

Aalto University School of Electrical Engineering

Waves in fluids and rectangular enclosures

ELEC-E5610 Acoustics and the Physics of Sound, Lecture 3

Georg Götz

Department of Information and Communications Engineering Aalto University School of Electrical Engineering

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Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

Fluids consist of *"particles that easily move and change their relative position without a separation of the mass and that easily yield to pressure."* (Merriam-Webster)



Fluids

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in practice, gases and liquids



Fluids

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- Wave Equation
- Speed of sound

Fluids consist of *"particles that easily move and change their"* relative position without a separation of the mass and that easily yield to pressure." (Merriam-Webster)

- in practice, gases and liquids
- gases
 - fill all the volume they're enclosed in
 - negligible particle interaction unless they collide
 - the relative deformations of volume and pressure have approximately the same magnitude



Fluids

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- in practice, gases and liquids
- gases
 - fill all the volume they're enclosed in
 - negligible particle interaction unless they collide
 - the relative deformations of volume and pressure have approximately the same magnitude
- liquids
 - certain volume in a certain temperature
 - strong interaction between particles
 - the relative deformation of volume is smaller than the deformation of pressure



The *ideal fluid* is a simplification of real fluids.

Fluids
Ideal Fluid
Ideal Gas
Continuums
Wave Equation
Speed of sound



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The *ideal fluid* is a simplification of real fluids. In particular, it possesses some interesting properties:

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boundary plate (2D, stationary)

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Bose-Einstein condensates (a.k.a *superfluids*, e. g. Helium below $2.2K^{\circ}$) fully act as ideal fluids (http://www.youtube.com/watch?v=2Z6UJbwxBZI).



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boundary plate (2D, stationary)

1 Ideal Gas

Fluids

Ideal Fluid

Ideal Gas

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Wave Equation

Speed of sound

The *ideal gas* is an ideal fluid that obeys the ideal gas law:

$$\frac{pV}{T} = \text{constant},$$

where p is pressure, V is volume, and T is temperature.



1 Ideal Gas

Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

The *ideal gas* is an ideal fluid that obeys the ideal gas law:

$$\frac{pV}{T} = \text{constant},$$

where p is pressure, V is volume, and T is temperature.

Although the ideal gas is an idealization, it is a **good approximation** to the behavior of many gases under many conditions.

Remember that pressure in gases arises from random molecular motion, it is scalar.



Fluids

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Wave Equation

Speed of sound

according to our current understanding, all matter consists of particles



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Fluids

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Speed of sound

according to our current understanding, all matter consists of particles

in theory, sound propagation laws could be derived from the "averaged" conservation laws of mass, momentum, and energy for these particles



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- Fluids
- Ideal Fluid
- Ideal Gas
- Continuums
- Wave Equation
- Speed of sound

- according to our current understanding, all matter consists of particles
- in theory, sound propagation laws could be derived from the "averaged" conservation laws of mass, momentum, and energy for these particles
- however, a more convenient model can be obtained by considering the medium as a continuum
 - \blacksquare \Rightarrow forget the molecular structure of the fluid.



Fluids

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Speed of sound

 according to our current understanding, all matter consists of particles

 in theory, sound propagation laws could be derived from the "averaged" conservation laws of mass, momentum, and energy for these particles

however, a more convenient model can be obtained by considering the medium as a *continuum*

■ ⇒ forget the molecular structure of the fluid. Valid assumption, if

$$Kn = rac{\Lambda}{L} \ll 1,$$

where Kn is Knudsen's number, Λ is the average free space between molecular collisions, and L is the length in the dimension of observation.



Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

- \blacksquare in the standard atmosphere $\Lambda=6\times 10^{-8}m$
- for sound fields $L = \lambda$



Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

 \blacksquare in the standard atmosphere $\Lambda=6\times 10^{-8}m$

- for sound fields $L = \lambda$
- can the air be considered continuous for audio frequencies?



Fluids

Ideal Fluid

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Continuums

Wave Equation

Speed of sound

• in the standard atmosphere $\Lambda=6\times 10^{-8} m$

- for sound fields $L = \lambda$
 - can the air be considered continuous for audio frequencies?
 - upper limit of audio frequencies *f* = 20 kHz

$$Kn = \frac{\Lambda}{L}$$



Fluids

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Wave Equation

Speed of sound

• in the standard atmosphere $\Lambda=6\times 10^{-8} m$

- for sound fields $L = \lambda$
 - can the air be considered continuous for audio frequencies?

• upper limit of audio frequencies f = 20 kHz

$$\Rightarrow \lambda = \frac{c}{2 \times 10^4}$$
$$Kn = \frac{\Lambda}{L}$$



Fluids

Ideal Fluid

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Wave Equation

Speed of sound

- in the standard atmosphere $\Lambda=6\times 10^{-8} m$
- for sound fields $L = \lambda$
 - can the air be considered continuous for audio frequencies?

• upper limit of audio frequencies f = 20 kHz

$$\Rightarrow \lambda = rac{344}{2 imes 10^4} pprox 0.0172 \,\mathrm{m}$$

$$Kn = \frac{\Lambda}{L} = \frac{6 \times 10^{-8}}{0.0172}$$



Fluids

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Continuums

Wave Equation

Speed of sound

- in the standard atmosphere $\Lambda=6\times 10^{-8} m$
- for sound fields $L = \lambda$
 - can the air be considered continuous for audio frequencies?
 - upper limit of audio frequencies f = 20 kHz $\Rightarrow \lambda = \frac{344}{2\times 10^4} \approx 0.0172$ m

$$Kn = \frac{\Lambda}{L} = \frac{6 \times 10^{-8}}{0.0172} \approx 3.5 \times 10^{-6}$$



Fluids

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Wave Equation

Speed of sound

- in the standard atmosphere $\Lambda = 6 \times 10^{-8}$ m
- for sound fields $L = \lambda$
 - can the air be considered continuous for audio frequencies?
 - upper limit of audio frequencies f = 20 kHz $\Rightarrow \lambda = \frac{344}{2 \times 424} \approx 0.0172 \text{ m}$

$$Kn = \frac{\Lambda}{L} = \frac{6 \times 10^{-8}}{0.0172} \approx 3.5 \times 10^{-6} \ll$$



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Wave Equation

Speed of sound

- in the standard atmosphere $\Lambda=6\times 10^{-8} m$
- for sound fields $L = \lambda$
 - can the air be considered continuous for audio frequencies?
 - upper limit of audio frequencies f = 20 kHz $\Rightarrow \lambda = \frac{344}{2 \times 10^4} \approx 0.0172$ m

$$Kn = \frac{\Lambda}{L} = \frac{6 \times 10^{-8}}{0.0172} \approx 3.5 \times 10^{-6} \ll 1$$

 \Rightarrow yes, air can be considered continuous for audio frequencies!



1 Disturbance

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Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

Sound is a disturbance of air density and pressure

- it is so rapid that no flow occurs
- no heat convection between compression and rarefaction occurs
- in a linear case we assume that the disturbance is the same regardless of the static air pressure

So, let's add a small disturbance to the static pressure ${\it P}_{\rm 0}$ and density $\rho_{\rm 0}$

$$P = P_0 + P_e \tag{1}$$

$$\rho = \rho_0 + \rho_e \tag{2}$$

so that $P_0 \gg P_e$ and $\rho_0 \gg \rho_e$



The physics of the sound waves involves three features I The gas moves and changes density

Fluids

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Continuums

Wave Equation

Speed of sound



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The physics of the sound waves involves three features

- Fluids
- Ideal Fluid
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- Continuums
- Wave Equation
- Speed of sound

- I The gas moves and changes density
- II The change in density corresponds to a change in pressure (adiabatic process)



Fluids Ideal Fluid

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Continuums Wave Equation

Speed of sound

The physics of the sound waves involves three features

- The gas moves and changes density
- II The change in density corresponds to a change in pressure (adiabatic process)
- III Pressure inequalities (spatial pressure gradients) generate gas motion



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Fluids

Ideal Gas

Continuums Wave Equation

Speed of sound

The physics of the sound waves involves three features

- I The gas moves and changes density
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 \Rightarrow See also: www.feynmanlectures.caltech.edu/I_47.html



III Spatial gradients generate gas motion

 $\rho_0 \frac{\partial^2 \chi}{\partial t^2} = -\frac{\partial P_e}{\partial x}$

Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound



III Spatial gradients generate gas motion

Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

combined with II Change in density is change in pressure $P_{e} = \kappa \rho_{e}$

 $\rho_0 \frac{\partial^2 \chi}{\partial t^2} = -\frac{\partial P_e}{\partial x}$



III Spatial gradients generate gas motion

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Wave Equation

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and finally I Gas moves and changes density $\rho_{e} = -\rho_{0} \frac{\partial \chi}{\partial x}$



III Spatial gradients generate gas motion

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Continuums

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Speed of sound

combined with II Change in density is change in pressure $P_{e} = \kappa \rho_{e}$

$$\rho_0 \frac{\partial^2 \chi}{\partial t^2} = -\kappa \frac{\partial \rho_e}{\partial x}$$

 $\rho_0 \frac{\partial^2 \chi}{\partial t^2} = -\frac{\partial P_e}{\partial \mathbf{x}}$

and finally I Gas moves and changes density $\rho_e = -\rho_0 \frac{\partial \chi}{\partial x}$

$$\frac{\partial^2 \chi}{\partial t^2} = \kappa \frac{\partial^2 \chi}{\partial x^2}$$

where $\kappa = 1/c^2 = \rho/\gamma P$ for an adiabatic process

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1 Combining features

Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

So, our one-dimensional wave equation for a fluid looks very familiar:

$$\frac{\partial^2 \chi}{\partial t^2} = \kappa \frac{\partial^2 \chi}{\partial x^2}$$

We have the inertial force from the fluid mass, and restoring force from the pressure.

Note that we now derived the equation for displacement, the same applies for pressure.

[Check again Feynman "Lectures on physics" from previous slides for details]



1 Adiabatic process - nonlinearity

Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

Note that in the adiabatic process the pressure-density relation is not linear

$$P = \alpha \rho^{\gamma}$$

However, the changes associated with audible sound are so small that we can approximate their variation with the tangent in the equilibrium point, like we did above.




Ideal Fluid

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Continuums

Wave Equation

Speed of sound



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 $c = \sqrt{\frac{K}{\rho}}$, how to get the bulk modulus, *K*, in practice?

Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

- $c = \sqrt{\frac{K}{\rho}}$, how to get the bulk modulus, *K*, in practice?
 - depends on the thermodynamic behaviour of the medium
 - **isothermal**: heat transfer between nodes and antinodes, $K = P_0$, P_0 is static air pressure



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Fluids

- Ideal Fluid
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- Continuums
- Wave Equation
- Speed of sound

- $c = \sqrt{\frac{\kappa}{\rho}}$, how to get the bulk modulus, κ , in practice?
 - depends on the thermodynamic behaviour of the medium
 - **isothermal**: heat transfer between nodes and antinodes, *K* = *P*₀, *P*₀ is static air pressure
 - **adiabatic**: no heat transfer, $K = \gamma P_0$, γ is the adiabatic index (gas property, for air $\gamma = 1.4$)



Fluids

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Speed of sound

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 - **isothermal**: heat transfer between nodes and antinodes, $K = P_0$, P_0 is static air pressure
 - adiabatic: no heat transfer, K = γP₀, γ is the adiabatic index (gas property, for air γ = 1.4)
 - normally, air behaves as adiabatic (contrary to what Newton thought)
 - near solids and in porous materials air behaves as isothermal



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Speed of sound

- $c = \sqrt{\frac{\kappa}{\rho}}$, how to get the bulk modulus, κ , in practice?
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 - **isothermal**: heat transfer between nodes and antinodes, $K = P_0$, P_0 is static air pressure
 - adiabatic: no heat transfer, K = γP₀, γ is the adiabatic index (gas property, for air γ = 1.4)
 - normally, air behaves as adiabatic (contrary to what Newton thought)
 - near solids and in porous materials air behaves as isothermal ⇒ porosity slows down sound



Fluids

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Continuums

Wave Equation Speed of sound

where T is temperature and c_0 is sound velocity at temperature T_0 .

 $c = c_0 \sqrt{T/T_0}$

A practical formula for the speed of sound in an ideal gas:



(R&F(6.13))

Fluids

Ideal Fluid

Ideal Gas

Continuums

Wave Equation

Speed of sound

A practical formula for the speed of sound in an ideal gas:

$$c = c_0 \sqrt{T/T_0},$$
 (R&F(6.13))

where T is temperature and c_0 is sound velocity at temperature T_0 . In practice, also

- inversely proportional to gas density
- increases with humidity

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Fluids

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Wave Equation

Speed of sound

A practical formula for the speed of sound in an ideal gas:

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inversely proportional to gas density

increases with humidity

Typical sound speeds

- gases: 200-1000 m/s
- liquids: 1200-1600 m/s
- solids: 3000-6000 m/s



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Loss Mechanisms



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2 Loss Types

Loss Types

Viscosity

Heat Conduction

Molecular Energy Transfer In real fluids, various phenomena attenuate sound waves: geometric attenuation



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2 Loss Types

Loss Types

Viscosity

Heat Conduction

Molecular Energy Transfer In real fluids, various phenomena attenuate sound waves:

- geometric attenuation
- Iosses due to boundaries



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2 Loss Types

Loss	Types
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Viscosity

- Heat Conduction
- Molecular Energy Transfer

In real fluids, various phenomena attenuate sound waves:

- geometric attenuation
- Iosses due to boundaries
- Iosses within the medium
 - viscosity
 - heat conduction
 - relaxation



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2 Viscosity

Loss Types

Viscosity

Heat Conduction Molecular Energy

Transfer

Viscosity causes a shear stress (force) σ_{xy} in the direction of the movement.





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2 Viscosity

Loss Types

Viscosity

Heat Conduction Molecular Energy Transfer Viscosity causes a shear stress (force) σ_{xy} in the direction of the movement. This stress force tries to cancel the velocity difference of adjacent fluid "layers" in the direction orthogonal to the movement (*y*-direction in the figure).





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2 Viscosity

Loss Types

Viscosity

Heat Conduction Molecular Energy Transfer Viscosity causes a shear stress (force) σ_{xy} in the direction of the movement. This stress force tries to cancel the velocity difference of adjacent fluid "layers" in the direction orthogonal to the movement (*y*-direction in the figure).



$$\sigma_{xy} = \mu \frac{\partial u_x}{\partial y}$$
, where

μ is the *shear viscosity* (property of the fluid)
 u_x is the velocity of the fluid



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2 Heat Conduction

Loss Types

Viscosity

Heat Conduction

Molecular Energy Transfer

- heat flows from compressed areas to uncompressed ones
- reduces pressure difference between wave maxima and minima
 - \blacksquare \Rightarrow wave attenuates
- compressibility no longer adiabatic
- typical case near solids



Loss Types Viscosity

Heat Conduction

Molecular Energy Transfer When a fluid is compressed, its pressure, density, and temperature increase.



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Loss Types Viscosity Heat Conduction

Molecular Energy Transfer When a fluid is compressed, its pressure, density, and temperature increase.

some of the energy of the compression is transferred into rotational and vibrational movement of the molecules



Loss Types Viscosity

Heat Conduction

Molecular Energy Transfer When a fluid is compressed, its pressure, density, and temperature increase.

some of the energy of the compression is transferred into rotational and vibrational movement of the molecules

 \blacksquare \Rightarrow pressure decreases!



Loss Types Viscosity

Heat Conduction

Molecular Energy Transfer When a fluid is compressed, its pressure, density, and temperature increase.

- some of the energy of the compression is transferred into rotational and vibrational movement of the molecules
 - ⇒ pressure decreases!
- this phenomenon is called relaxation
 - irreversible process



Loss Types Viscosity

Heat Conduction

Molecular Energy

Transfer

When a fluid is compressed, its pressure, density, and temperature increase.

- some of the energy of the compression is transferred into rotational and vibrational movement of the molecules
 - ⇒ pressure decreases!
- this phenomenon is called relaxation
 - irreversible process
- it takes a certain time until the energy transfer is finished and an equilibrium is found
 - this time is called relaxation time







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Rectangular Enclosure	Consider a rectangular enclosure	
Solution	-	
Room Modes		
Mode Types		
Axial Modes		
Tangential Modes		
Oblique Modes		
Eigenfrequencies		
Source Location		
Below the Lowest		
Eigenfrequency		
Modal Density		
Reverberation		
Further Info		



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Rectangular Enclosure

Solution

Room Modes

Mode Types

Axial Modes

Tangential Modes

Oblique Modes

Eigenfrequencies

Source Location

Below the Lowest Eigenfrequency

Modal Density

Reverberation

Further Info

Consider a rectangular enclosure with dimensions $a \times b \times c$.



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Rectangular

- Enclosure Solution
- Room Modes
- Mode Types
- Axial Modes
- Tangential Modes **Oblique Modes**
- Eigenfrequencies
- Source Location
- Below the Lowest Eigenfrequency
- Modal Density
- Reverberation
- Further Info

Consider a rectangular enclosure with dimensions $a \times b \times c$. Assume rigid walls

- total reflection
- $u_n = 0$ at the boundary
- p maximum at the boundary



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Rectangular

- Enclosure Solution
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Consider a rectangular enclosure with dimensions $a \times b \times c$. Assume rigid walls

- total reflection
- $u_{\rm n} = 0$ at the boundary
- *p* maximum at the boundary
- Wave equation has the familiar form

$$\ddot{p} = c^2 \nabla^2 p$$





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Rectangular

- Enclosure Solution
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- **Oblique Modes**
- Eigenfrequencies
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Consider a rectangular enclosure with dimensions $a \times b \times c$. Assume rigid walls

- total reflection
- $u_n = 0$ at the boundary
- p maximum at the boundary
- Wave equation has the familiar form

$$\ddot{p} = c^2 \nabla^2 p$$

with the boundary condition $z_{wall} = \infty$



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Rectangular Enclosure

Solution

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Oblique Modes

Eigenfrequencies

Source Location

Below the Lowest Eigenfrequency

Modal Density

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Further Info

$$p(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} p_{lmn}(x, y, z, t)$$
$$p_{lmn}(x, y, z, t) = A_{lmn} \cos\left(\frac{l\pi x}{a}\right) \cos\left(\frac{m\pi y}{b}\right) \cos\left(\frac{n\pi z}{c}\right) \sin(\omega t)$$

(R&F:6.55,FF:9.1)

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$$\omega = \pi c \left(\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2}\right)^{1/2}$$
(3)

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Rectangular Enclosure

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Below the Lowest Eigenfrequency

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Further Info

$$(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} p_{lmn}(x, y, z, t)$$

$$= A_{lmn} \cos\left(\frac{1}{a}\right) \cos\left(\frac{1}{b}\right) \cos\left(\frac{1}{c}\right) \sin(\omega t)$$
(R&F:6.55,FF:9.1)
$$= \pi c \left(\frac{l^2}{a} + \frac{m^2}{m^2} + \frac{n^2}{a}\right)^{1/2}$$
(3)



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Rectangular Enclosure

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Further Info

$$p(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} p_{lmn}(x, y, z, t)$$

$$p_{lmn}(x, y, z, t) = A_{lmn} \cos\left(\frac{l\pi x}{a}\right) \cos\left(\frac{m\pi y}{b}\right) \cos\left(\frac{n\pi z}{c}\right) \sin(\omega t)$$
(R&F:6.55,FF:9.1)
$$\omega = \pi c \left(\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2}\right)^{1/2}$$
(3)



Rectangular Enclosure

Solution

Room Modes

Mode Types

Axial Modes

Tangential Modes

Oblique Modes

Eigenfrequencies

Source Location

Below the Lowest Eigenfrequency

Modal Density

Reverberation

Further Info

$$(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} p_{lmn}(x, y, z, t)$$
$$(x, y, z, t) = \mathbf{A}_{lmn} \cos\left(\frac{l\pi x}{a}\right) \cos\left(\frac{m\pi y}{b}\right) \cos\left(\frac{n\pi z}{c}\right) \sin(\omega t)$$
$$(\mathbf{R}\&\mathbf{F}:\mathbf{6}.\mathbf{55}, \mathbf{F}\mathbf{F}:\mathbf{9}.\mathbf{1})$$
$$\omega = \pi c \left(\frac{l^2}{2} + \frac{m^2}{12} + \frac{n^2}{2}\right)^{1/2}$$
(3)

which consists of a sum over three kind of modes. Each mode is given by a product of **modal amplitude**



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Source Location

Below the Lowest Eigenfrequency

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Further Info

$$(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} p_{lmn}(x, y, z, t)$$
$$(x, y, z, t) = A_{lmn} \cos\left(\frac{l\pi x}{a}\right) \cos\left(\frac{m\pi y}{b}\right) \cos\left(\frac{n\pi z}{c}\right) \sin(\omega t)$$
$$(R&F:6.55,FF:9.1)$$
$$\omega = \pi c \left(\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2}\right)^{1/2}$$
(3)

- modal amplitude
- standing waves in the x,



Rectangular Enclosure

Solution

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Eigenfrequencies

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Further Info

$$p(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} \rho_{lmn}(x, y, z, t)$$

$$p(x, y, z, t) = A_{lmn} \cos\left(\frac{l\pi x}{a}\right) \cos\left(\frac{m\pi y}{b}\right) \cos\left(\frac{n\pi z}{c}\right) \sin(\omega t)$$

$$(R\&F:6.55, FF:9.1)$$

$$\omega = \pi c \left(\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2}\right)^{1/2}$$
(3)

- modal amplitude
- standing waves in the x, y,



Rectangular Enclosure

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Oblique Modes

Eigenfrequencies

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Modal Density

Reverberation

Further Info

$$p(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} p_{lmn}(x, y, z, t)$$

$$n(x, y, z, t) = A_{lmn} \cos\left(\frac{l\pi x}{a}\right) \cos\left(\frac{m\pi y}{b}\right) \cos\left(\frac{n\pi z}{c}\right) \sin(\omega t)$$
(R&F:6.55,FF:9.1)
$$\omega = \pi c \left(\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2}\right)^{1/2}$$
(3)

- modal amplitude
- standing waves in the x, y, and z directions



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Further Info

$$p(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} p_{lmn}(x, y, z, t)$$
$$p_{lmn}(x, y, z, t) = A_{lmn} \cos\left(\frac{l\pi x}{a}\right) \cos\left(\frac{m\pi y}{b}\right) \cos\left(\frac{m\pi z}{c}\right) \sin(\omega t)$$

(R&F:6.55,FF:9.1)

$$\omega = \pi c \left(\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2} \right)^{1/2}$$
(3)

which consists of a sum over three kind of modes. Each mode is given by a product of

- modal amplitude
- standing waves in the x, y, and z directions

temporal vibration



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Further Info

$$p(x, y, z, t) = \sum_{l=0}^{\infty} \sum_{m=0}^{\infty} \sum_{n=0}^{\infty} p_{lmn}(x, y, z, t)$$
$$p_{lmn}(x, y, z, t) = A_{lmn} \cos\left(\frac{l\pi x}{a}\right) \cos\left(\frac{m\pi y}{b}\right) \cos\left(\frac{n\pi z}{c}\right) \sin(\omega t)$$

(R&F:6.55,FF:9.1)

$$\omega = \pi c \left(\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2} \right)^{1/2}$$
(3)

- modal amplitude
- standing waves in the x, y, and z directions
- temporal vibration


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y, (m) x, (l)



Solution

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Modal Density

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Further Info

Mode numbers *I*, *m*, *n* represent the number of pressure zero-crossings along the coordinate axes.



Let's visualize the pressure distribution with grayscale color (white denotes high absolute pressure value, black denotes pressure zero-crossing):



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y, (m) x, (l)



Solution

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Further Info

Mode numbers *I*, *m*, *n* represent the number of pressure zero-crossings along the coordinate axes.



Let's visualize the pressure distribution with grayscale color (white denotes high absolute pressure value, black denotes pressure zero-crossing):

This is mode (1,0,0), since there is one zero-crossing along the *x*-axis



y, (m) | x, (l)

Rectangular Enclosure

Solution

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Further Info

Mode numbers *I*, *m*, *n* represent the number of pressure zero-crossings along the coordinate axes.



Let's visualize the pressure distribution with grayscale color (white denotes high absolute pressure value, black denotes pressure zero-crossing): How about this?



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y, (m) | x, (l)

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Further Info

Mode numbers *I*, *m*, *n* represent the number of pressure zero-crossings along the coordinate axes.



Let's visualize the pressure distribution with grayscale color (white denotes high absolute pressure value, black denotes pressure zero-crossing):

This is mode (0,2,0), since there are two zero-crossings along the *y*-axis





Solution

Room Modes

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Further Info

Mode numbers *I*, *m*, *n* represent the number of pressure zero-crossings along the coordinate axes.



Let's visualize the pressure distribution with grayscale color (white denotes high absolute pressure value, black denotes pressure zero-crossing):

And what about this?

y, (m) x, (l)



y, (m) x, (l)



Solution

Room Modes

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Further Info

Mode numbers *I*, *m*, *n* represent the number of pressure zero-crossings along the coordinate axes.



Let's visualize the pressure distribution with grayscale color (white denotes high absolute pressure value, black denotes pressure zero-crossing):

This is mode (0,0,3), since there are three zero-crossings along the *z*-axis



y, (m) | x, (l)

Rectangular Enclosure

Solution

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Mode numbers *I*, *m*, *n* represent the number of pressure zero-crossings along the coordinate axes.



Let's visualize the pressure distribution with grayscale color (white denotes high absolute pressure value, black denotes pressure zero-crossing):

Bonus: which mode is this?



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y, (m) x, (l)

Rectangular Enclosure

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Mode numbers *I*, *m*, *n* represent the number of pressure zero-crossings along the coordinate axes.



Let's visualize the pressure distribution with grayscale color (white denotes high absolute pressure value, black denotes pressure zero-crossing):

This is mode (3,2,0). See also

http://www.falstad.com/modebox/



3 Axial Modes





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3 Axial Modes



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3 Tangential Modes



Modes for which two numbers (I, m, or n) are nonzero, are



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3 Tangential Modes

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3 Oblique Modes



Modes for which all numbers (I, m, or n) are nonzero, are





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3 Oblique Modes





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The eigenfrequencies of a rectangular enclosure are given as

 $f_{lmn} = \frac{c_0}{\sqrt{\frac{l^2}{m^2} + \frac{m^2}{m^2} + \frac{n^2}{m^2}}}$

Rectangular

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where
$$c_0$$
 is the speed of sound, and $l, m, n = 0, 1, 2, 3, ...$

(4)

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as

The eigenfrequencies of a rectangular enclosure are given

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$$f_{lmn} = \frac{c_0}{2} \sqrt{\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2}},$$
 (4)

where c_0 is the speed of sound, and l, m, n = 0, 1, 2, 3, ...After a transient sound excitation, there are frequency components only at the eigenfrequencies (compare e.g. to a string). As before, we neglect losses.



as

The eigenfrequencies of a rectangular enclosure are given

Rectangular

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$$f_{lmn} = \frac{c_0}{2} \sqrt{\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2}},$$
 (4)

where c_0 is the speed of sound, and l, m, n = 0, 1, 2, 3, ...After a transient sound excitation, there are frequency components only at the eigenfrequencies (compare e.g. to a string). As before, we neglect losses.

in steady state: only standing waves!



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The eigenfrequencies of a rectangular enclosure are given

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$$f_{lmn} = \frac{c_0}{2} \sqrt{\frac{l^2}{a^2} + \frac{m^2}{b^2} + \frac{n^2}{c^2}},$$
 (4)

where c_0 is the speed of sound, and l, m, n = 0, 1, 2, 3, ...After a transient sound excitation, there are frequency components only at the eigenfrequencies (compare e.g. to a string). As before, we neglect losses.

in steady state: only standing waves!

However, with a continuous source, also other frequency components exist. In other words, solution (3) is valid only in the "free vibration case".



Rectangular Enclosure Solution

Room Modes

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Further Info

Similarly as with the string and membrane, the location of the source affects the resulting modes.



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Further Info

Similarly as with the string and membrane, the location of the source affects the resulting modes.

https://www.acs.psu.edu/drussell/Demos/RoomModes/driving.html



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Further Info

Similarly as with the string and membrane, the location of the source affects the resulting modes.

https://www.acs.psu.edu/drussell/Demos/RoomModes/driving.html

Typically, strong room modes (at low frequencies) are to be avoided in audio applications, since they add significant coloration



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Similarly as with the string and membrane, the location of the source affects the resulting modes.

https://www.acs.psu.edu/drussell/Demos/RoomModes/driving.html

Typically, strong room modes (at low frequencies) are to be avoided in audio applications, since they add significant coloration

considering this, what is the worst place to position a loudspeaker in a room?



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Similarly as with the string and membrane, the location of the source affects the resulting modes.

https://www.acs.psu.edu/drussell/Demos/RoomModes/driving.html

Typically, strong room modes (at low frequencies) are to be avoided in audio applications, since they add significant coloration

- considering this, what is the worst place to position a loudspeaker in a room?
 - corner, since there it will excite the room modes best



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Similarly as with the string and membrane, the location of the source affects the resulting modes.

https://www.acs.psu.edu/drussell/Demos/RoomModes/driving.html

Typically, strong room modes (at low frequencies) are to be avoided in audio applications, since they add significant coloration

considering this, what is the worst place to position a loudspeaker in a room?

corner, since there it will excite the room modes best

Depending on the listening position, one will either measure/hear a frequency (at antinodes of standing wave) or not (at nodes of standing wave).



3 Several subwoofers

Abou

t 50% of modes are in opposite phase in two corners



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3 Several subwoofers

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About 50% of modes are in opposite phase in two corners If you put a subwoofer in each corner, and drive it with in-phase signal, what happens?



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3 Several subwoofers

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About 50% of modes are in opposite phase in two corners

- If you put a subwoofer in each corner, and drive it with in-phase signal, what happens?
- Antiphasic modes are cancelled



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3 Sound Field Below the Lowest Eigenfrequency

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Further Info

If the frequency of an excitation < the lowest eigenfrequency, the sound field consists mainly of the mode (0, 0, 0)



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3 Sound Field Below the Lowest Eigenfrequency

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Further Info

If the frequency of an excitation < the lowest eigenfrequency, the sound field consists mainly of the mode (0, 0, 0)

does not depend on location



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3 Sound Field Below the Lowest Eigenfrequency

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Further Info

If the frequency of an excitation < the lowest eigenfrequency, the sound field consists mainly of the mode (0,0,0)

does not depend on location

the room acts as an acoustic capacitance

$$p=rac{q}{i\omega C},$$

where $C = \frac{V}{K}$ is the acoustic capacitance, *V* is room volume, and *K* is the bulk modulus of air.



3 Modal Density

The average densities (eigenfrequencies per Hz) can be given for the different mode types as:

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$$n_{
m a} = rac{L}{2c_0}$$
 for axial modes
 $n_{
m t} = rac{\pi A f}{c_0^2} - rac{L}{2c_0}$ for tangential modes
 $n_{
m o} = rac{4\pi f^2 V}{c_0^3} - rac{\pi A f}{c_0^2}$ for oblique modes

where V is room volume, A is area of the surfaces, and L is the total length of the edges.



3 Modal Density

The average densities (eigenfrequencies per Hz) can be given for the different mode types as:

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$$\begin{split} n_{\rm a} &= \frac{L}{2c_0} \text{ for axial modes} \\ n_{\rm t} &= \frac{\pi A f}{c_0^2} - \frac{L}{2c_0} \text{ for tangential modes} \\ n_{\rm o} &= \frac{4\pi f^2 V}{c_0^3} - \frac{\pi A f}{c_0^2} \text{ for oblique modes} \end{split}$$

where *V* is room volume, *A* is area of the surfaces, and *L* is the total length of the edges. An approximation for the total modal density for high frequencies is $n \approx \frac{4\pi V f^2}{c_n^2}$



3 Modal Density II

Rectangular Enclosure Solution Room Modes

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Tangential Modes

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Further Info

The modes should be evenly spaced in frequency to avoid distinct resonances at low frequencies

room dimensions should not have integer relations



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3 Modal Density II

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The modes should be evenly spaced in frequency to avoid distinct resonances at low frequencies

- room dimensions should not have integer relations
- typical problem with small rooms
 - living rooms
 - recording studios
 - sound control rooms



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3 Modal Density II

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The modes should be evenly spaced in frequency to avoid distinct resonances at low frequencies

- room dimensions should not have integer relations
- typical problem with small rooms
 - living rooms
 - recording studios
 - sound control rooms

Table:Some widely usedrecommendations

Source	Dimension ratio
ASHRAE	1:1.17:1.47
Bolt	1:1.28:1.54
IAC	1:1.25:1.60
Sepmeyer	1:1.14:1.39
1:∛2:√2	1:1.26:1.41



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3 Distribution of modes in two rooms



Fig. 6.3. Distribution of mode frequencies for two rectangular rooms of equal volume and dimension ratios (a) 2:2:2 and (b) 1:2:3. Where mode frequencies are coincident, the relevant line has been lengthened proportionally.



Rectangular

Room Modes

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3 Reverberation

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- **Oblique Modes**
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- Source Location
- Below the Lowest Eigenfrequency
- Modal Density

Reverberation

Further Info



Figure 3: Sound propagation from an impulse source to a microphone inside a room (a), and the corresponding impulse response signal (b), as recorded by the microphone.



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3 Reverberation

- Rectangular Enclosure Solution
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Figure 3: Sound propagation from an impulse source to a microphone inside a room (a), and the corresponding impulse response signal (b), as recorded by the microphone.

Reverberation time, T_{60} is the time it takes for the reverberation to decay by 60 dB.



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Figure 3: Sound propagation from an impulse source to a microphone inside a room (a), and the corresponding impulse response signal (b), as recorded by the microphone.

Reverberation time, T_{60} is the time it takes for the reverberation to decay by 60 dB. In diffuse sound fields with low wall absorption: approximated with *Sabine's formula*:

$$T_{60}=0.161\frac{V}{A_{\alpha}},$$

where A_{α} is the equivalent absorption area (calculated from material properties and absorption area).



3 Further Info

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