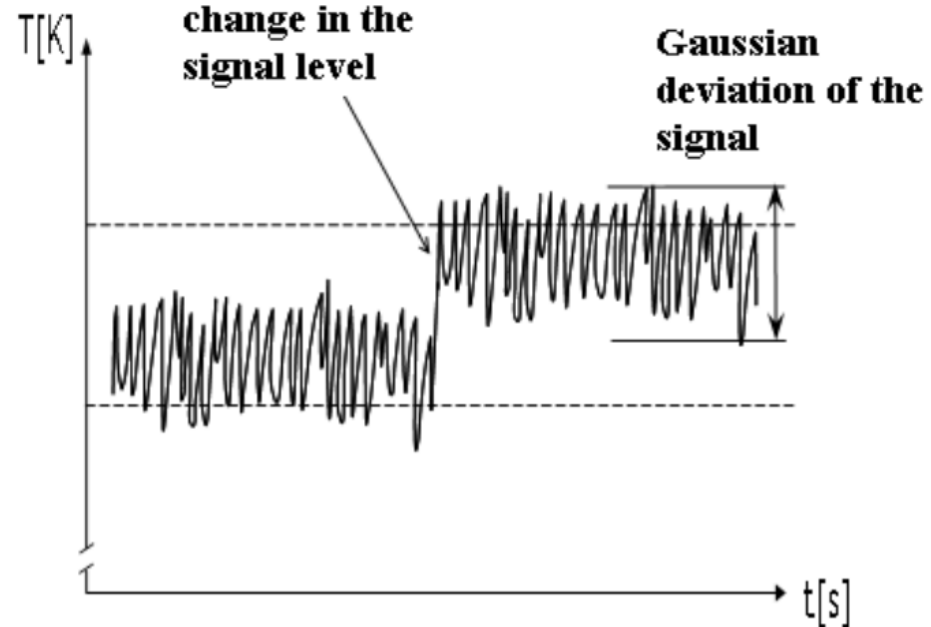


Measurement accuracy

- ✓ The best accuracy which can be achieved by a measurement of a radiometer is called **Radiometric Resolution**:

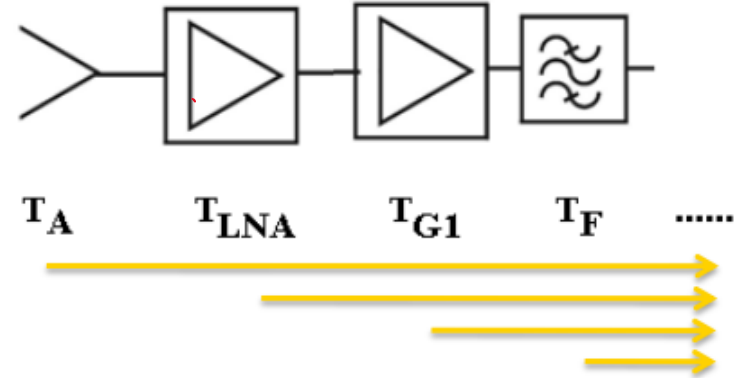
$$\Delta T = C \frac{T_{SYS}}{\sqrt{B\tau}}$$

T_{SYS} system temperature
 B bandwidth
 τ integration time
 C Constant depending on receiver type



Receiver generates and amplifies it's own noise as well!

- ✓ As predicted by Planck's law, also all electrical components create noise.
- ✓ This noise mixes with the noise from the antenna. In order to solve the noise from antenna, receiver's own noise needs to be characterized.
- ✓ Receivers equivalent noise temperature T_R reduces all the noise components to the input of the receiver.



$$T_R = T_1 + \frac{T_2}{G_1} + \frac{T_3}{G_1 G_2} + \dots$$

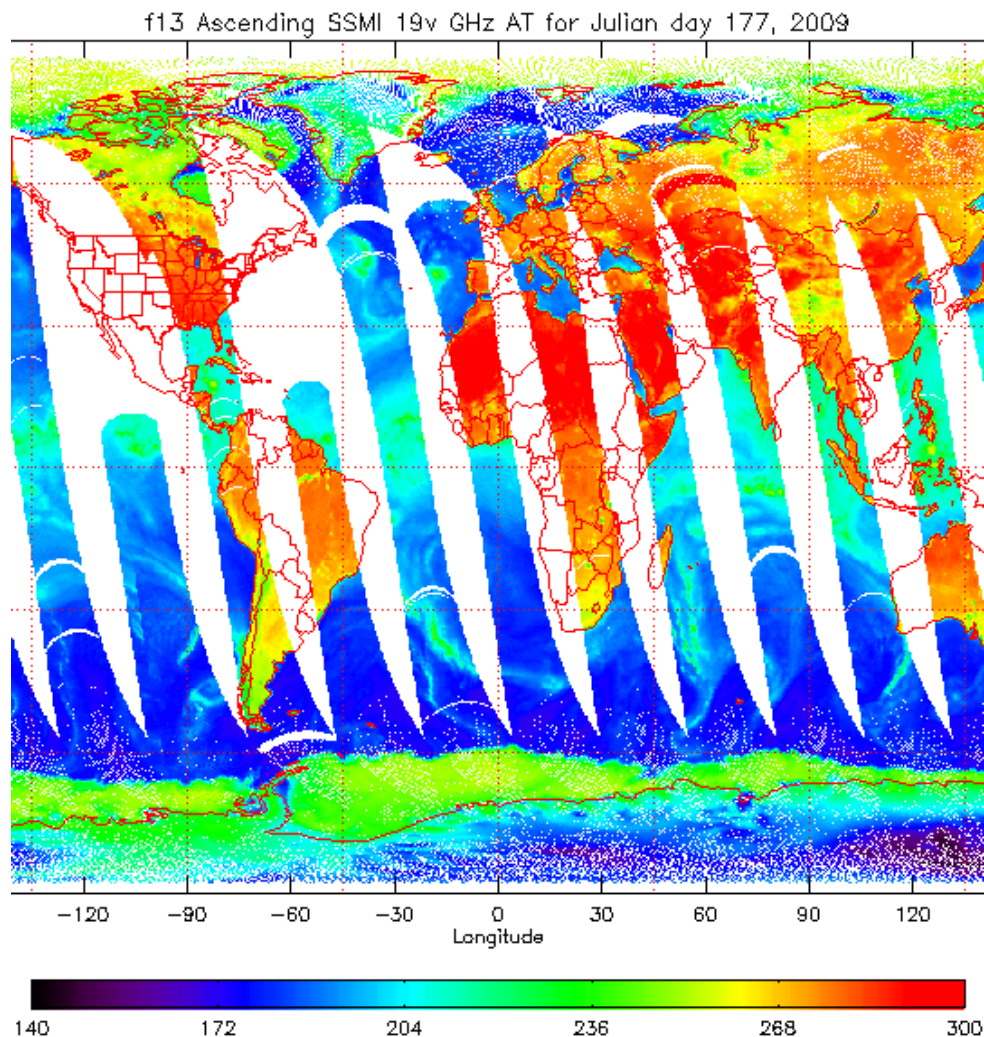


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Radiometer systems types

Chapter 7 in the course book

Image by SSMI radiometer



Radiometers

Noise-injection radiometer

Dicke-type radiometer

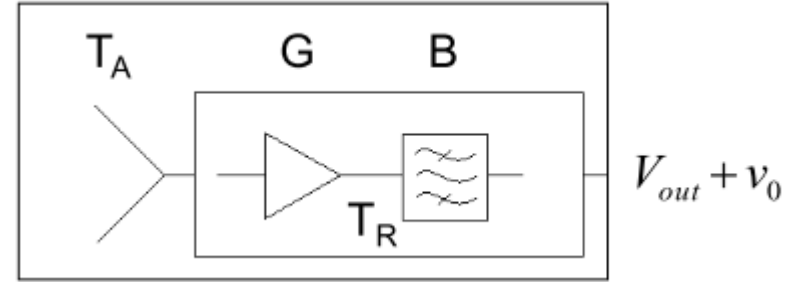
Total power radiometer

Complexity, performance, price, etc.
increases

Simple radiometer

The radiometer -

- antenna collects radiation from the target
- amplifies (G=gain) and filters (B=bandwidth) the collected signal
- detects the signal intensity, i.e. the noise power coming from the antenna.

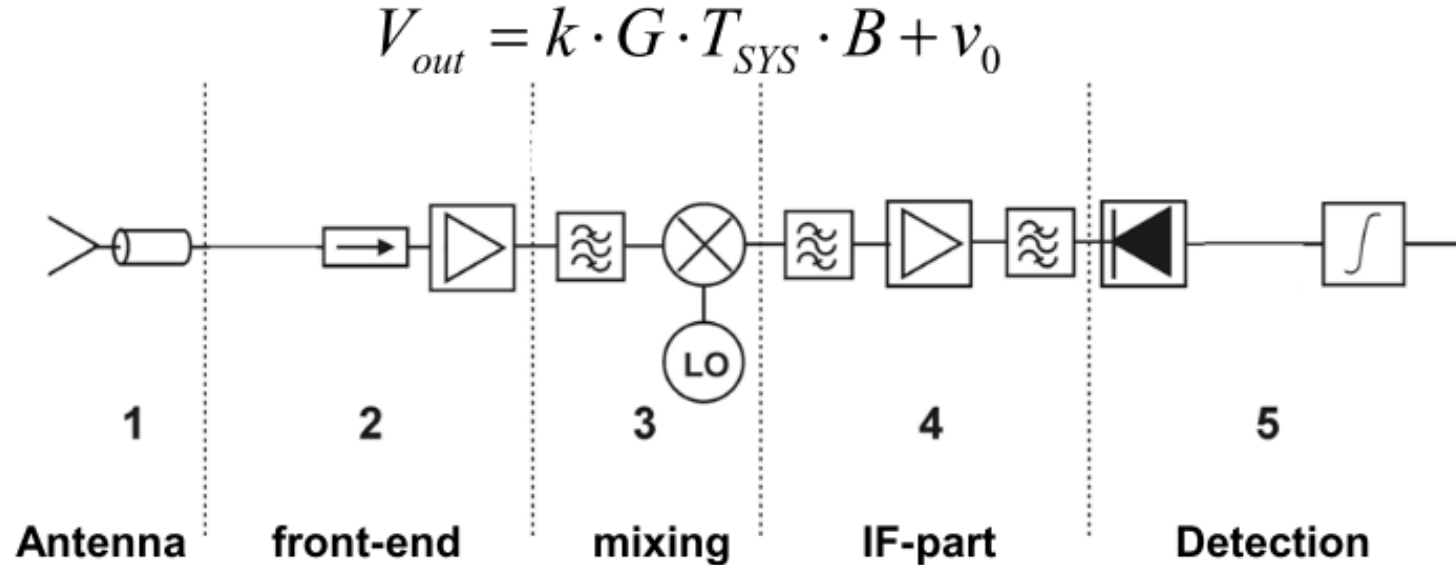


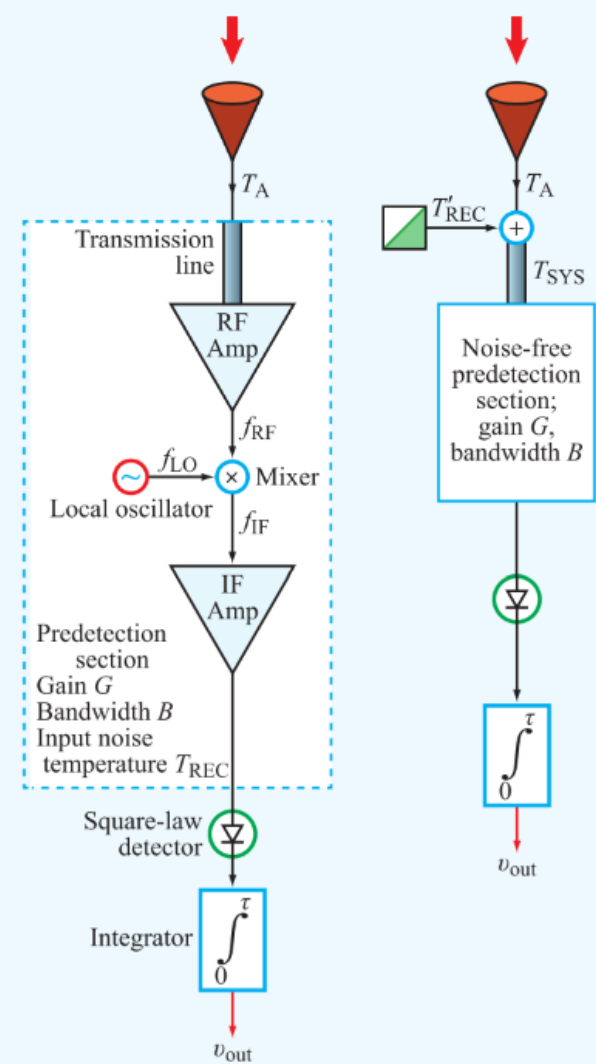
$$T_{RECEIVER} = T_{SYS}$$

$$V_{out} = k \cdot G \cdot T_{SYS} \cdot B + v_0 = k \cdot G \cdot (T_A + T_R) \cdot B + v_0$$

Total power radiometer

The name emerges from the fact that the receiver simply detects the total power that is propagated through the receiver chain.





(a) Total-power receiver

(b) Equivalent receiver

Figure 7-13: The representation in (b) replaces the predetection section with a noise-free equivalent and refers the receiver noise to the antenna terminals.

Total power radiometer

$$V_{out} = k \cdot G \cdot (T_R + T_A) \cdot B + v_0 \quad \Leftrightarrow \quad T_A = \frac{V_{out} - v_0}{kGB} - T_R$$

- In order to calculate T_A from the detected output voltage V_{out} , one needs to know G , B , v_0 , and T_R .
- Solving of these by some means is called **RADIOMETER CALIBRATION**, and accuracy of it determines the accuracy of the radiometer. This, because error in any of the parameters propagates directly to the error of T_A .
- Radiometric resolution of the total power radiometer is ($C=1$):

$$\Delta T = \frac{T_{SYS}}{\sqrt{B\tau}}$$

Typically e.g.:

$$\begin{aligned} T_{SYS} &= 600 \text{ K;} \\ B &= 20 \text{ MHz;} \\ \tau &= 1 \text{ s} \end{aligned}$$



$$\Delta T \approx 0.13 \text{ K}$$

Total power radiometer

$$V_{out} = k \cdot G \cdot (T_R + T_A) \cdot B + v_0 \quad \Leftrightarrow \quad T_A = \frac{V_{out} - v_0}{kGB} - T_R$$

- In practice, the gain (and T_R) is the most unstable parameter in radiometers. Changes in gain propagate to errors in T_A . In the case of a total power radiometer this influences T_{SYS} :

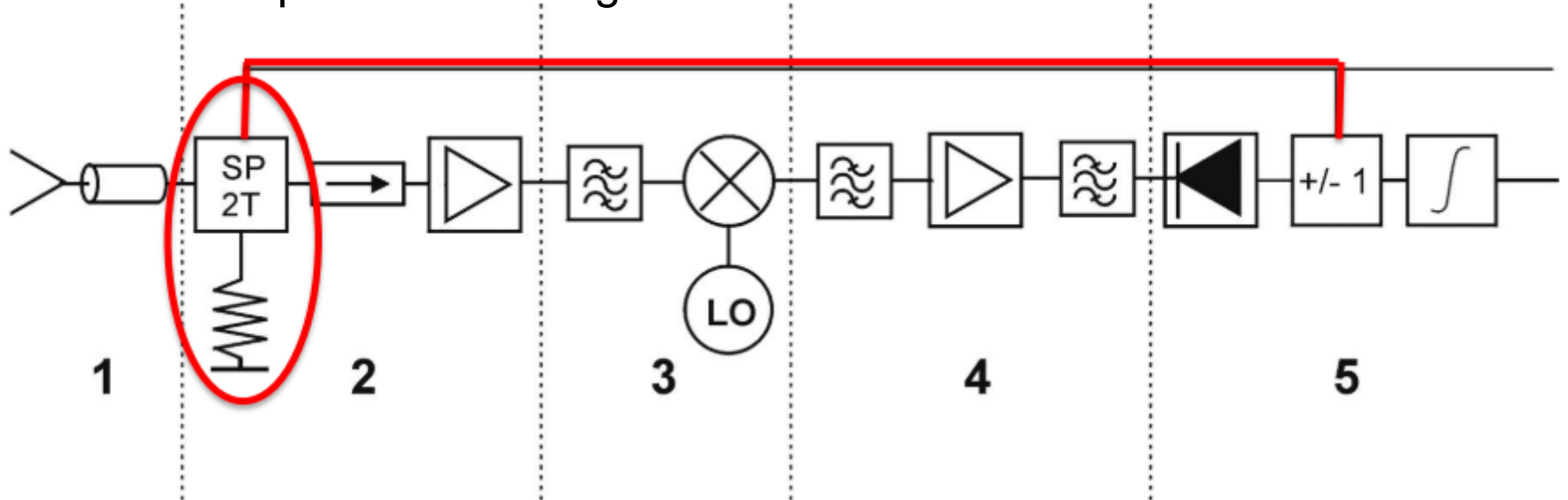
$$\Delta T_A = -\frac{V_{out} - v_0}{kG^2 B} \Delta G = -\frac{V_{out} - v_0}{kGB} \frac{\Delta G}{G} = T_{SYS} \frac{\Delta G}{G}$$

e.g. 600 K

- Typically $\Delta G/G \approx 10^{-2}..10^{-3}$, so, $\Delta T_A \approx < 10K$. This dominates the radiometric resolution. This is the weakness of total power radiometers.

Dicke Radiometer

- Dicke Radiometer measures the antenna the first half of the integration period, and a reference load for the second half.
- This is implemented using so-called Dicke-switch

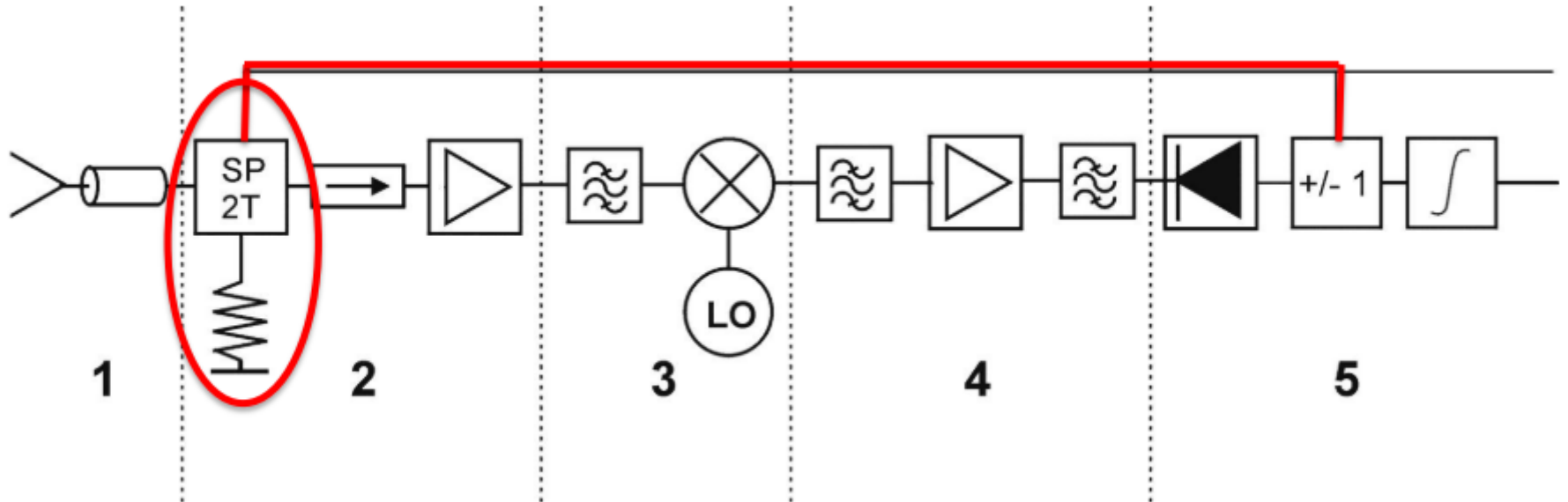


Dicke Radiometer

$$V_{out} = V_{out_{T_D}} - V_{out_{T_A}}$$

$$V_{out_{T_A}} = kB G(T_R + T_A) + v_0$$

$$V_{out_{T_D}} = kB G(T_R + T_D) + v_0$$



Dicke Radiometer

$$V_{out} = kB G(T_D - T_A) \quad \Leftrightarrow \quad T_A = T_D - \frac{V_{out}}{kGB}$$

- Solving the antenna temperature from this doesn't require calibration of receiver noise temperature TR or voltage offset v0.
- Radiometric resolution of Dicke radiometer is decreased, since the antenna is measured only half of the integration period (C=2):

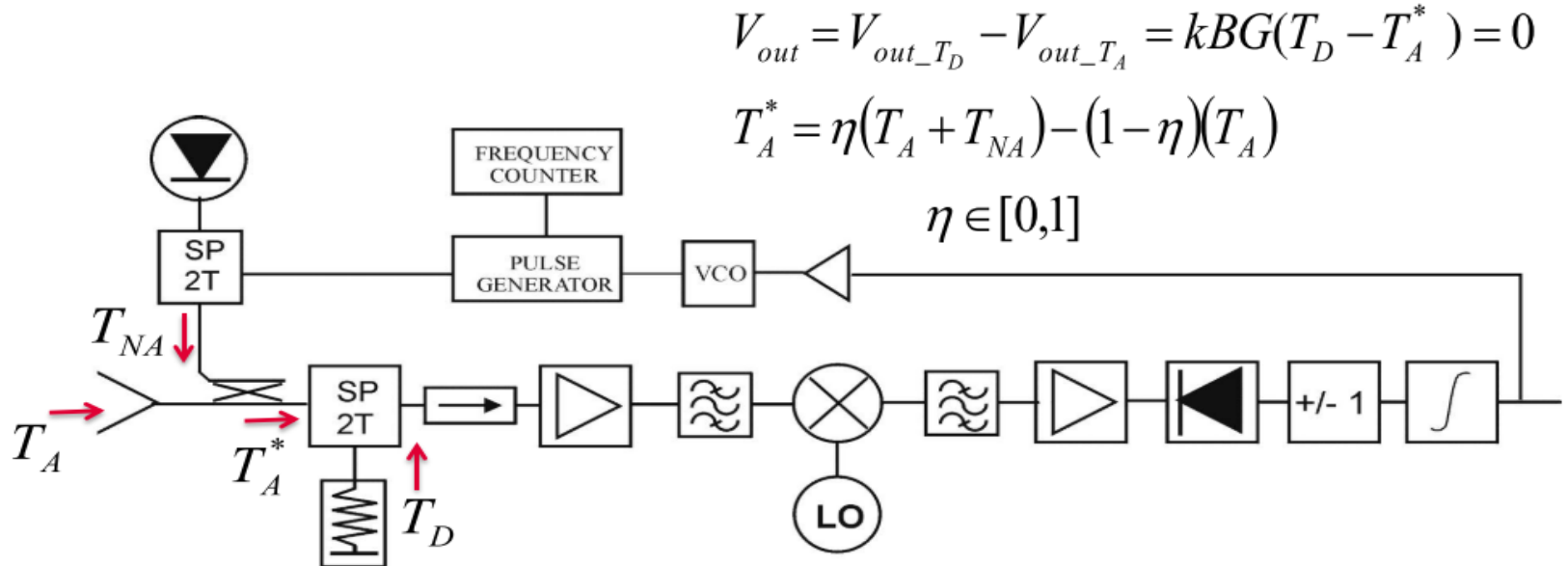
$$\Delta T = \frac{2 \cdot T_{SYS}}{\sqrt{B\tau}}$$

$$\Delta T_A = \frac{V_{out}}{kG^2 B} \Delta G = -\frac{V_{out}}{kGB} \frac{\Delta G}{G} = (T_A - T_D) \frac{\Delta G}{G}$$

Balanced Dicke Radiometer (noise injection radiometer)

- A Dicke radiometer with additional noise diode circuitry, from which an additional noise pulse is coupled into the antenna branch at each integration period
- The length η of the injected noise pulse is controlled by a feedback loop so, that the power from antenna branch equals to the noise power from the reference load.

A?

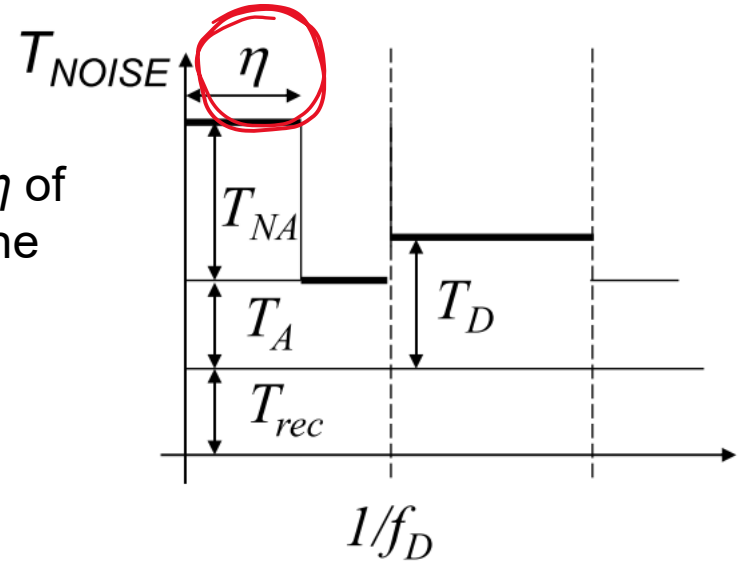


Balanced Dicke Radiometer (noise injection radiometer)

- One integration period of a noise injection radiometer:
- $V_{out}=0$ by definition
- The actual measurement result is the length η of the injected pulse that is needed to balance the powers.

$$T_D = \eta(T_{NA} + T_A) + (1 - \eta)T_A$$

$$T_A = T_D - \eta T_{NA}$$



Balanced Dicke Radiometer (noise injection radiometer)

$$T_A = T_D - \eta T_{NA}$$

- Measurement of noise injection radiometer is not depending on Gain, bandwidth, of receiver noise temperature calibration.
- The only variable that needs calibration is the noise power of the noise injection circuitry, T_{NA} .
- Radiometric resolution is constant, (i.e. Independent on target):

$$\Delta T_A \approx \frac{2 \cdot (T_{ref} + T_{rec})}{\sqrt{B\tau}}$$



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Calibration of a radiometer



Radiometer calibration

- Calibration means determination of instrument's parameters so that the calculation of the main observable becomes possible:

TOTAL POWER RADIOMETER:

$$T_A = \frac{V_{out} - v_0}{kGB} - T_R \quad G; B; T_R; v_0$$

DICKE RADIOMETER

$$T_A = T_D - \frac{V_{out}}{kGB} \quad G; B$$

NOISE INJECTION RADIOMETER

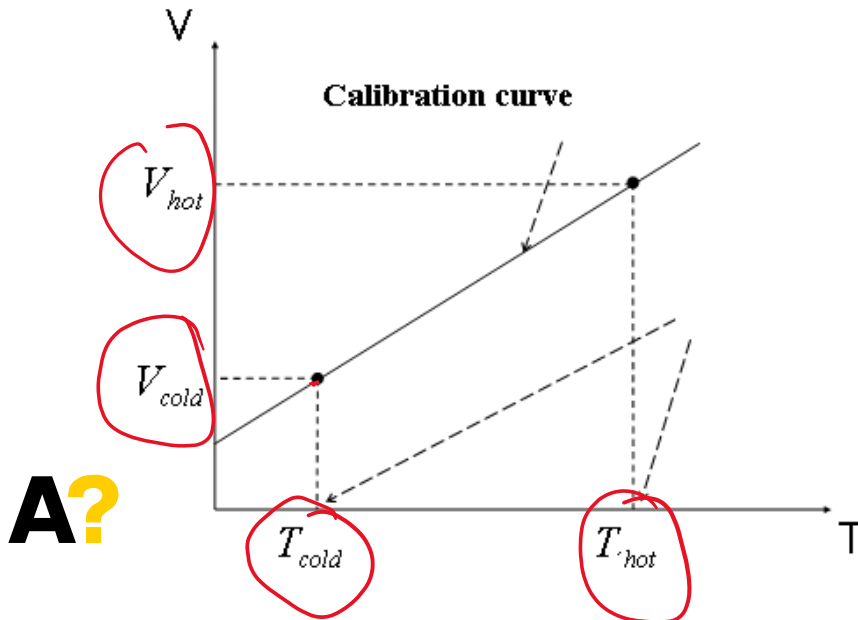
$$T_A = T_D - \eta T_{NA} \quad T_{NA}$$

Calibrating with two known temperatures

- Total power and Dicke radiometers require calibration of receiver's G and B.
- Linear model with A and B
- Requires measurements of two known targets.
- **Neglects the voltage offset**

$$T_A = \frac{V_{out}}{kGB} - T_R$$

$$T_A = AV + B$$



$$\left\{ \begin{array}{l} T_{hot} = \frac{V_{hot}}{kGB} - T_R \\ T_{cold} = \frac{V_{cold}}{kGB} - T_R \end{array} \right.$$

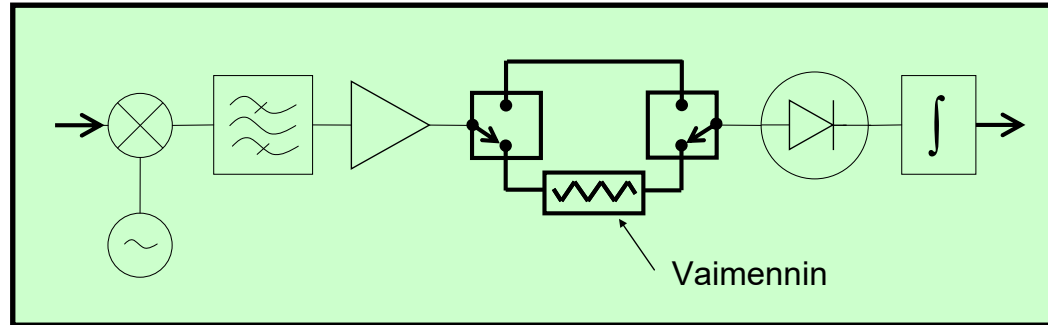
$$A = \frac{T_{hot} - T_{cold}}{V_{hot} - V_{cold}}$$

$$B = \frac{T_{cold}V_{hot} - T_{hot}V_{cold}}{V_{hot} - V_{cold}}$$

Four point calibration

- Utilizes a tunable attenuator at IF
- Compensates for voltage offset v_0

- Four measurements altogether: Two known sources with two IF attenuator values



→ Gain and offset are retrieved without the knowledge of the absolute values of the known temperatures (only the difference matters)!

$$\begin{aligned}
 V_1 &= V_{off} + G(T_{cold} + T_{rec}) \\
 V_2 &= V_{off} + G(T_{hot} + T_{rec}) \\
 V_3 &= V_{off} + \frac{G}{L}(T_{cold} + T_{rec}) \\
 V_4 &= V_{off} + \frac{G}{L}(T_{hot} + T_{rec})
 \end{aligned}$$

$$\begin{aligned}
 V_{off} &= \frac{V_2 V_3 - V_1 V_4}{(V_2 - V_4) - (V_1 - V_3)} \\
 G &= \frac{V_2 - V_1}{T_{hot} - T_{cold}} \\
 T_{rec} &= \frac{a T_{cold} - T_{hot}}{1 - a}, \quad a = \frac{V_2 - V_{off}}{V_1 - V_{off}} \\
 T'_{rec} &= \frac{Y T_{cold} - T_{hot}}{1 - Y}, \quad Y = \frac{V_2}{V_1}
 \end{aligned}$$

Calibration of noise injection radiometer

$$T_A = T_D - \eta T_{NA}$$

$$T_{NA}$$

➤ Only one unknown: only one calibration target ($T_{A,KNOWN}$) required

→
$$T_{NA} = \frac{T_D - T_{A,KNOWN}}{\eta}$$

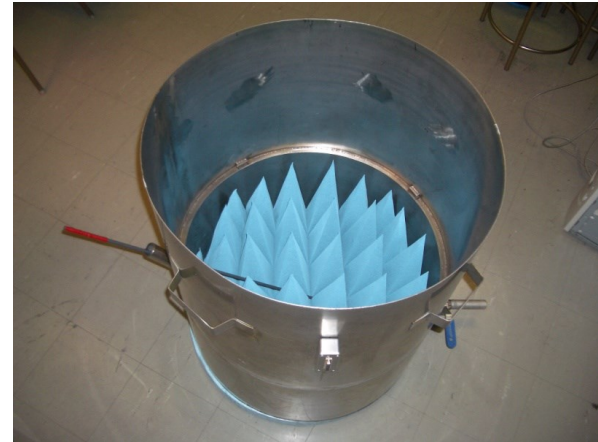
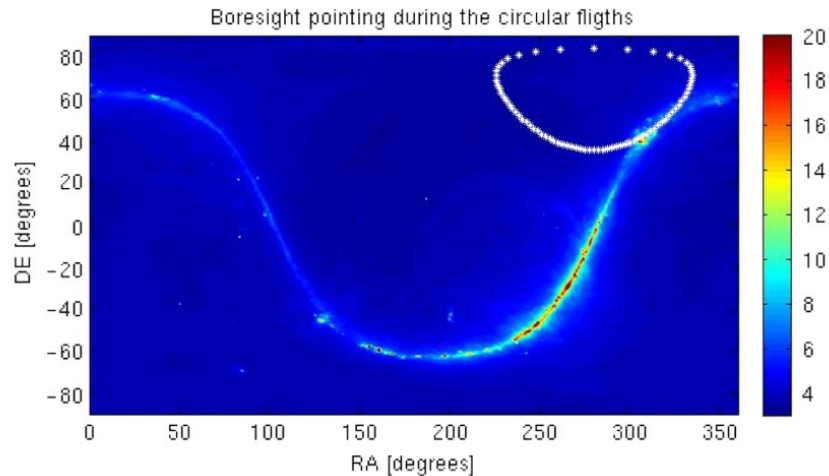
➤ In practice, the stability of T_{NA} (with respect to time/temperature/linearity) becomes a dominant factor.

Radiometer calibration

Calibration sources

Through the antenna:

- High absorption materials ($\epsilon \sim 1$)
- Liquid Nitrogen cooling
- Cky (CMB+atmosphere)
- Water

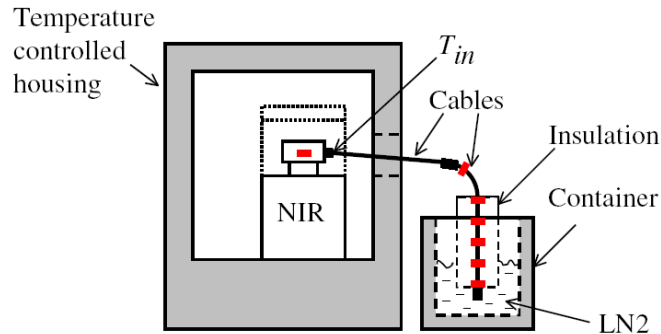
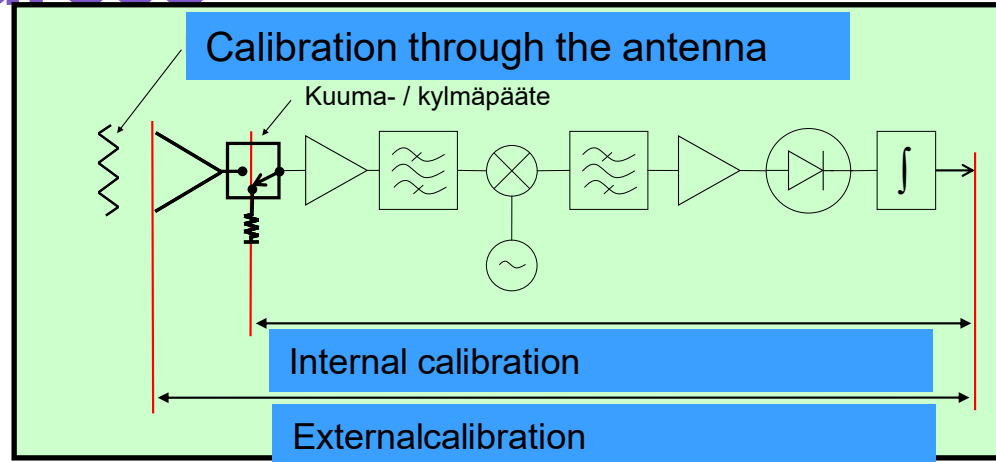


Radiometer calibration

Calibration sources

Without the antenna:

- Terminated loads
- Active loads (cool or hot)
- Liquid nitrogen cooling



Radiometer measurement

$$T_B(\theta) = e(\theta, \varepsilon, \mu, \lambda) T_{fys}$$

$e(\theta)$ = emissivity, $0 \leq e \leq 1$

For homogeneous target brightness temperature

T_{fys} = target physical temperature (K)

θ = incidence angle off nadir

Radiometer measured temp
called antenna temperature T_A

$$T_A = \frac{\iint T_B(\theta, \phi) F_n(\theta, \phi) d\Omega}{\iint_{4\pi} F_n(\theta, \phi) d\Omega}$$

F_n = normalized antenna power pattern (value between 0 and 1)

Calibration the entire system

Measure a target with radiometer => antenna effects are included in calibration

Method 1: Measure with radiometer so called calibration targets

- Absorbing material, whose emissivity = 1 ("blackbody")
 - Then brightness temperature = physical temperature
- Use two calibration targets
 - Hot load: high brightness temperature ($T_B \sim 290\text{ K}$)
 - Cold load: low brightness temperature ($T_B < 100\text{ K}$)
=> If radiometer is linear, region 100...290 K is calibrated

Method 2: Measure with radiometer natural targets, whose T_B at desired frequencies, polarizations and incidence angles is known

- Calm water surface: T_B can be calculated accurately
- Sky (no clouds, normal humidity, avoiding known radio emitters)
- This approach is only used as an additional method to make sure that calibration based on Method 1 is OK

Fundamental restrictions

Receiver

- The amount of radiation collected by the antenna is **VERY small in powers** (typically in the order of 10^{-13} – 10^{-14} Watts). In order to detect the signal level amplification is needed.
- Typically, square-law detectors are used. (Output voltage is linear with input noise power)
- The bandwidth under measurement must be well known for the sake of
 - 1) Power control
 - 2) Frequency regulations
 - 3) Interference control



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Antennas and scanning

Chapter 6 in the course book

SMOS antenna



Antenna properties

Beam Efficiency – main lobe related to side lobe

$$\eta_b = \frac{\Omega_m}{\Omega_p} .$$

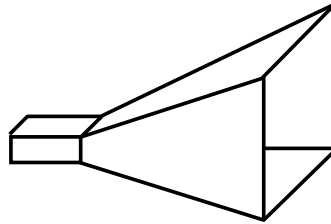
Radiation Efficiency – how much losses are in antenna

$$T_A = \xi \eta_b T_{ML} + \xi (1 - \eta_b) T_{SL} + (1 - \xi) T_0 .$$



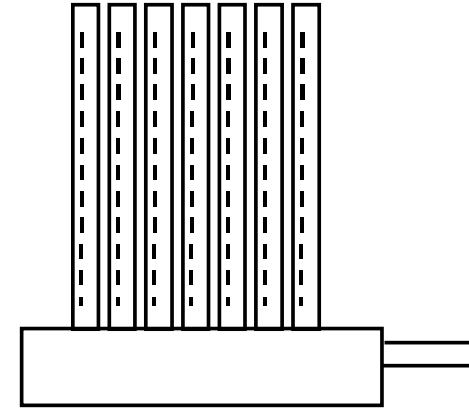
Used antenna types

Horn antenna: Often used onboard satellite as feed antenna for a paraboloid antenna



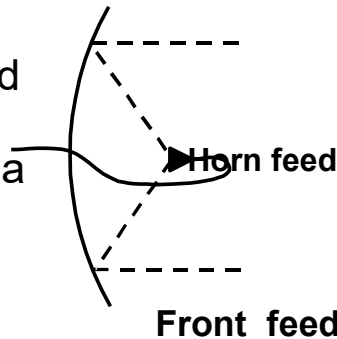
Horn antenna

Phased array: Many elements, each equipped with a phase shifter => narrow beam, scanning antenna

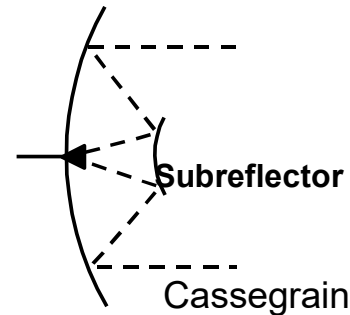


Phased array

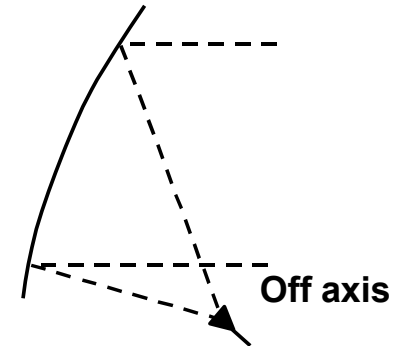
Paraboloid: Cassegrain feed mostly used (good cross-polarization properties with a corrugated horn)



Front feed



Cassegrain



Off axis

Paraboloid antennas with various feed systems

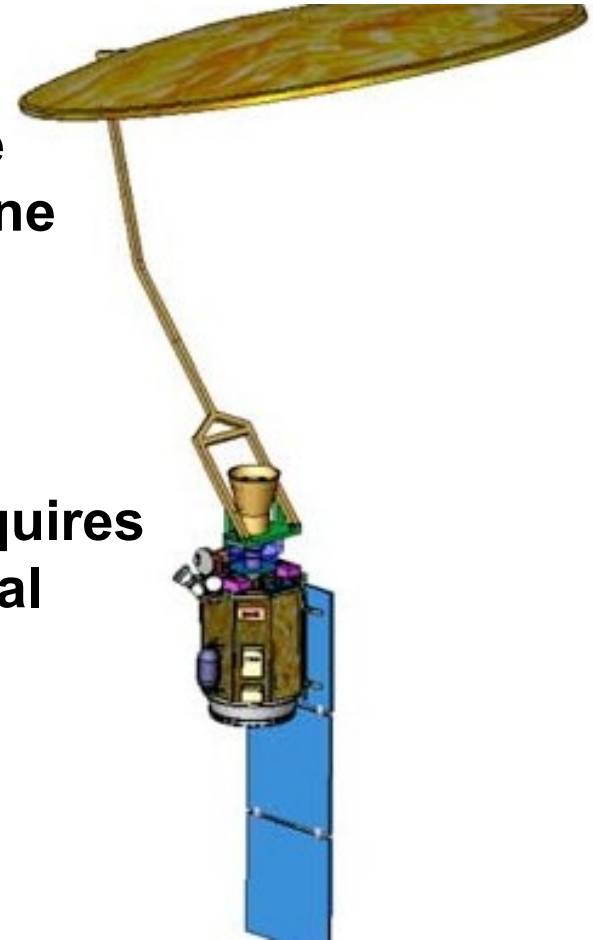
Antenna opening angle and resolution

Opening angle of the antenna determines the angular resolution. For parabolic antennas one can estimate:

$$\theta_{3dB} \approx 1.4 \frac{\lambda}{D}$$

At microwaves, even mediocre resolution requires antennas in size of tens centimeters to several meters!

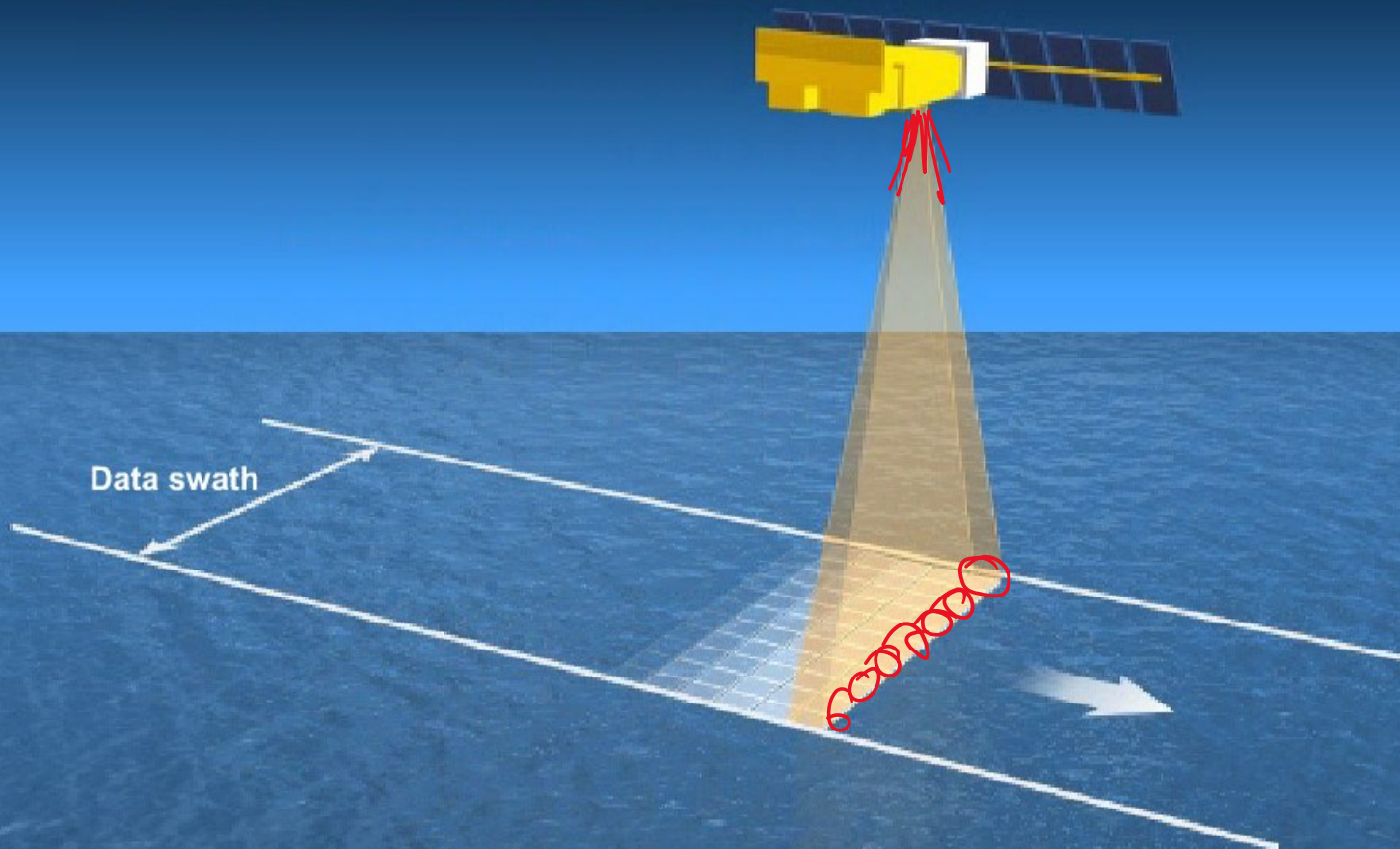
How to form an image?



Scanning methods in satellite radiometry

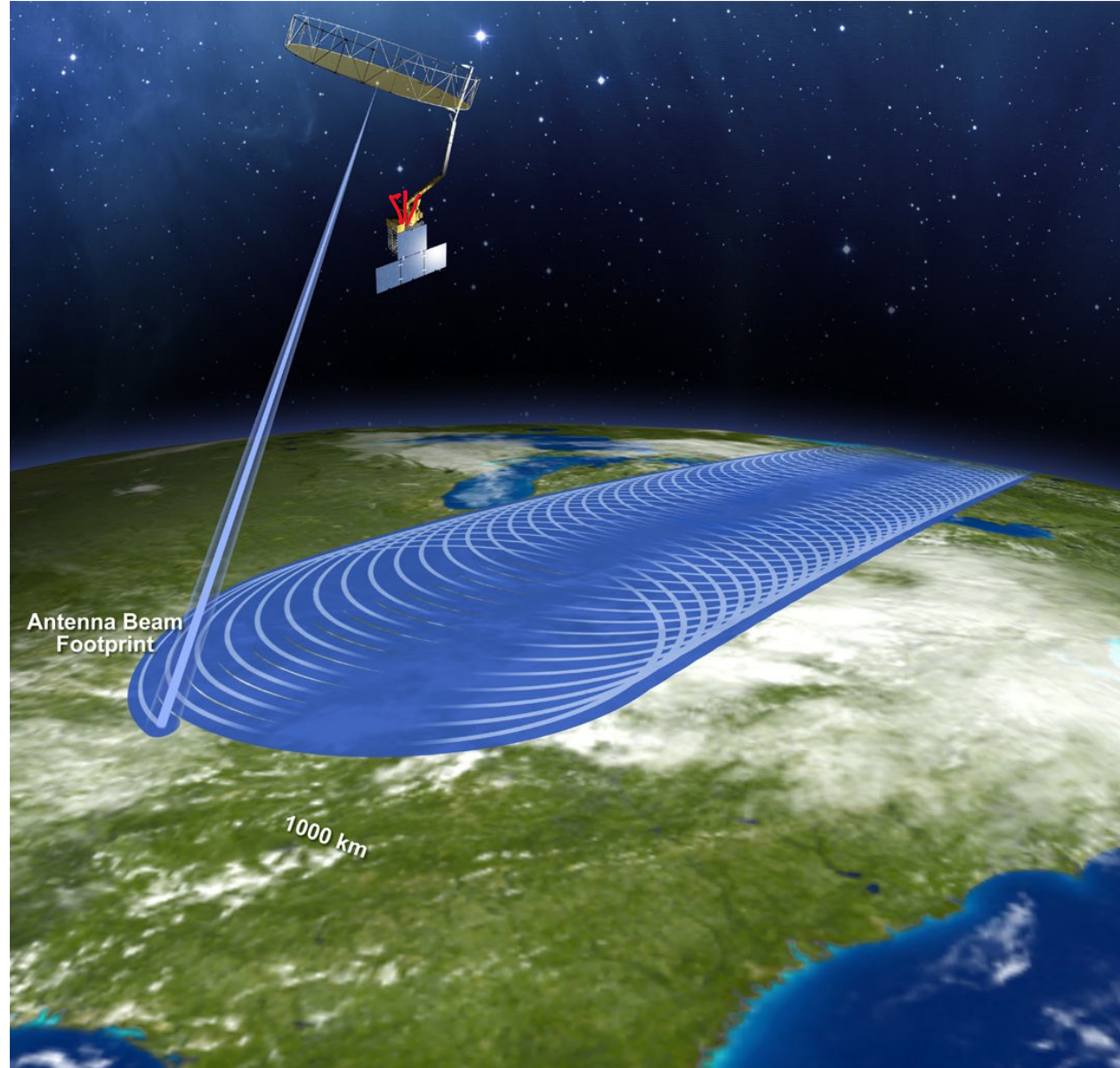
- **Conical scanning:** antenna moves as along the surface of a cone, eg. at 50° incidence angle off nadir
- **Pushbroom techniques:** several antennas producing beams next to each other either along flight direction or perpendicular to it
 - *Several receivers*
- **Interferometry:** Image formation without mechanical/electrical scanning by correlating outputs from all antenna pairs
 - *Several antennas with individual receivers (several receivers)*

Pushbroom scanning



Conical scanning

SMAP



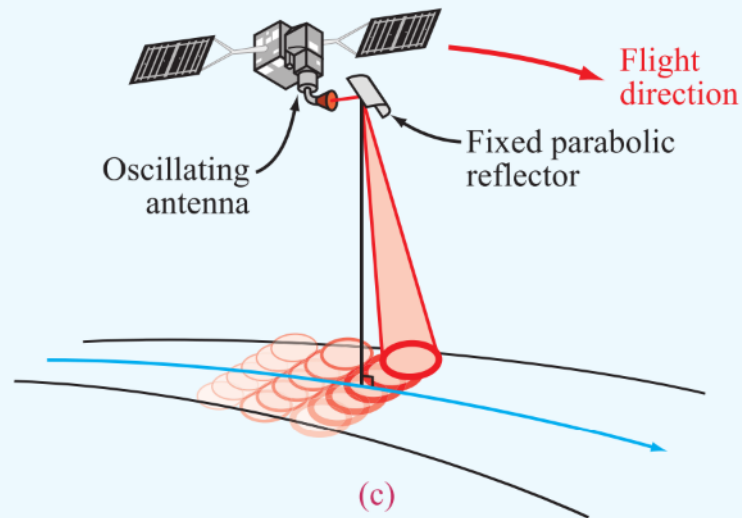
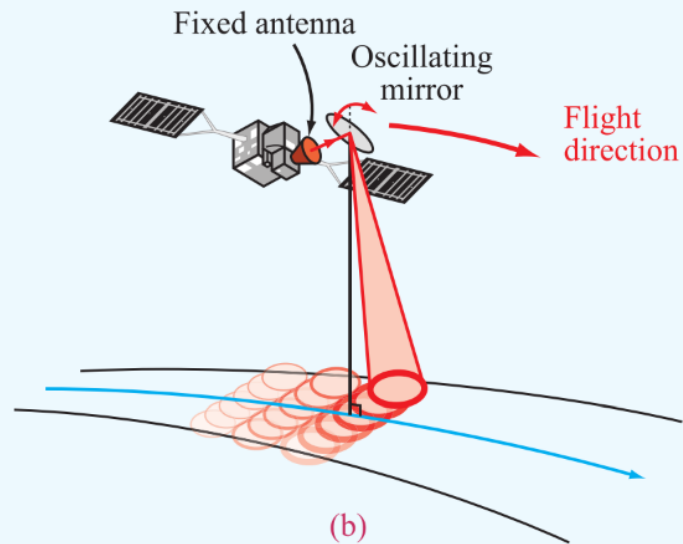
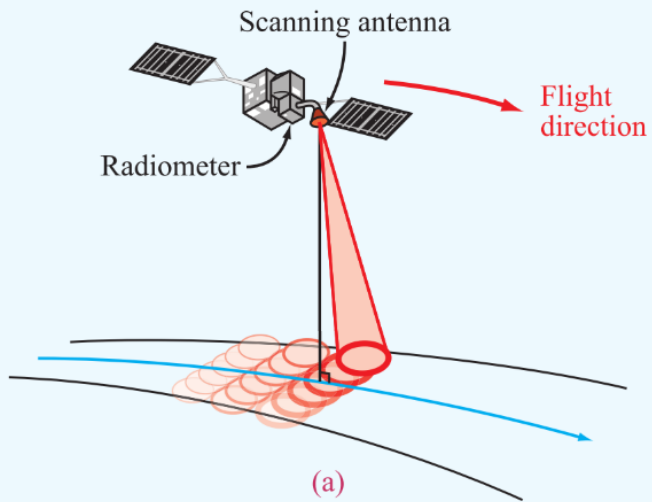


Figure 7-31: Mechanical scanning configurations: (a) scanning antenna; (b) fixed antenna and oscillating reflector; (c) fixed parabolic reflector and oscillating antenna feed.

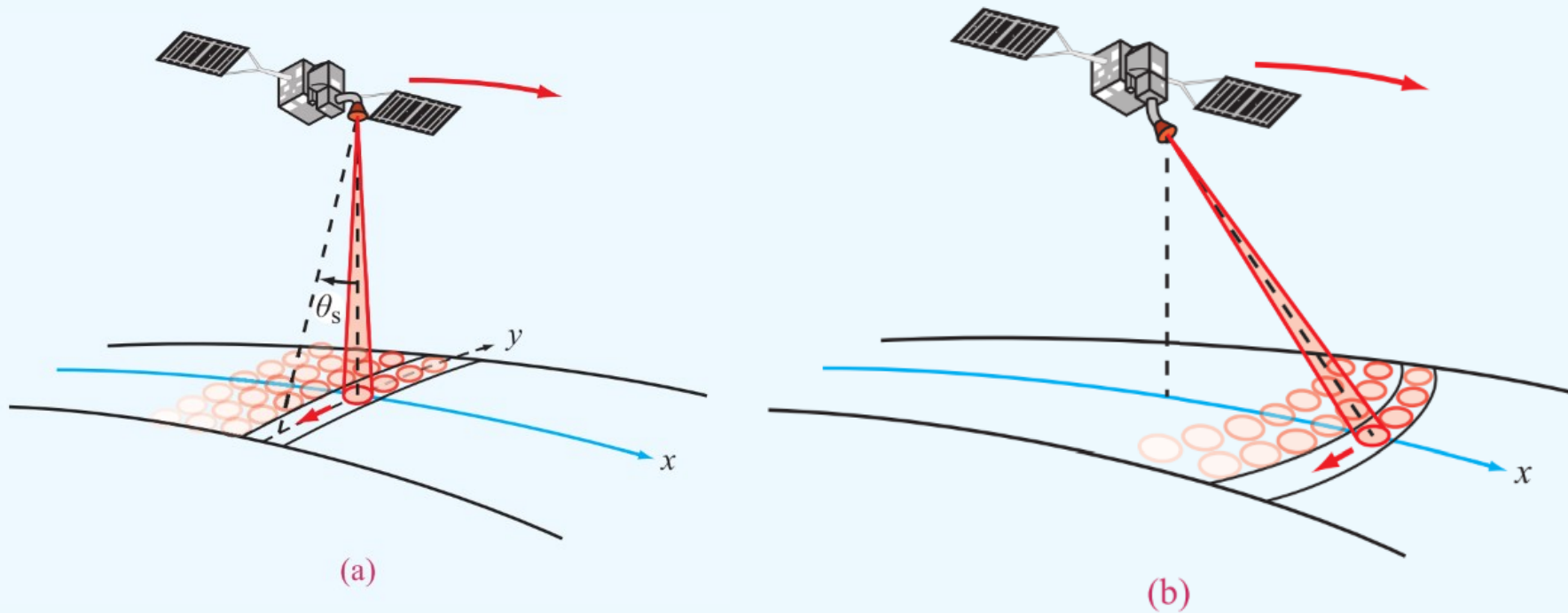


Figure 7-33: Radiometric imaging by (a) cross-track scanning in the plane normal to the direction of flight, and (b) conical scanning.



RBI - Cross-Track Scanning Radiation Budget Instrument



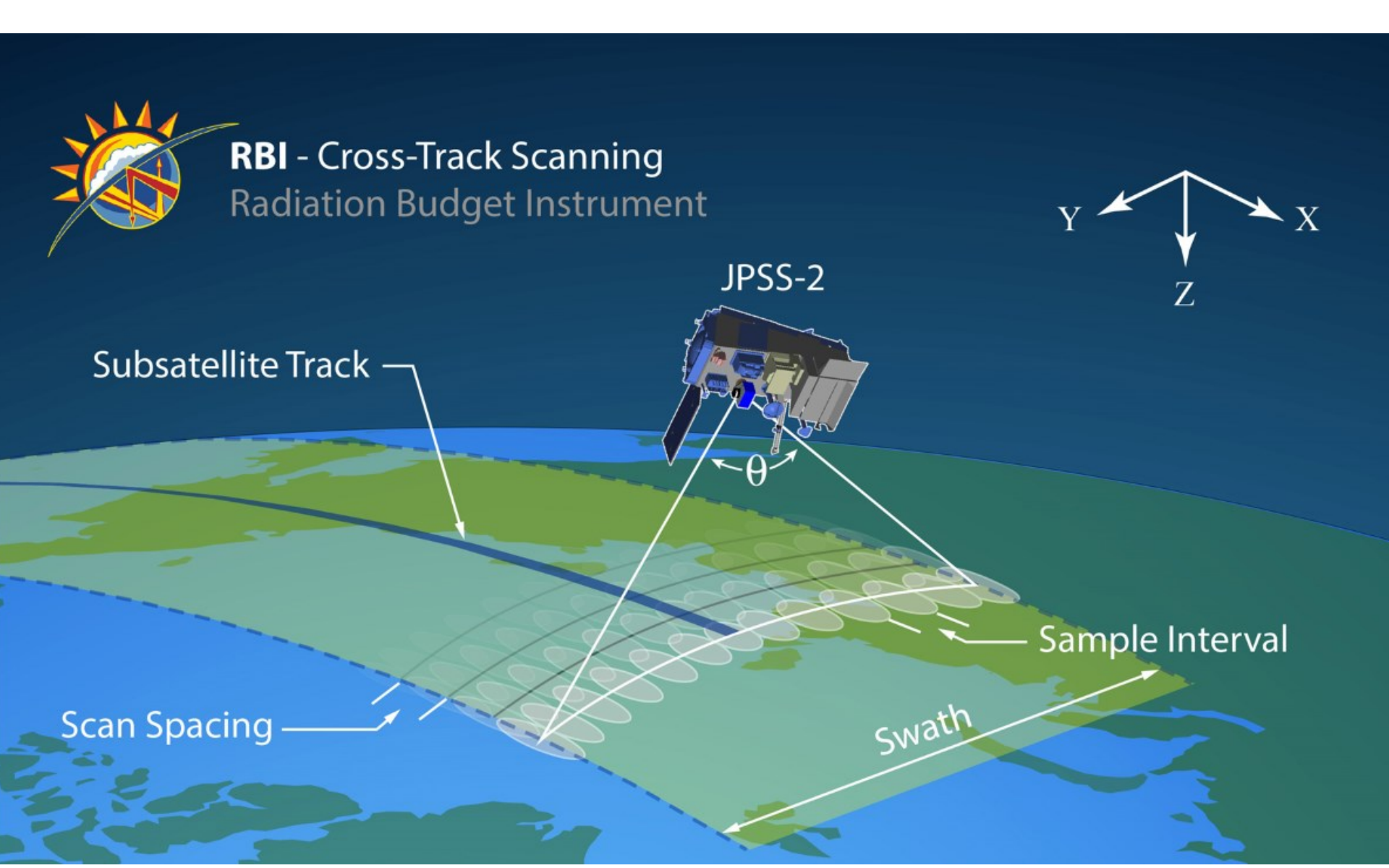
JPSS-2

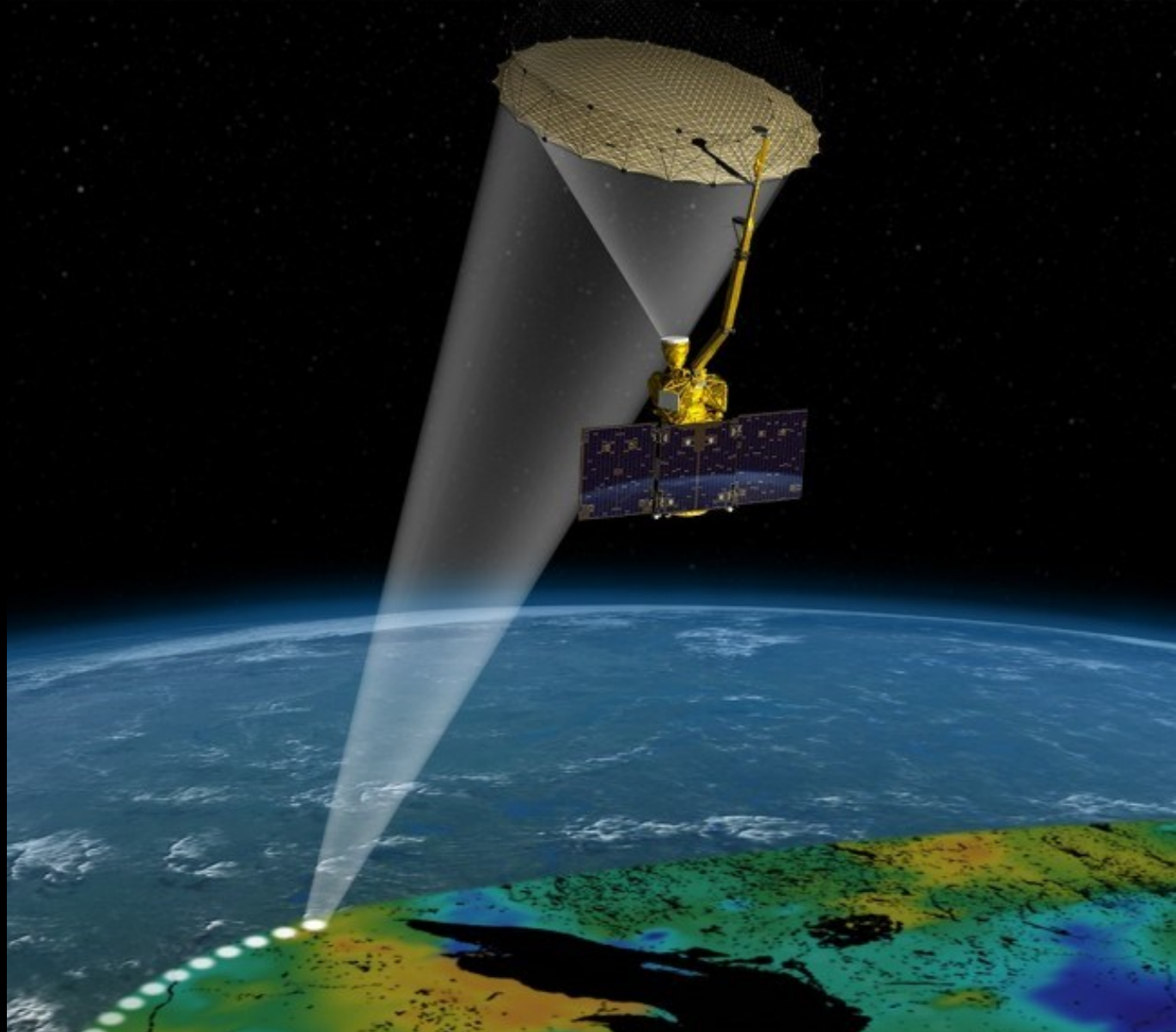
Subsatellite Track

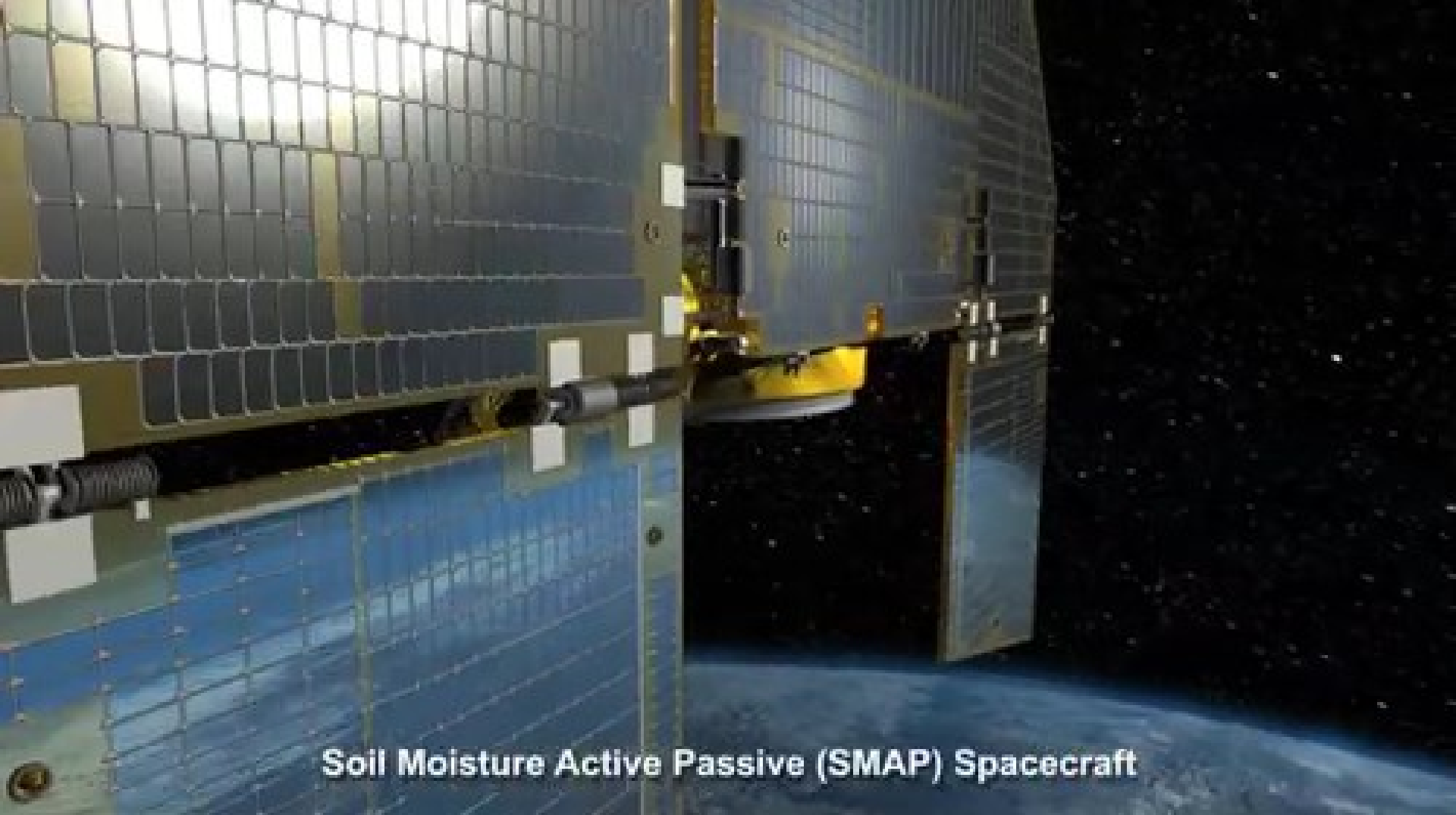
Scan Spacing

Sample Interval

Swath







Soil Moisture Active Passive (SMAP) Spacecraft



**Time-Resolved Observations of
Precipitation structure and storm
Intensity with a Constellation of Smallsats**

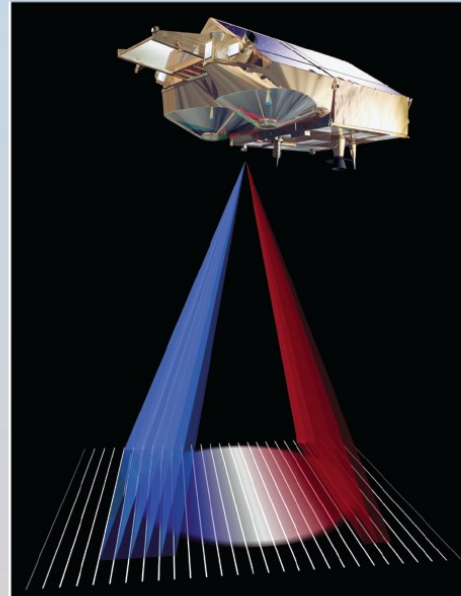
MIT Lincoln Laboratory (proposing organization)

William J. Blackwell, Principal Investigator; Scott Braun (NASA GSFC), Project Scientist

A constellation of identical 3U CubeSats provide sounding (left CubeSat has a temperature profile of a simulated Tropical Cyclone (TC) from a numerical weather prediction (NWP) model) and 12-channel radiometric imagery (center CubeSat has simulated radiances from NWP model and radiative transfer model and the near right CubeSat has a single-channel radiance image of a TC) with a median revisit rate approaching 30 minutes to meet most PATH requirements.

- Ulaby
- Long
- Blackwell
- Elachi
- Fung
- Ruf
- Sarabandi
- Zebker
- Van Zyl

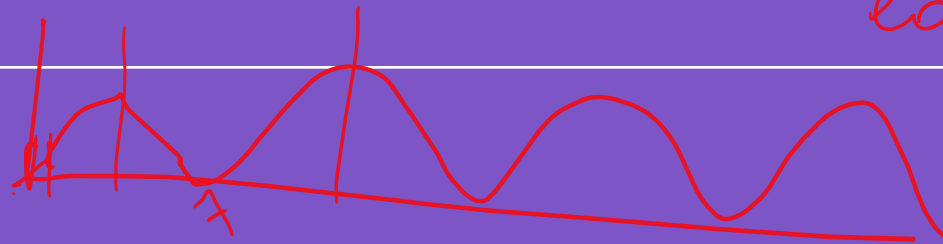
Microwave Radar and Radiometric Remote Sensing



These PowerPoint slides are intended for educational use. They should not be used for sale or financial profit.

END

$$E = \frac{1}{r} e^{-i(\omega t + kx)}$$



Effect of Antenna in Radiometry

Definition of antenna (radiometric) temperature

- Atmosphere ignored

$$T_A = \frac{\iint T_B(\theta, \phi) F_n(\theta, \phi) d\Omega}{4\pi \iint F_n(\theta, \phi) d\Omega}$$

Effect of main lobe and side lobes*

$T_A = \text{term 1} + \text{term 2}$

$$T_A = \frac{\iint_{\text{main lobe}} T_B(\theta, \phi) F_n(\theta, \phi) d\Omega}{\iint_{4\pi} F_n(\theta, \phi) d\Omega} + \frac{\iint_{\text{side lobes}} T_B(\theta, \phi) F_n(\theta, \phi) d\Omega}{\iint_{4\pi} F_n(\theta, \phi) d\Omega}$$

We define:

Main-lobe antenna temperature T_{ML} :

(actually: effective apparent temperature of the main lobe contribution)

$$T_{ML} = \frac{\iint_{\text{main lobe}} T_B(\theta, \phi) F_n(\theta, \phi) d\Omega}{\iint_{\text{main lobe}} F_n(\theta, \phi) d\Omega}$$

Side-lobe antenna temperature T_{SL} :

(actually: effective apparent temperature of the side-lobe contribution)

$$T_{SL} = \frac{\iint_{\text{side lobes}} T_B(\theta, \phi) F_n(\theta, \phi) d\Omega}{\iint_{\text{side lobes}} F_n(\theta, \phi) d\Omega}$$

Effect of Antenna in Radiometry

We recall:

$$\eta_b = \frac{\Omega_m}{\Omega_p}$$

\Rightarrow *Term 1 = $\eta_b T_{ML}$ and Term 2 = $\eta_m T_{SL}$

Antenna temperature:

- Brightness temperature modified by lossless antenna

$$T_A = \eta_b T_{ML} + (1 - \eta_b) T_{SL}$$

LOSSY ANTENNA:

Radiation emitted by passive component (antenna)

- L = antenna attenuation
- ξ = antenna radiation efficiency
- T_0 = antenna physical temperature

$$T_N = \left(1 - \frac{1}{L}\right) T_0 = (1 - \xi) T_0$$

Noise power arriving at the receiver (attenuated T_A + antenna emission)

$$T_A = \xi T_A' + (1 - \xi) T_0$$

Final antenna temperature (by employing T_A from above)

$$T_A = \xi \eta_b T_{ML} + \xi (1 - \eta_b) T_{SL} + (1 - \xi) T_0$$

Radiometer Measurement Ambiguity

Introduced mainly by sidelobes

$$T_{\text{ML}} = \left(\frac{1}{\xi \eta_b}\right) T_A - \left(\frac{1 - \eta_b}{\eta_b}\right) T_{\text{SL}} - \left(\frac{1 - \xi}{\xi \eta_b}\right) T_0. \quad (6.42)$$

Sidelobe factor

$$T_{\text{ML}} = aT_A + b, \quad (6.43)$$

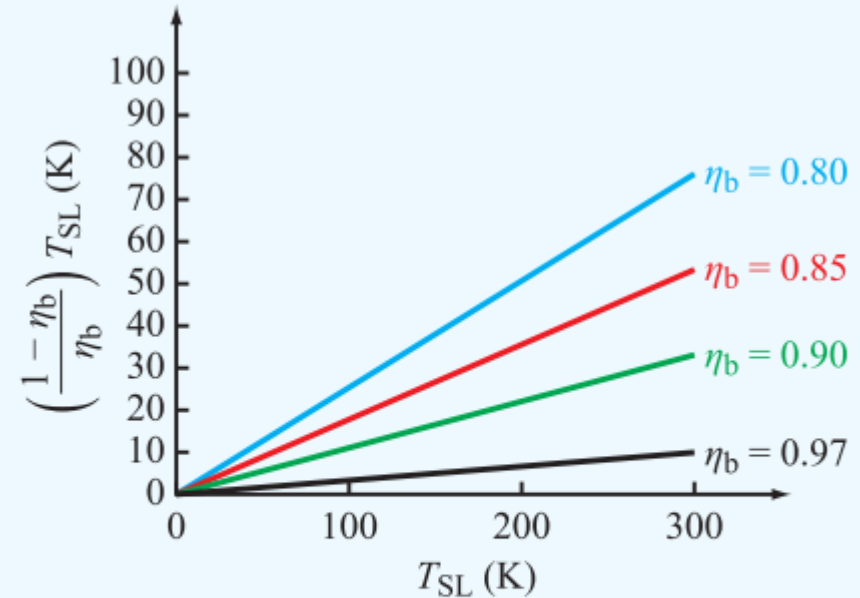
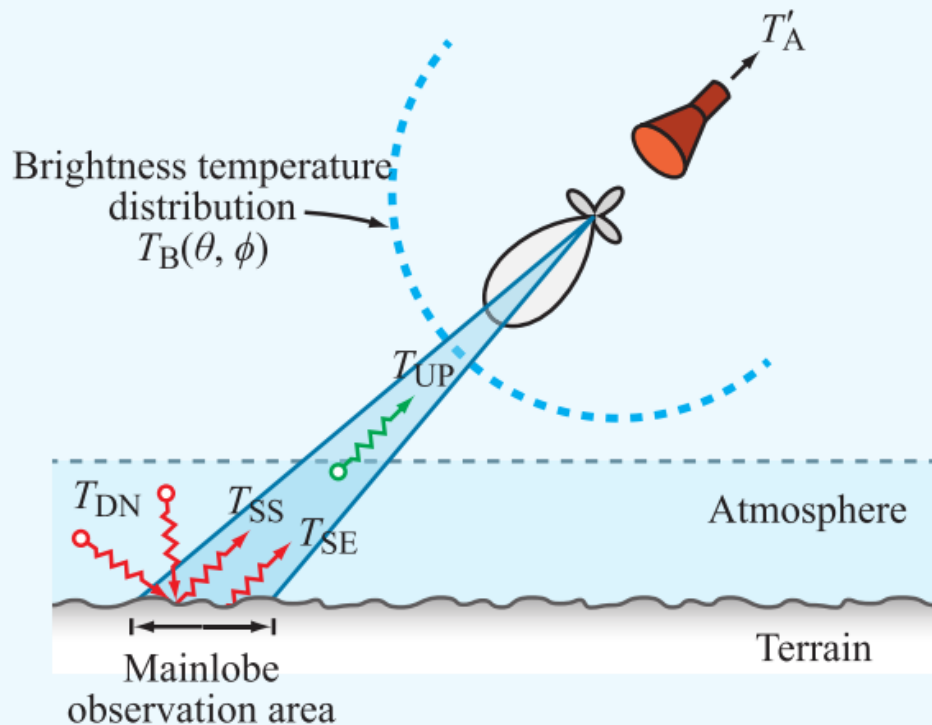
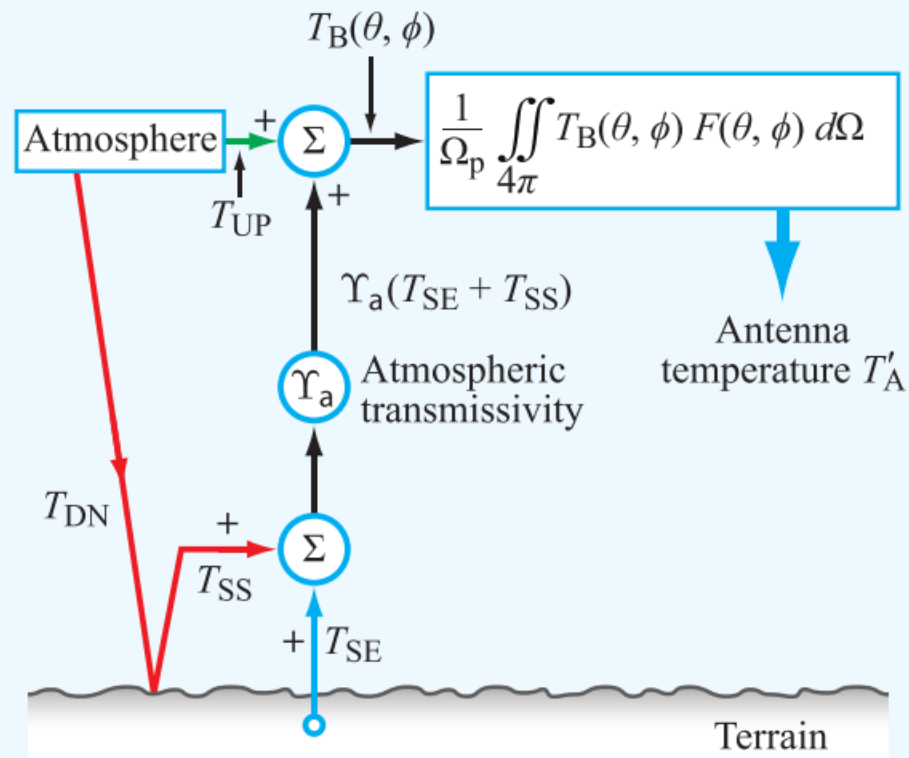


Figure 6-10: Sidelobe factor as a function of the incident sidelobe brightness temperature T_{SL} for each of several values of the beam efficiency η_b .

T_{SE} = Surface emission
 T_{DN} = Atmospheric downward emission
 T_{UP} = Atmospheric upward emission
 T_{SS} = Surface scattered radiation



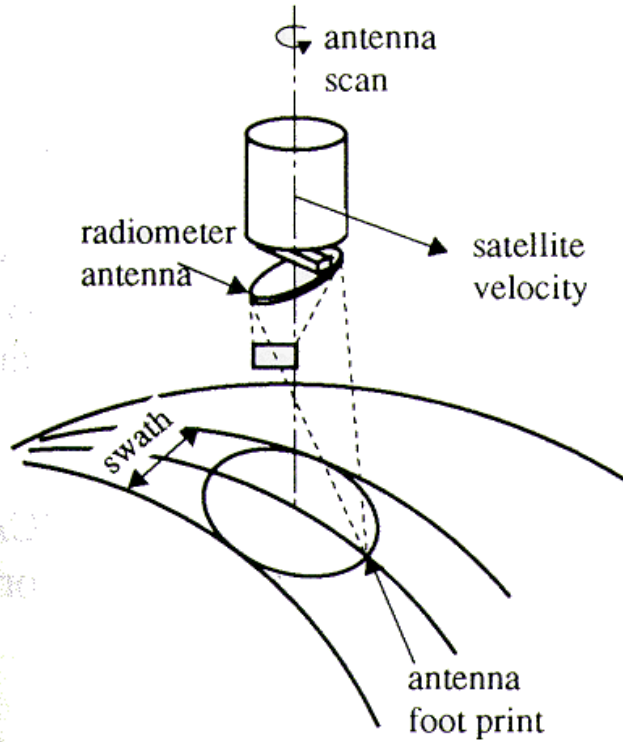
(a) Emission by surface and atmosphere



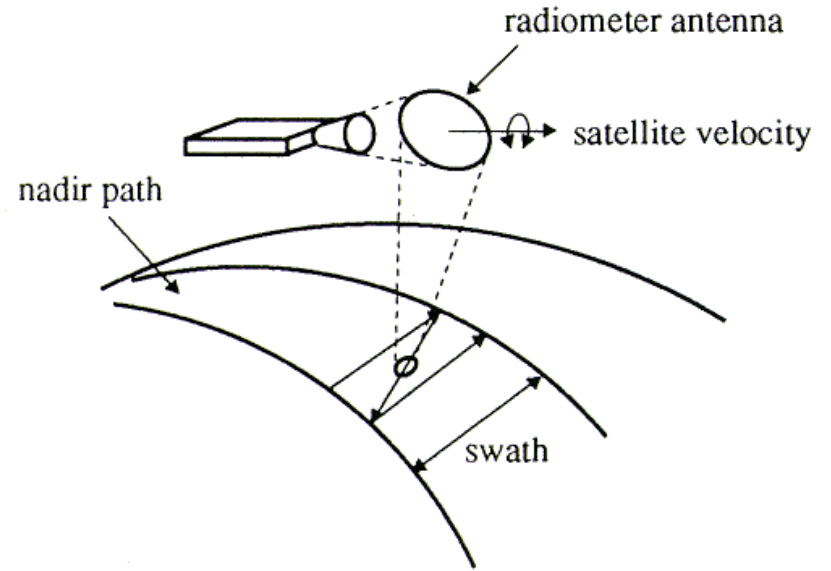
(b) Block diagram

TB brightness temperature distribution

Scanning

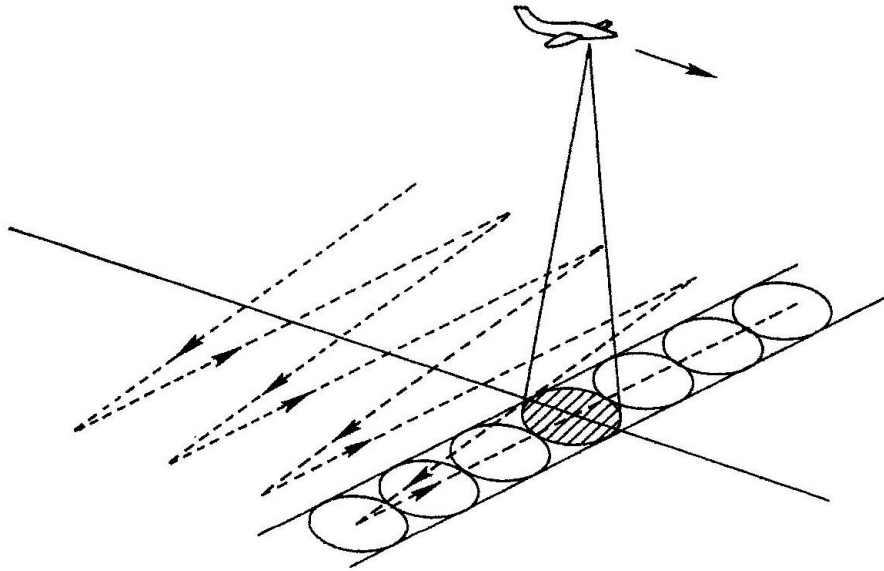


(a) conical scanning type

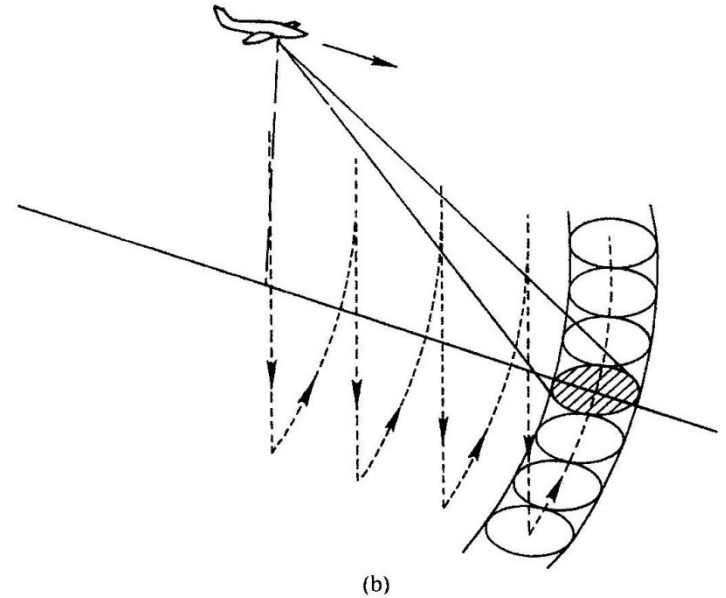


(b) cross-track scanning type

Scanning

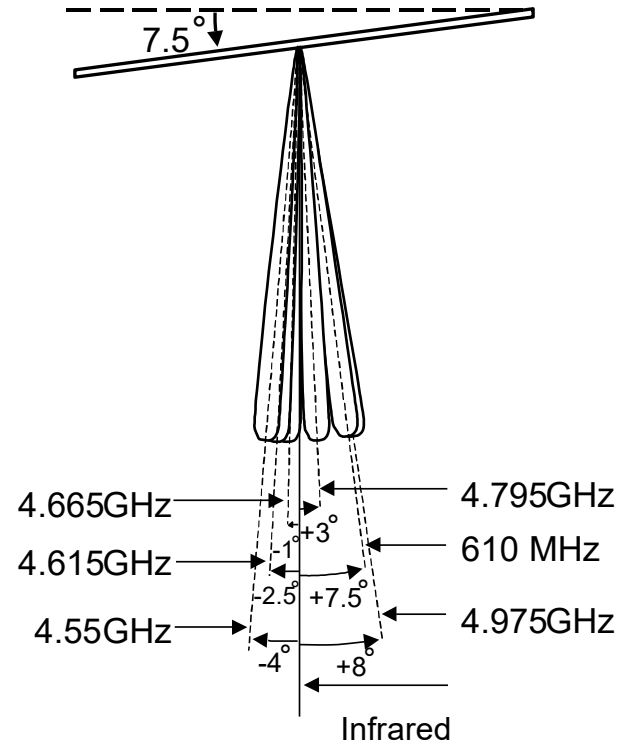
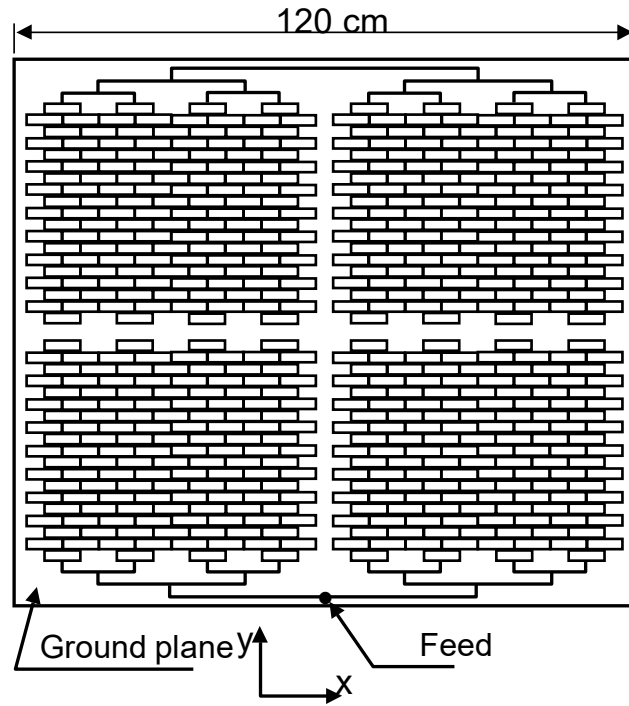


Optical scanner (looking down)
Polarization and incidence angle vary



Conical scanning (looking forward)
Polarization and incidence angle
preserved from pixel to pixel

Example: airborne 5 GHz antenna



- Printed circuit technology
- By choosing antenna dimensions vs. frequency properly the main beam direction depends on frequency as shown, swath at 610 MHz is equal to that at 4.5 to 5 GHz

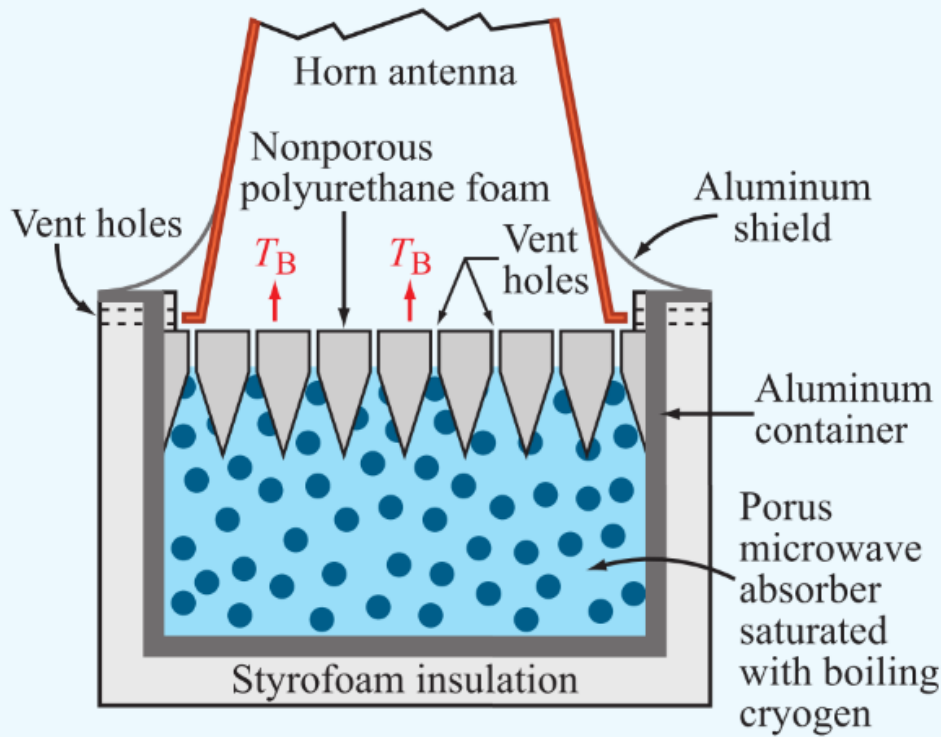


Figure 7-27: Construction of cryoload for calibration of radiometer antenna [after Hardy et al., 1974].

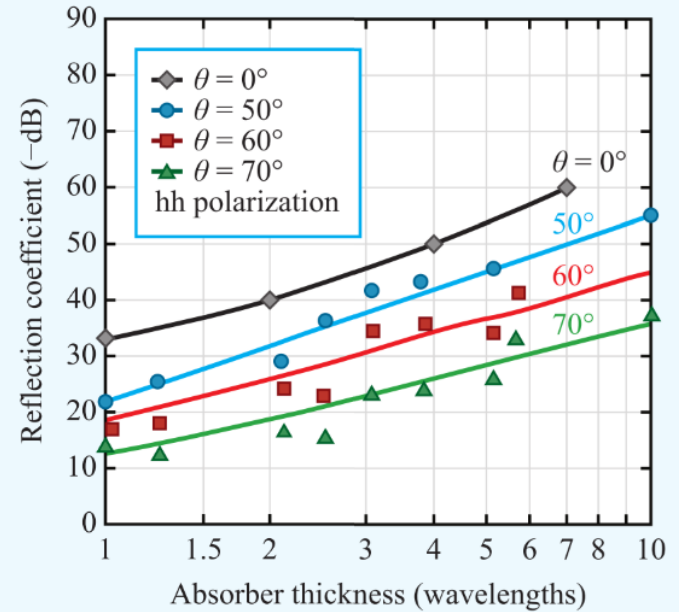


Figure 7-28: Characteristics of RF absorbers [after Emerson, 1973].

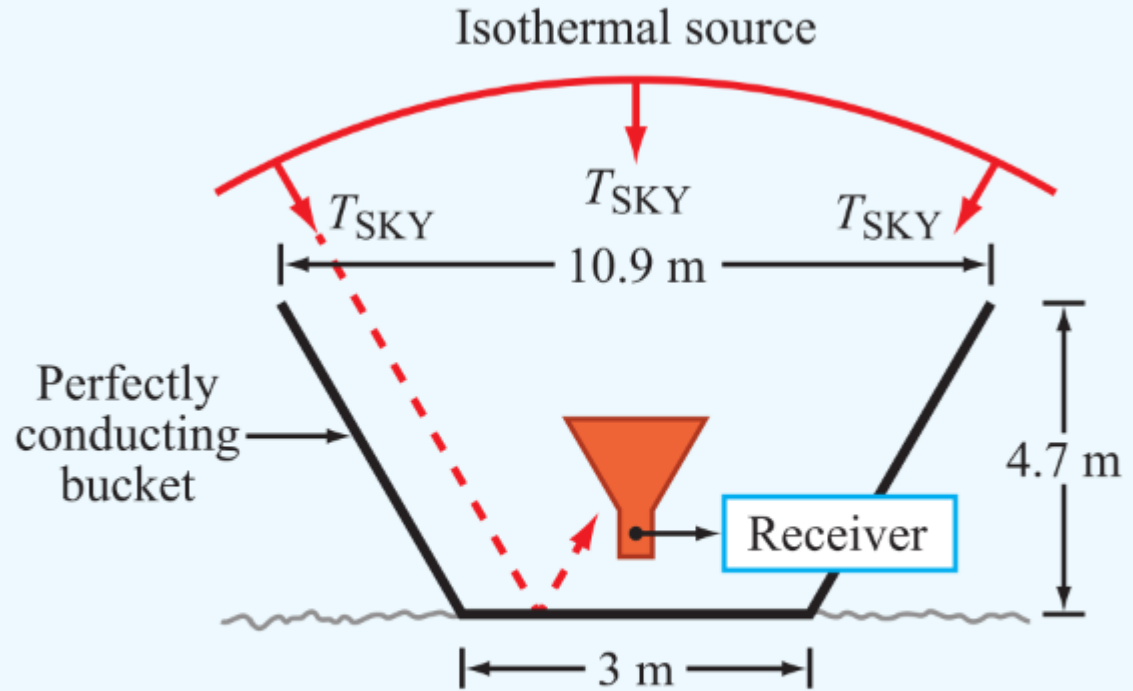


Figure 7-29: The bucket method for measuring the radiation efficiency of an antenna [after Carver, 1975].

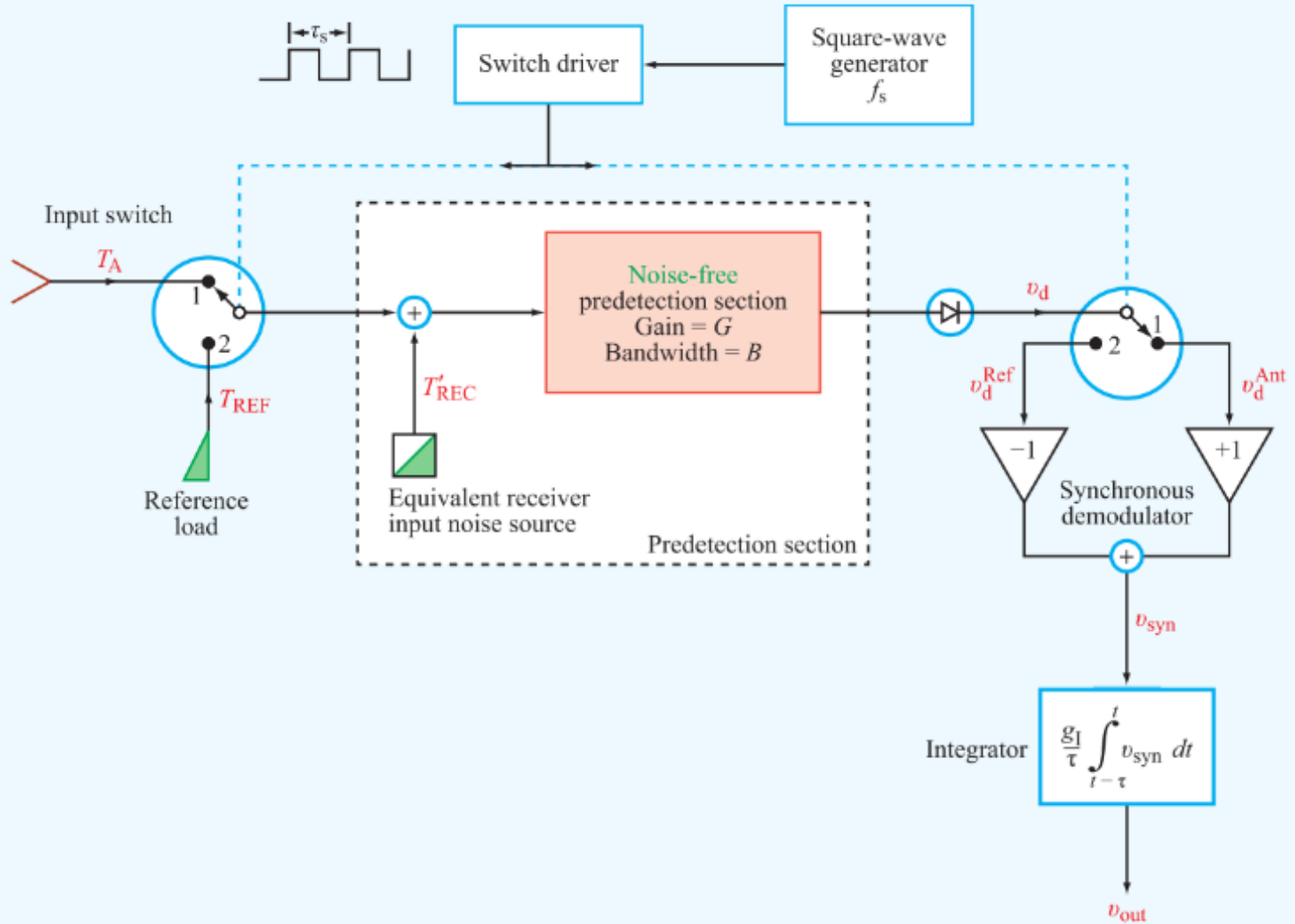


Figure 7-16: Functional block diagram of a Dicke radiometer.

From: Microwave Radar and Radiometric Remote Sensing, by Ulaby and Long, 2014, with permission.

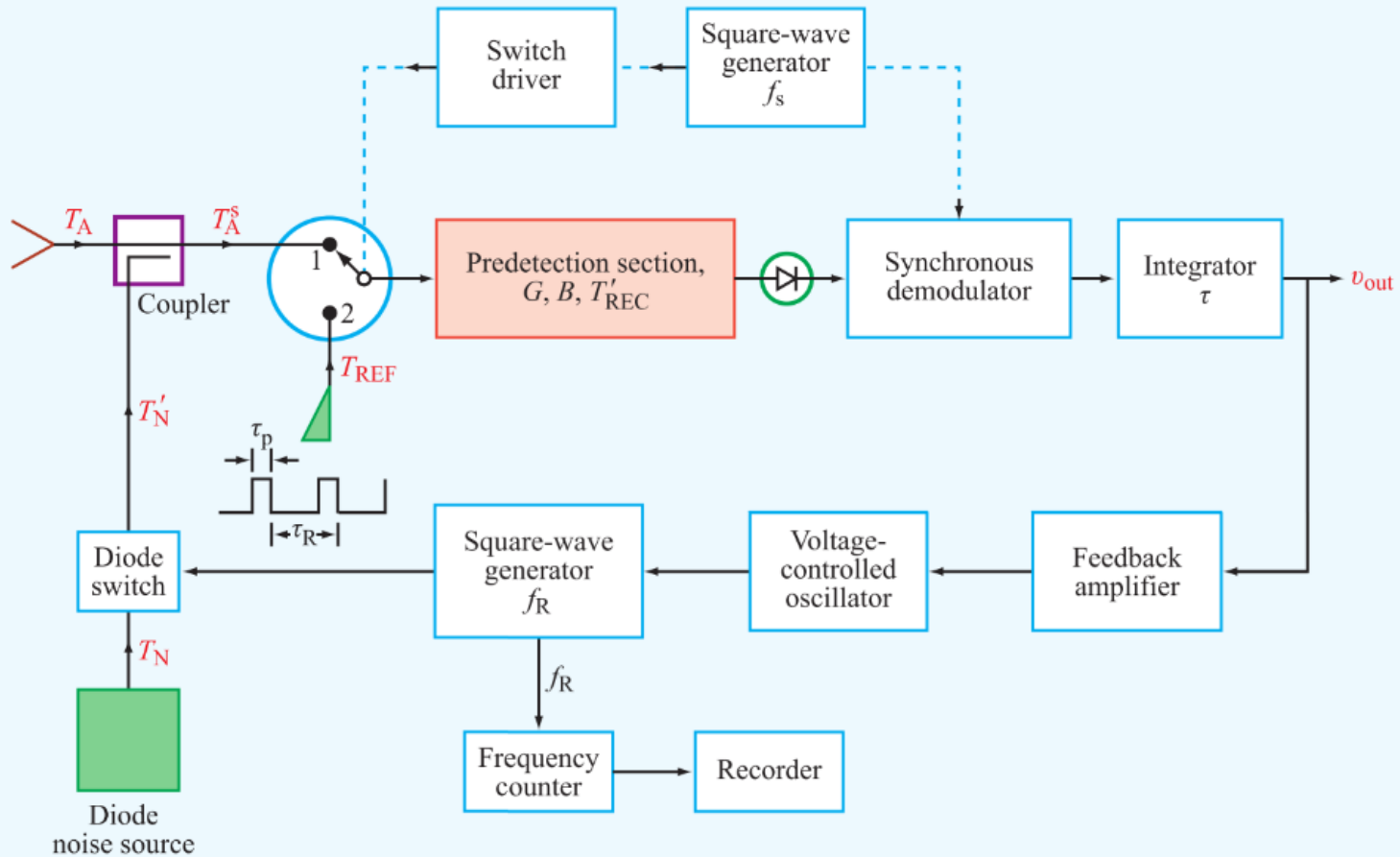


Figure 7-19: Balanced Dicke radiometer, using pulsed noise-injection to maintain $T_A^s = T_{REF}$. The output indicator of T_A is the pulse repetition frequency f_R .