

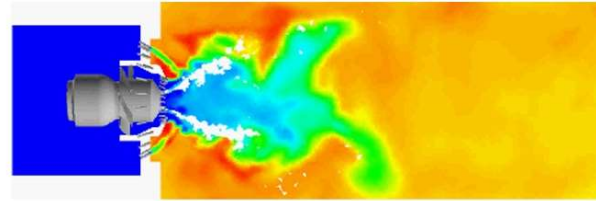
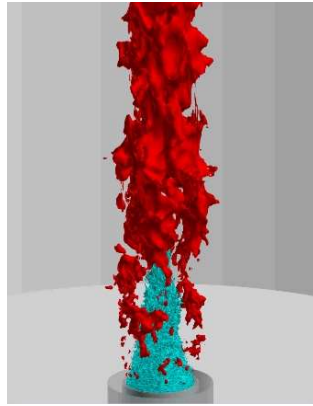


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Fundamentals of Sprays

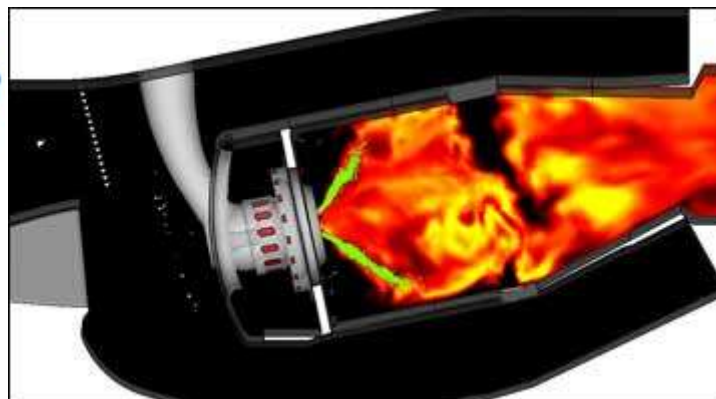
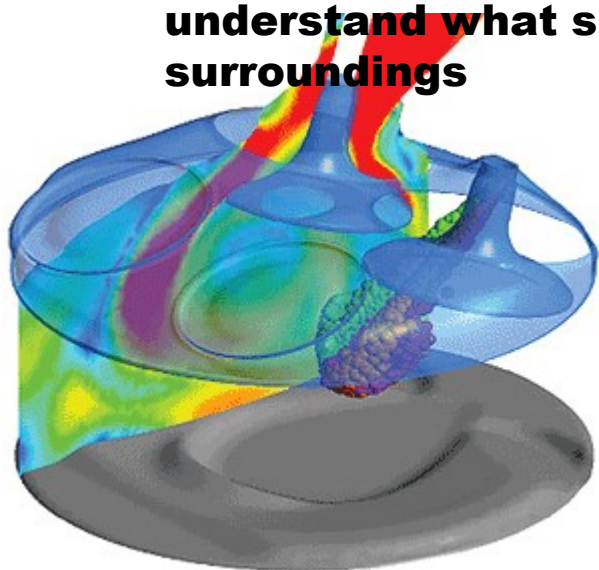
AAE-E3030 Numerical Modeling of Multiphase Flows
25.2.2019

D.Sc. (Tech) Ossi Kaario



Motivation

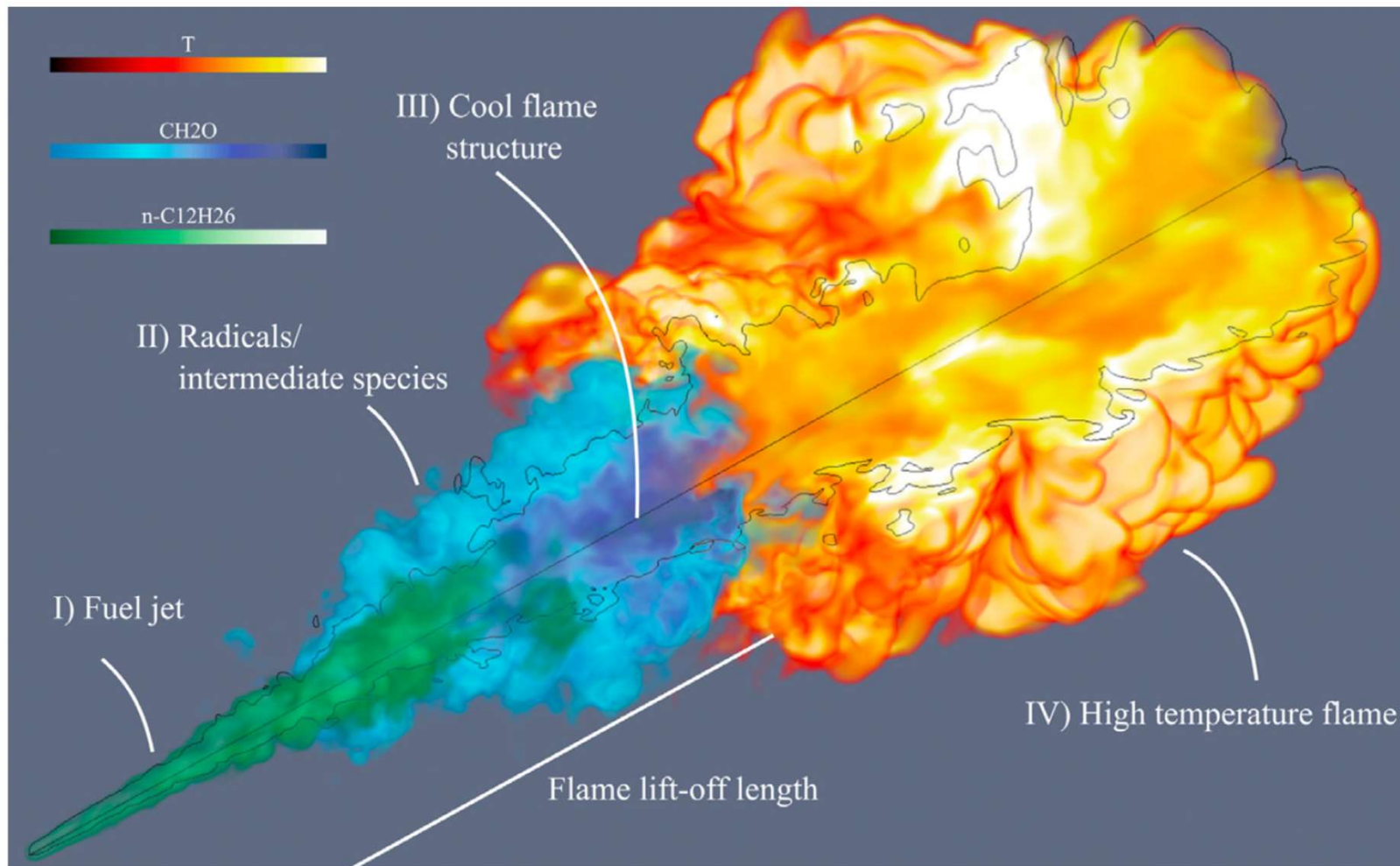
- **Why learn about sprays ?**
- **Sprays are used in many industrial applications: engines (diesel & gasoline), furnaces, gas turbines, rockets and in many other applications: aerosols, water taps...**
- **Therefore, understanding concepts related to sprays might be beneficial**
- **This lesson is about sprays: general topics of sprays in order to better understand what sprays are and how they form and interact with their surroundings**





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Course related phenomena



Wehrfritz et al., Comb. Flame 2016.

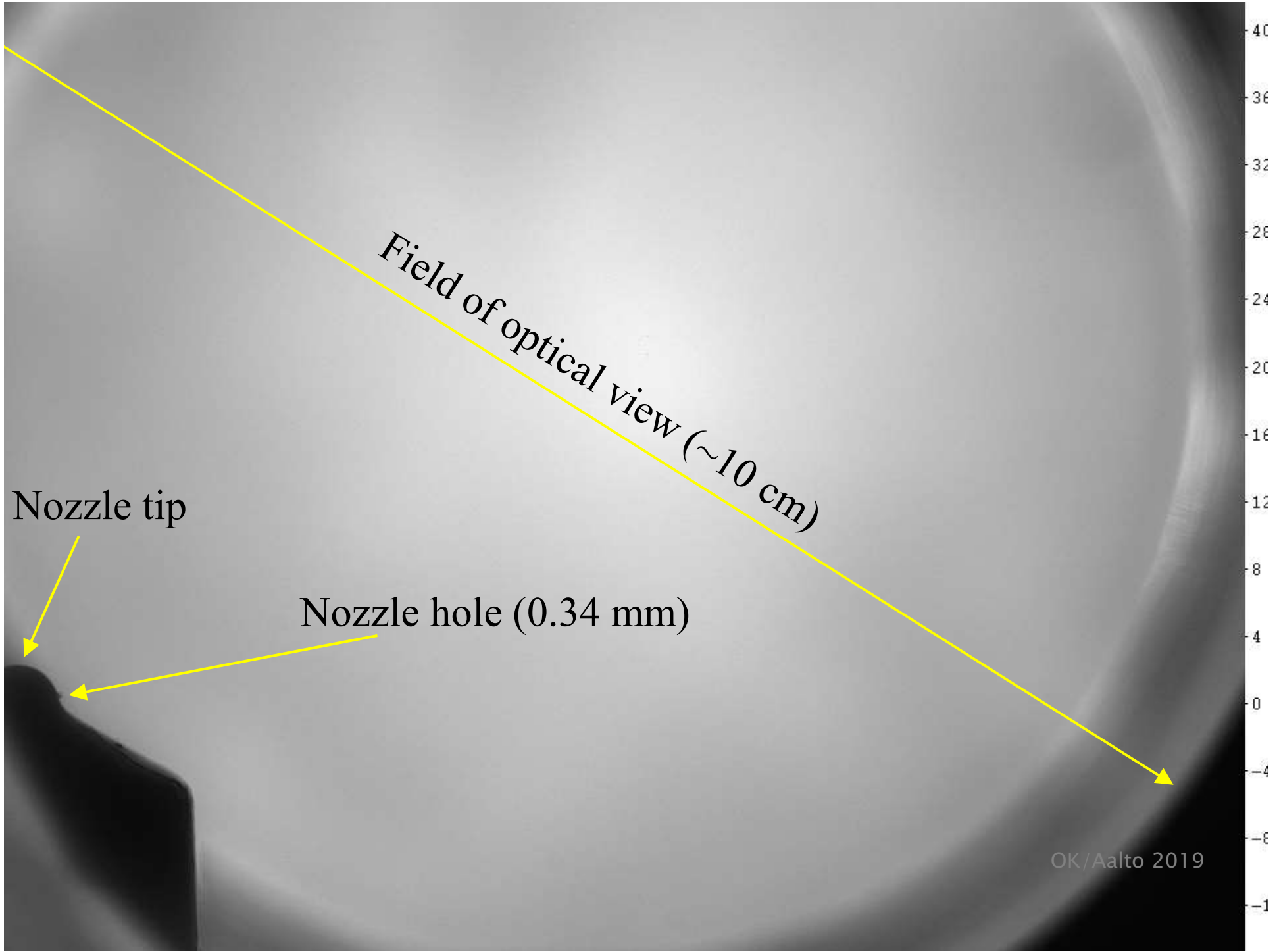


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What does a fuel spray look like ?

Optical Spray Investigations

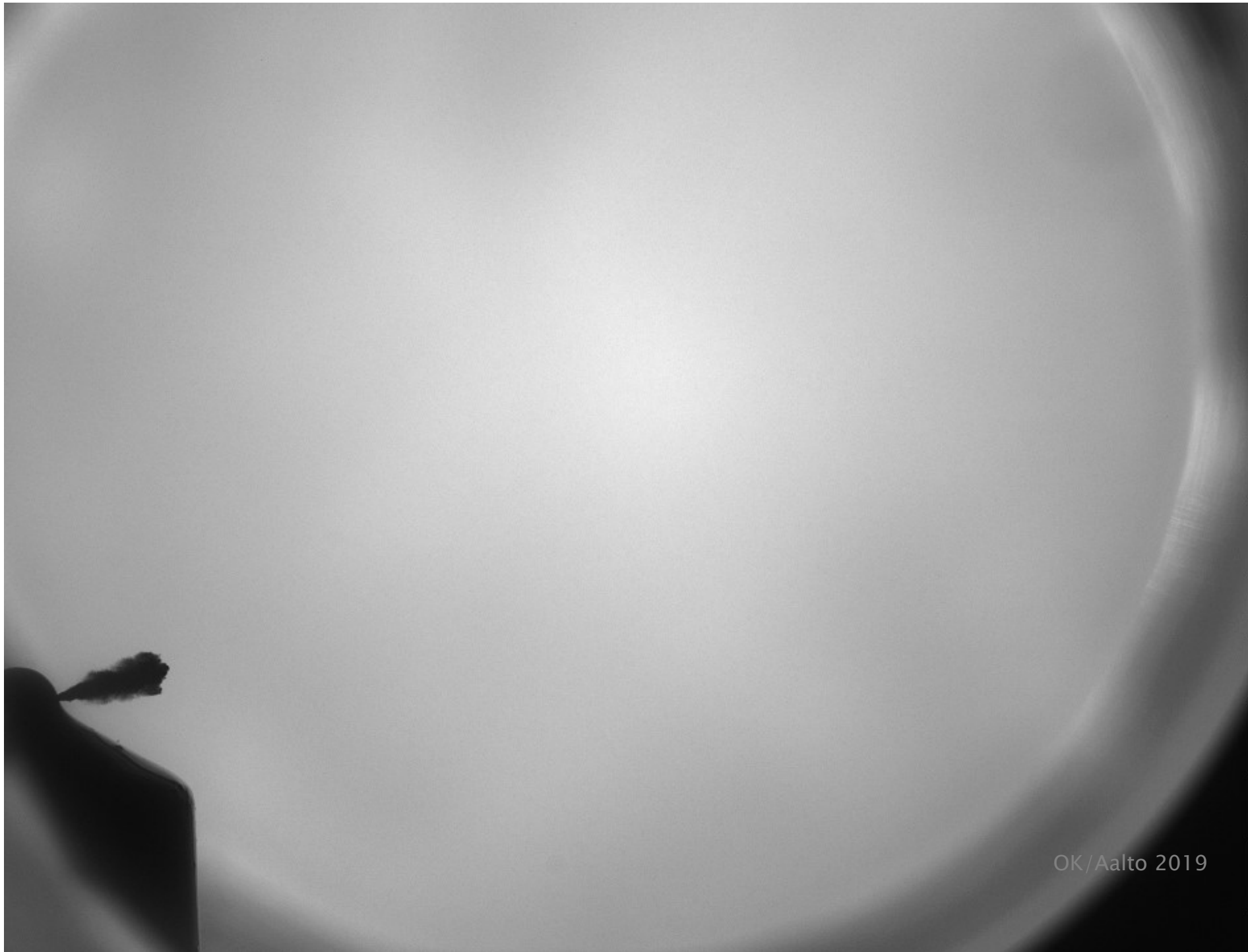
- **Pictures by Harri Hillamo, 2010**
- **Fuel injection pressure is 1400bar**
- **Gas density is 35 kg/m³**
- **Nozzle diameter $d=0.34\text{mm}$**



Nozzle tip

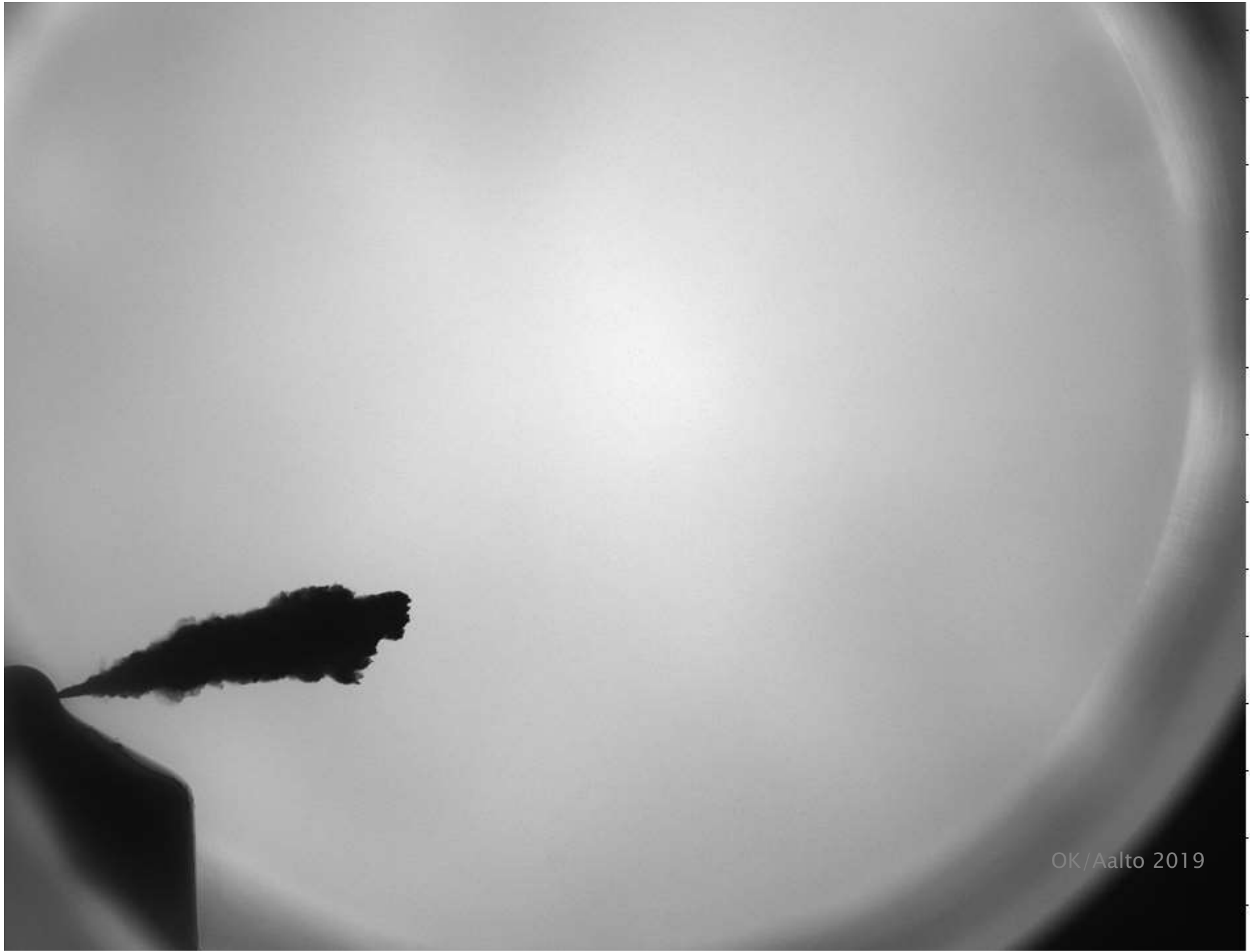
Nozzle hole (0.34 mm)

Field of optical view (~10 cm)



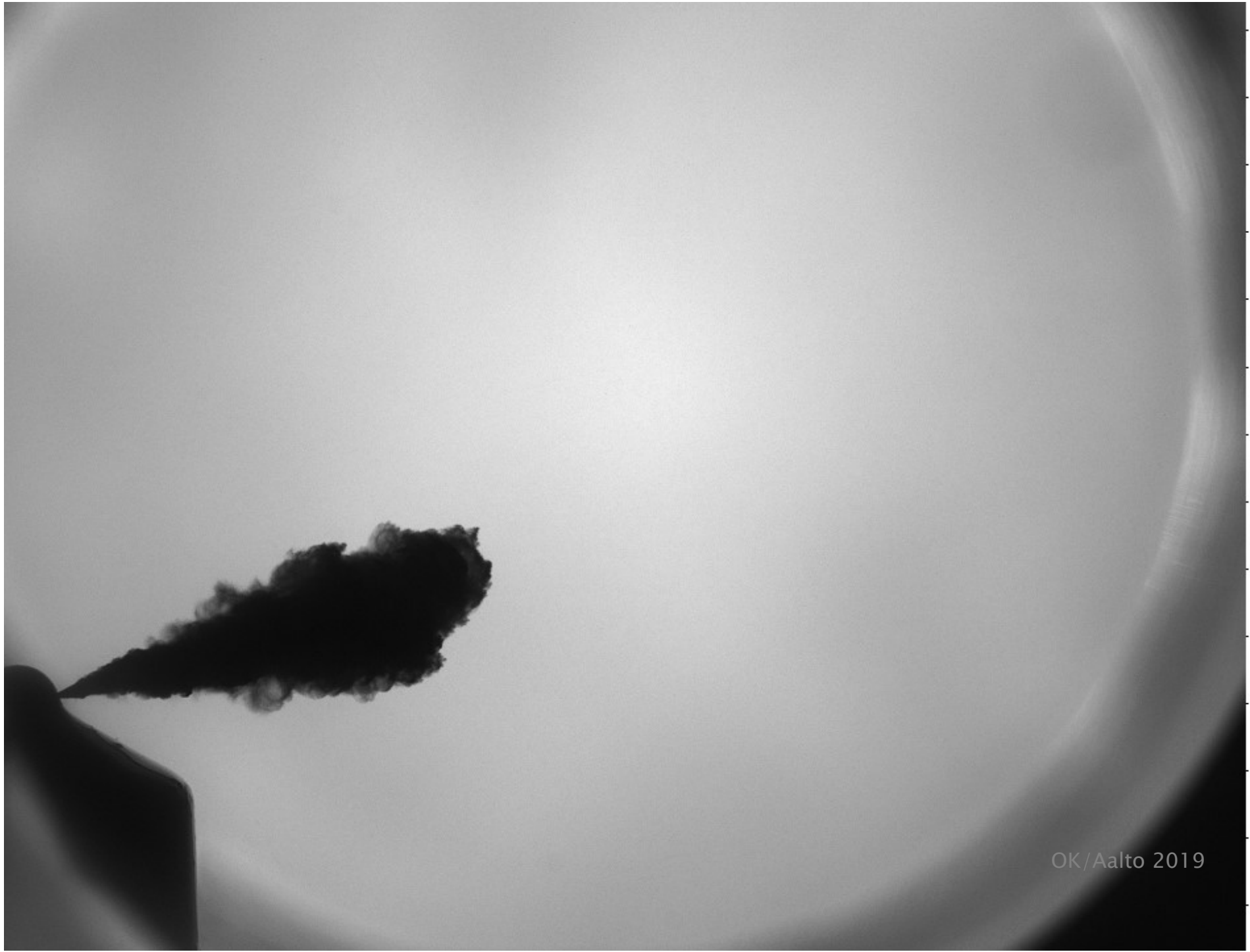
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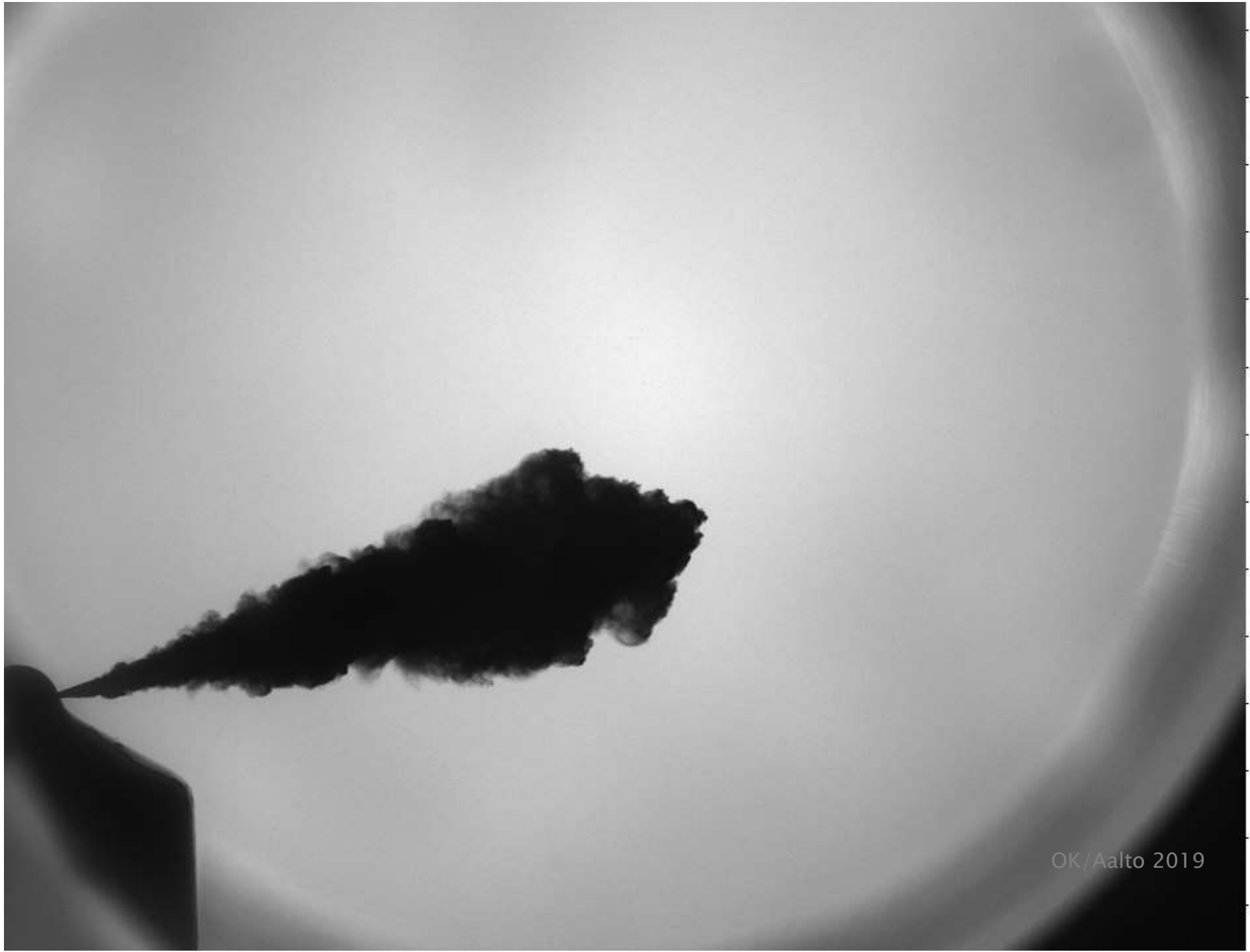
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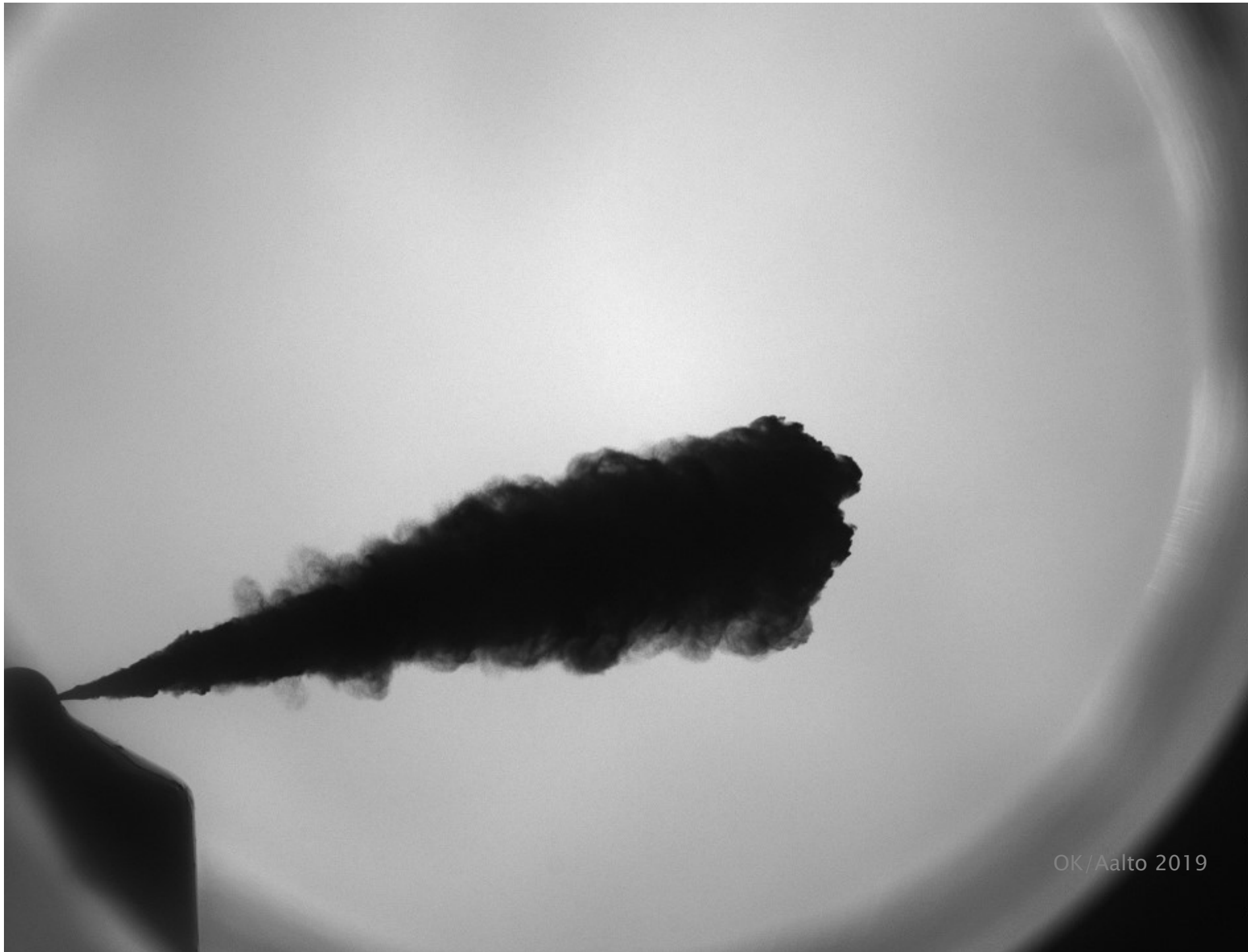


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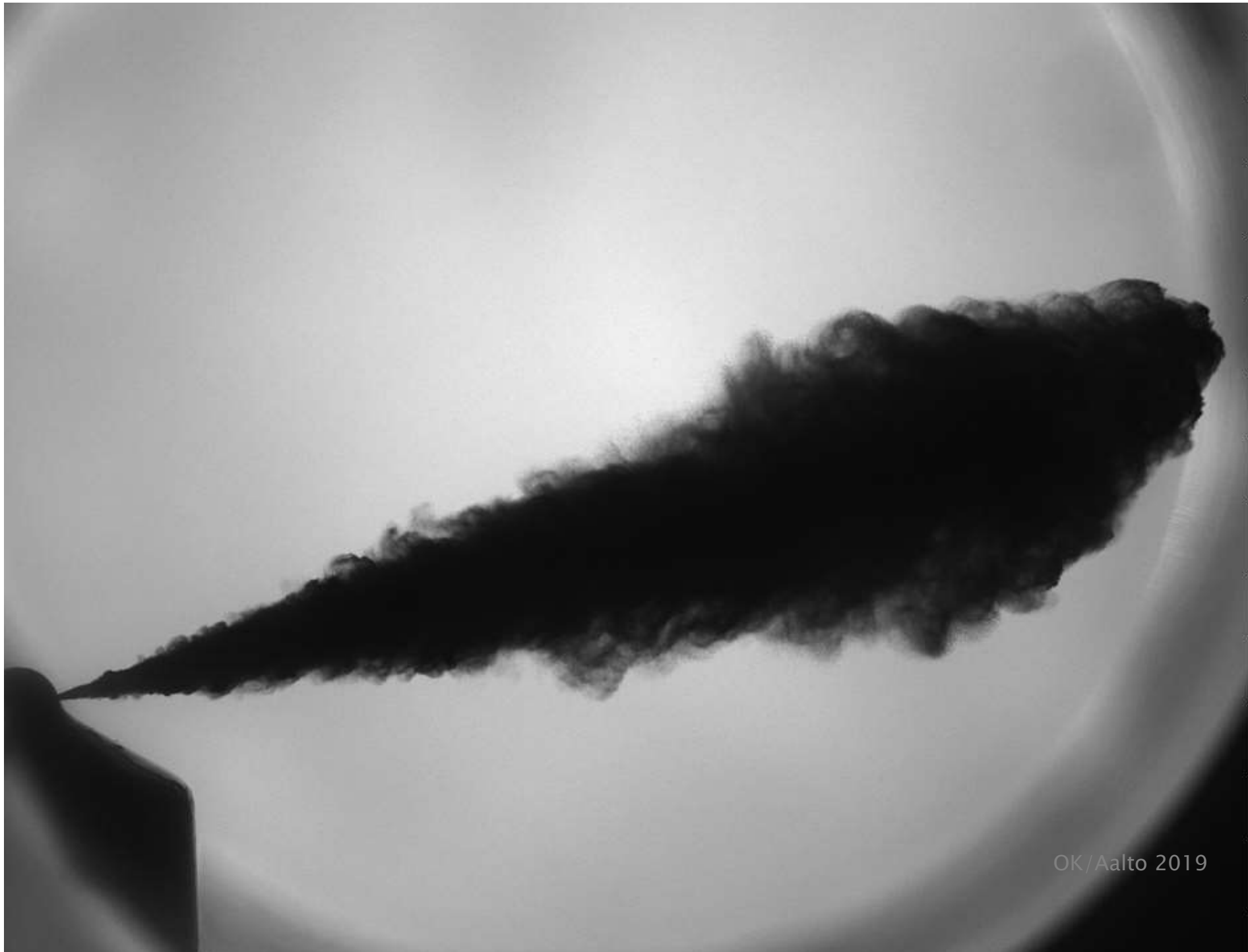






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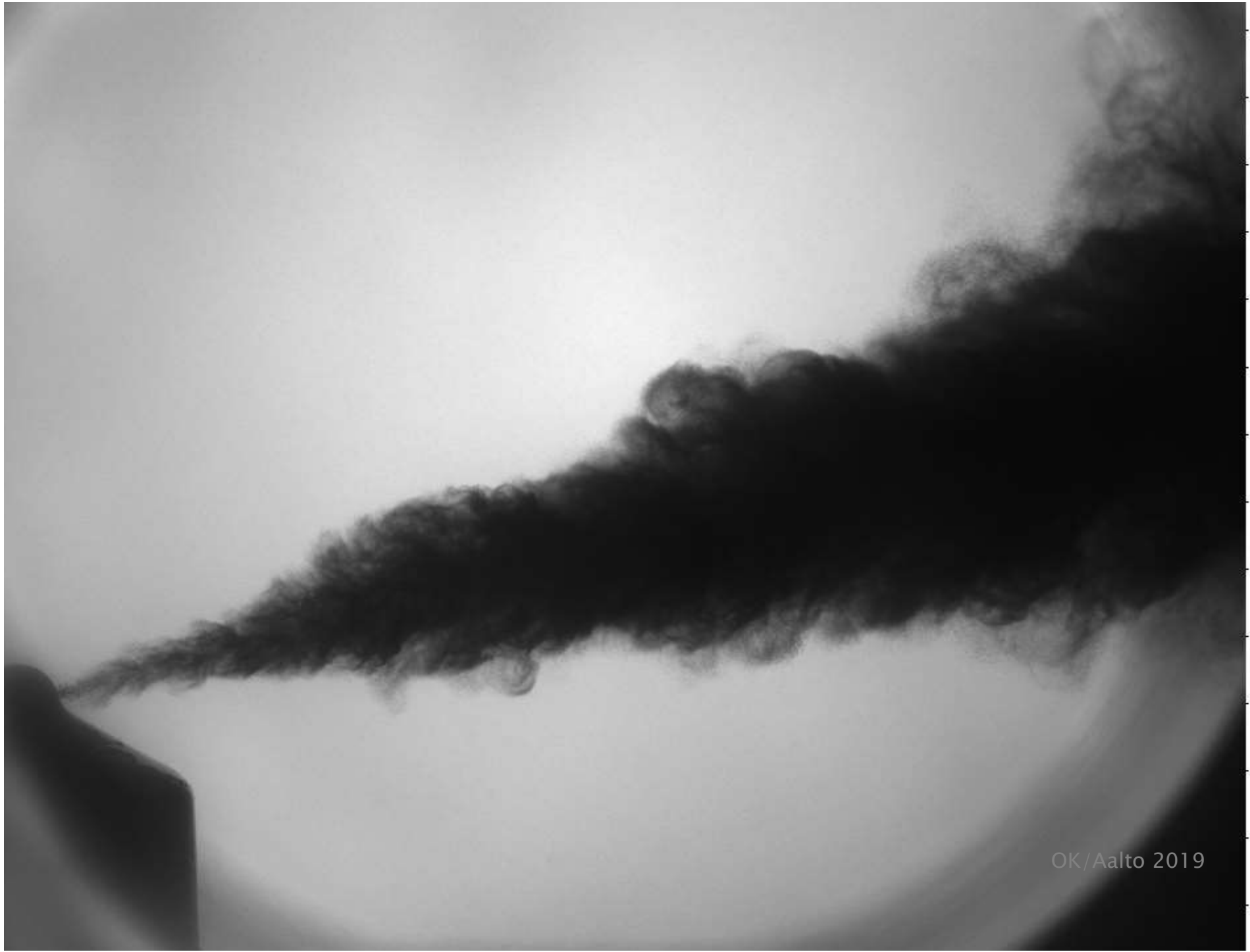
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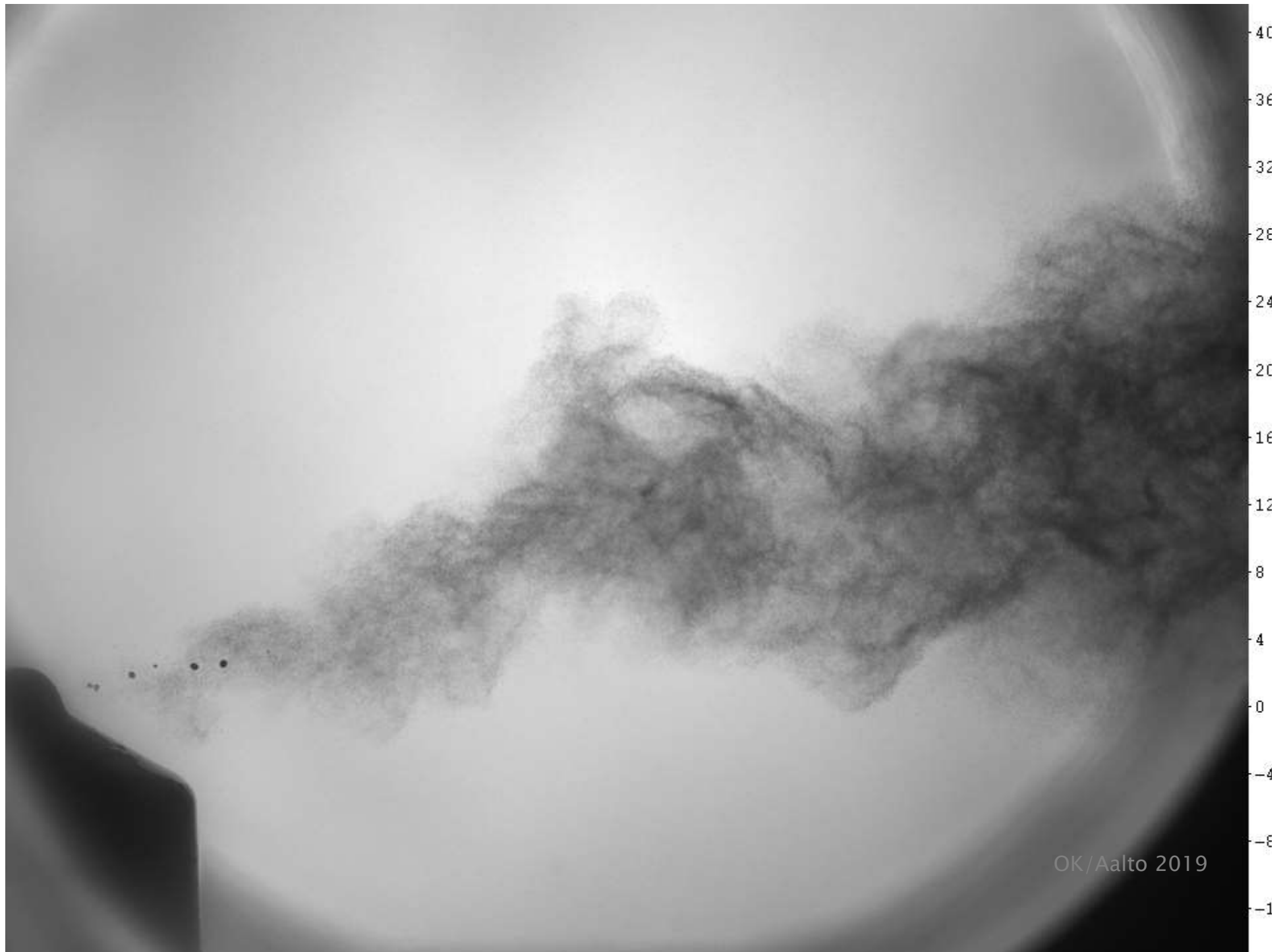
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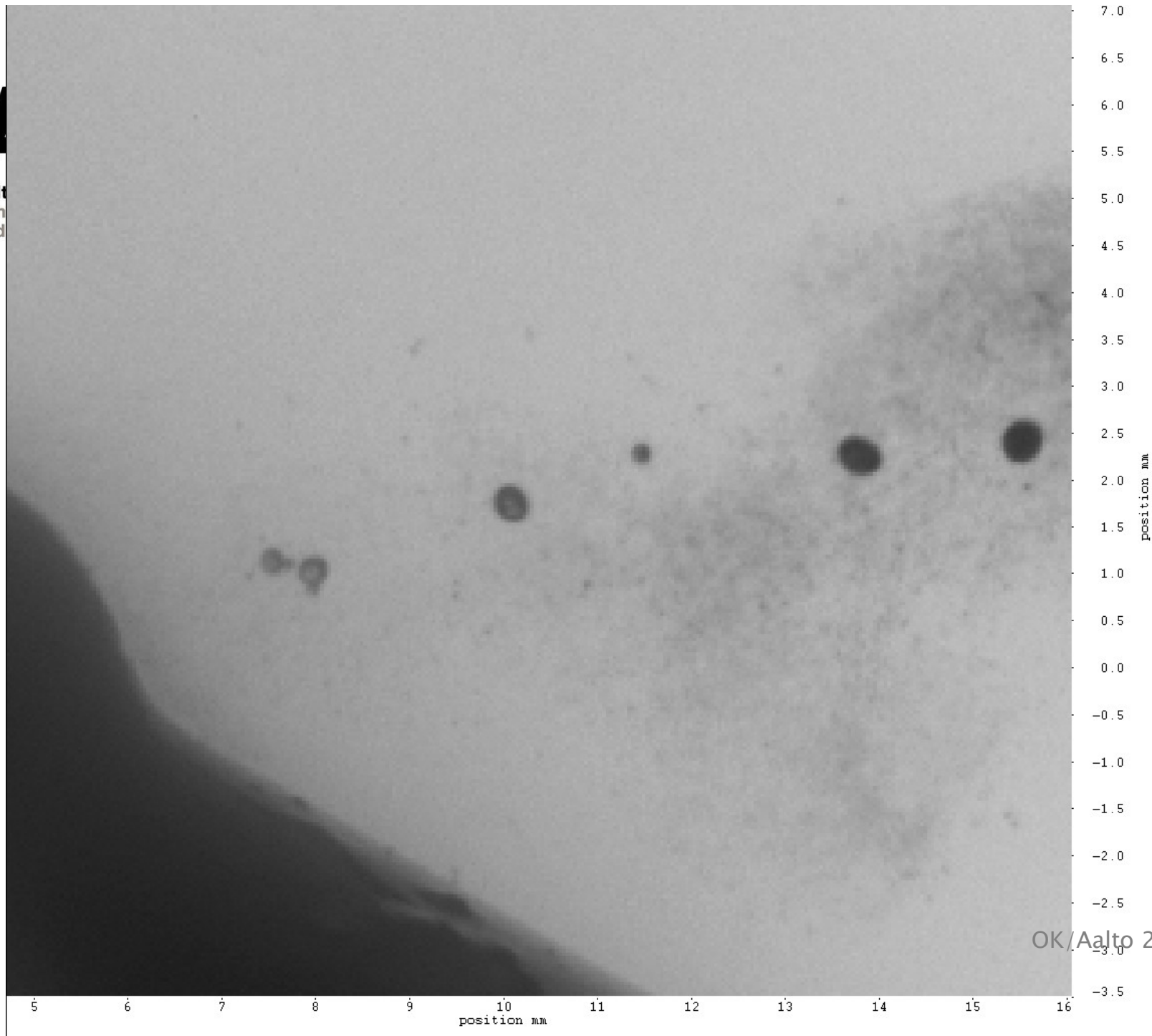
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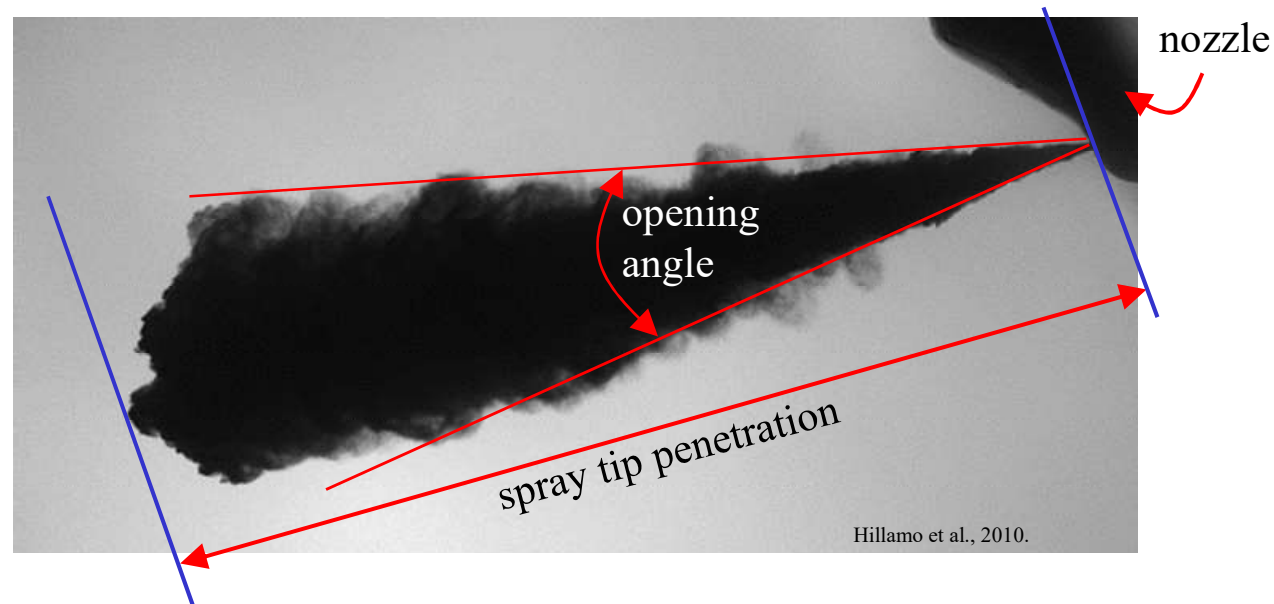


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Spray penetration & opening angle





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Spray penetration correlations

Dent (1971)

$$S = 3.07 \left(\frac{\Delta p}{\rho_g} \right)^{1/4} (tD)^{1/2} \left(\frac{294}{T_g} \right)^{1/4}$$

Hiroyasu & Kadota
(1980)

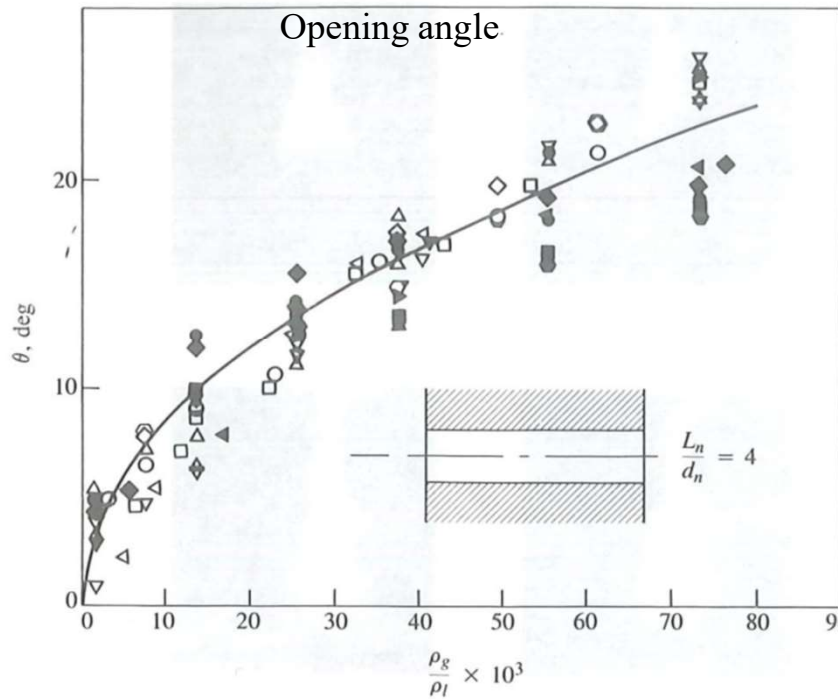
$$t < t_{break} : S = 0.39 \left(\frac{2\Delta p}{\rho_l} \right)^{1/2} t$$

$$t > t_{break} : S = 2.95 \left(\frac{\Delta p}{\rho_g} \right)^{1/4} (tD)^{1/2}$$

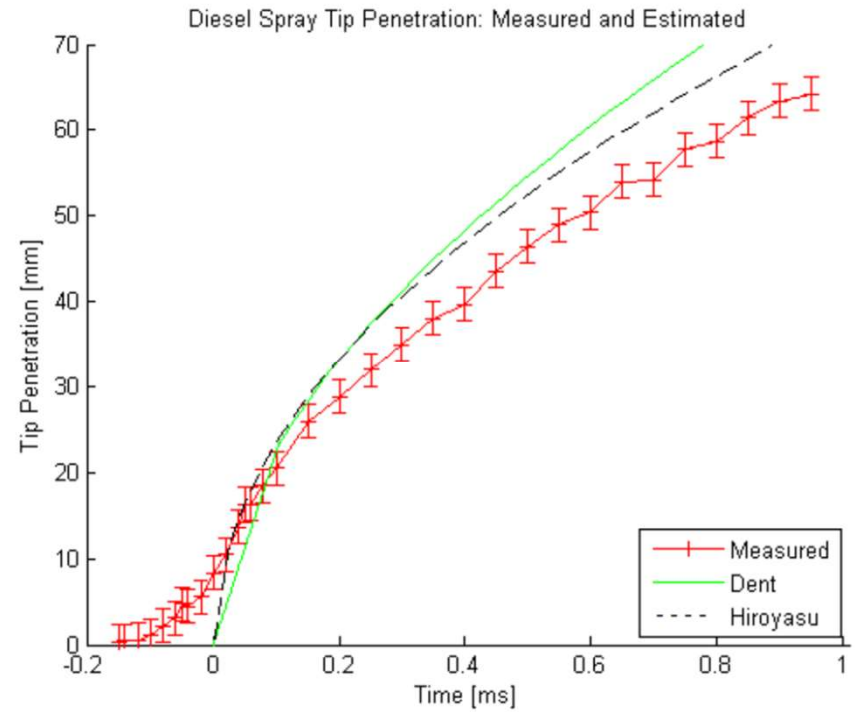
$$t_{break} = \frac{29\rho_l D}{(\rho_g \Delta p)^{1/2}}$$

ρ_g Gas density (kg/m³)
 ρ_l Liquid density (kg/m³)
 Δp Pressure difference (Pa)
 t Time (s)
 D Nozzle diameter (m)

Spray penetration & opening angle



Heywood, 1988

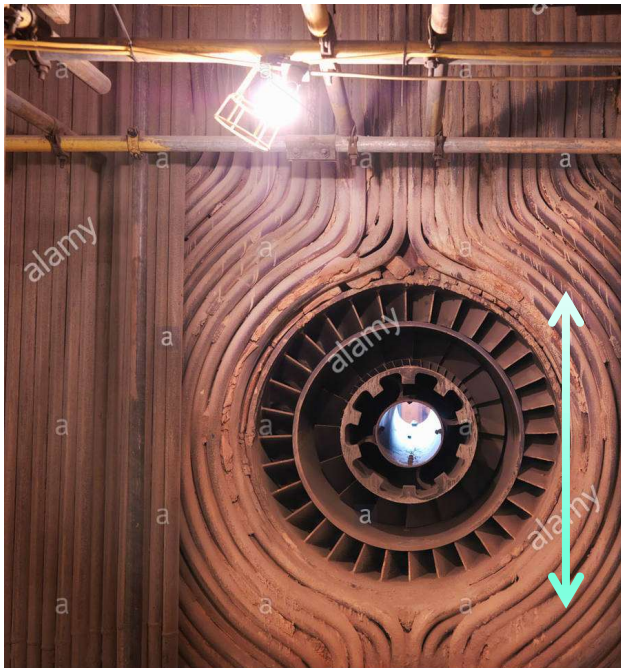


Hillamo et al., SAE 2008



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Nozzle types

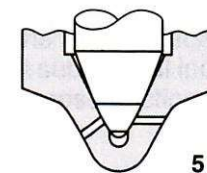
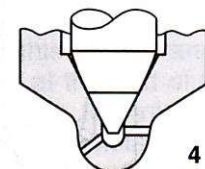
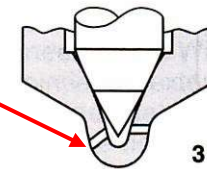
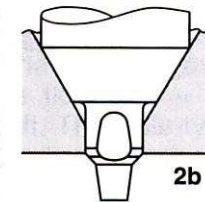
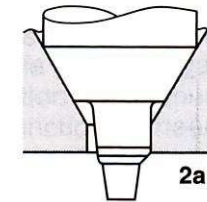
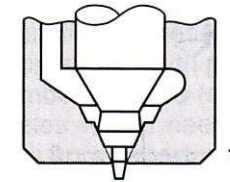


Furnaces:
diameter ~ 1m

Engines:
hole diameter
 $D \sim 0.1-0.5\text{mm}$

Nozzle shapes

- 1 Throttling-pintle nozzle,
- 2 Throttling-pintle nozzle with flat-cut pintle,
2a Side view, 2b Front view,
- 3 Hole-type nozzle with conical blind hole,
- 4 Hole-type nozzle with cylindrical blind hole,
- 5 Sac-less (vco) nozzle.



Some liquid fuel properties

Fuel	HVO	Fischer-Tropsch diesel	FAME (RME)	EN 590
Density at +15C° (kg/m ³)	775–785	770–785	885	835
Viscosity at +40C° (mm ² /s)	3.0–3.5	3.2–4.5	4.5	3.5
Cetane number	80–99	73–81	51	53
Distillation range (C°)	180–320	180–360	350–370	180–360
Heating value (MJ/kg)	44	43	37.5	43



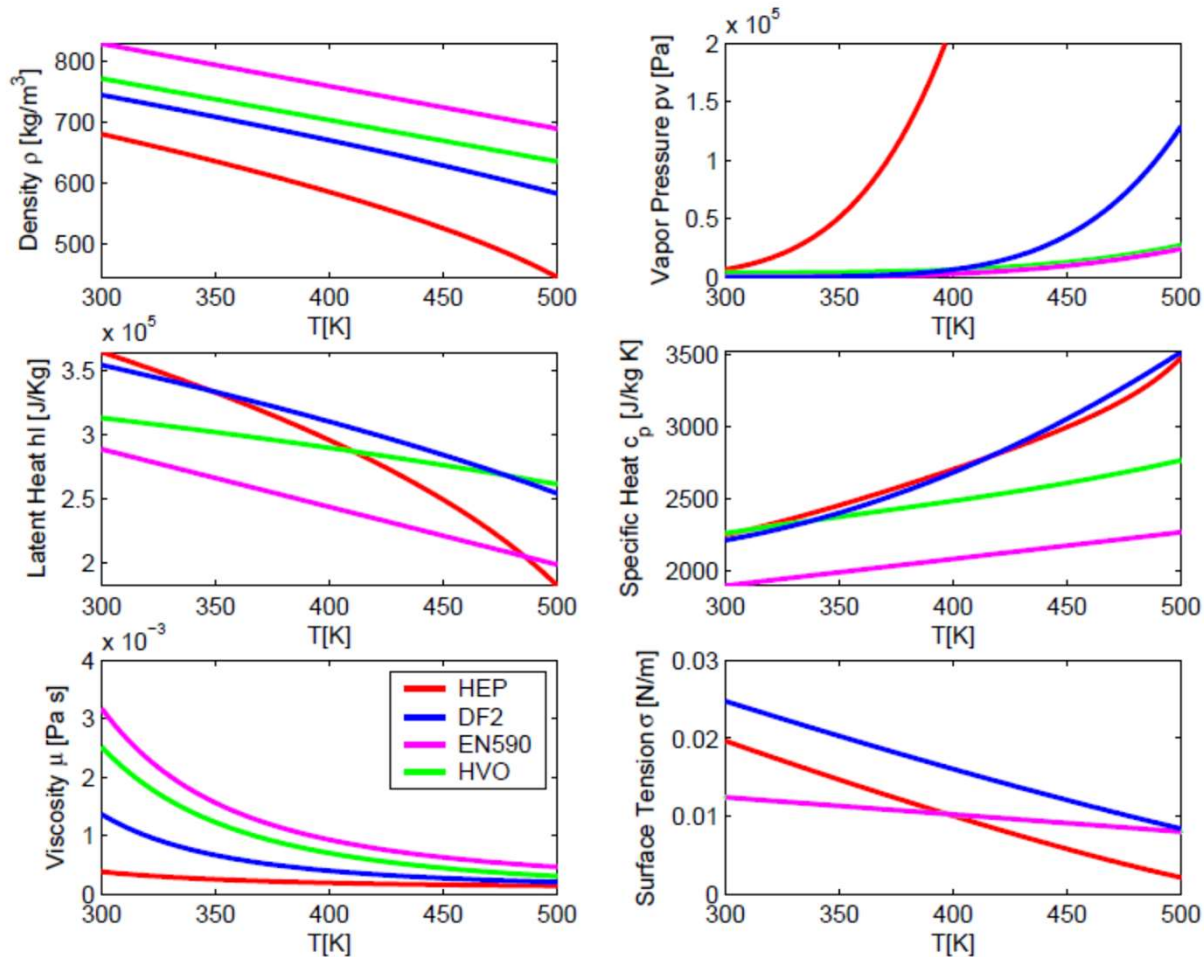
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Some liquid fuel properties

Fuel	Methanol	DME
Density at +15C° (kg/m ³)	722.1	612
Viscosity at +40C° (mm ² /s)	0.45	0.22
Cetane number	~ 5	55
Heating value (MJ/kg)	22.7	29



Some liquid fuel properties



Y. Gong et al., SAE 2010.

HEP = Heptane ($\text{C}_7 \text{H}_{16}$)
DF2= U.S. Diesel fuel
EN590=Standard Diesel in EU
HVO=Hydrotreated vegetable oil



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Topics on Spray Velocity



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Sprays are pressure driven

Cavitation

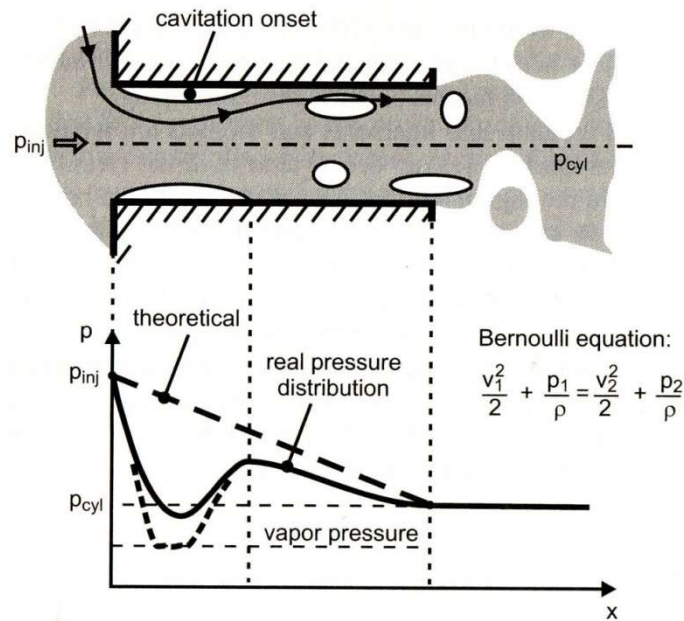


Fig. 5.6. Schematic illustration of cavitation formation inside the nozzle hole

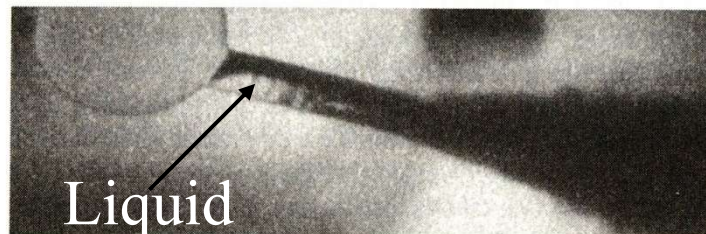
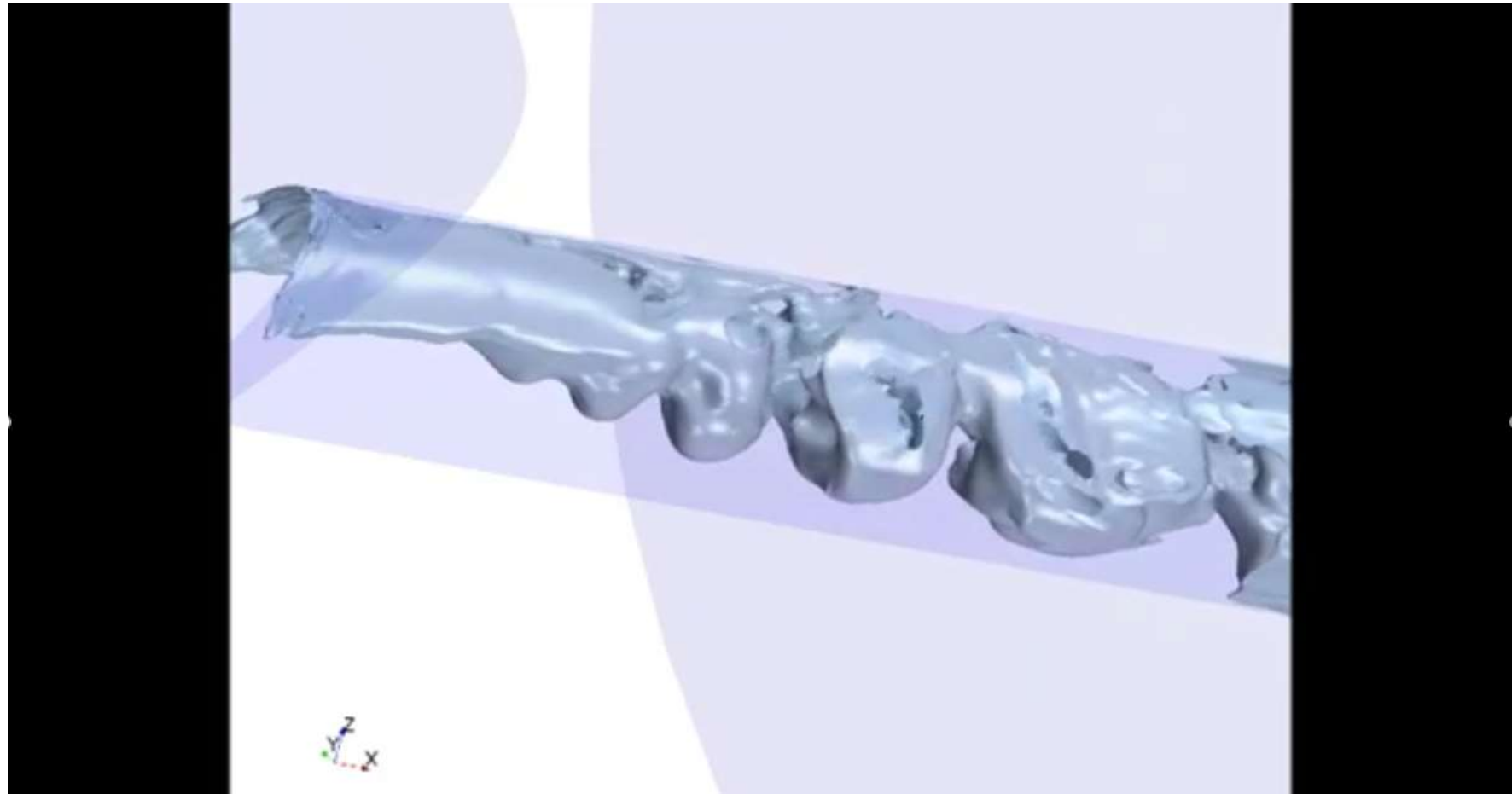


Fig. 5.7. Cavitation inside an acrylic glass diesel injection nozzle. The liquid phase is transparent, the gas phase is opaque. $p_{inj} = \text{MPa}$, $p_{cyl} = \text{MPa}$, $d_{noz} = \text{mm}$ [15]



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Cavitation





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Sprays are pressure driven

Calculating the theoretical fuel exit velocity

- **Theoretical fuel exit velocity from nozzle**

$$p_1 + \frac{1}{2} \rho u_1^2 = p_2 + \frac{1}{2} \rho u_2^2 \quad \longrightarrow \quad U_{Exit} = \sqrt{\frac{2 \Delta P}{\rho_f}}$$

- **Velocity in practice**

$$U_{Exit} = C_v \sqrt{\frac{2 \Delta P}{\rho_f}}$$

Discharge coefficient C_d

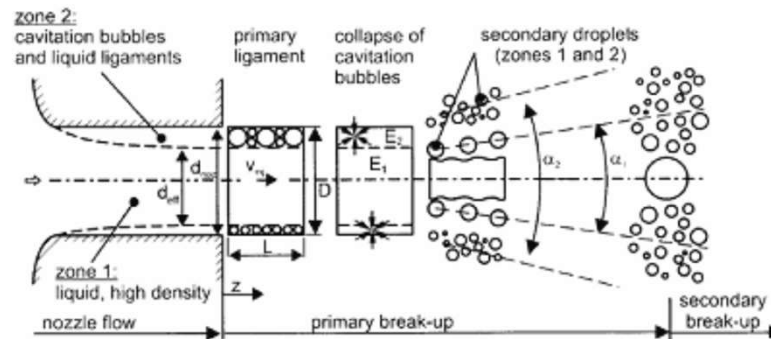
- Discharge coefficient**

$$C_d = C_v \cdot C_a$$

C_d used when calculating mass flow $\dot{m} = C_d A \rho_f \sqrt{\frac{2 \Delta P}{\rho_f}}$

C_v 'velocity coefficient', used when calculating real injection velocity

C_a 'area contraction coefficient', used when calculating effective nozzle hole diameter





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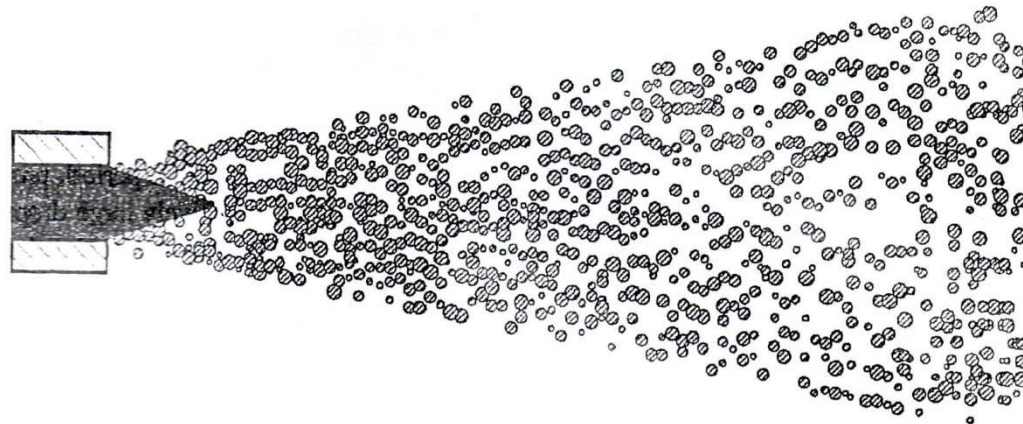
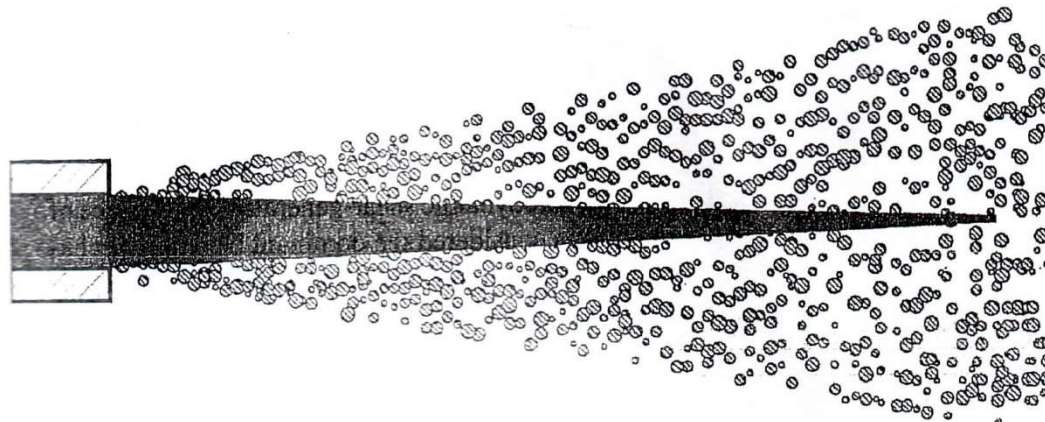
Additional material

Intact Liquid Core



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Additional material Intact liquid core ?



Smallwood, G., and Gulder, O., Atom & sprays, 2000.

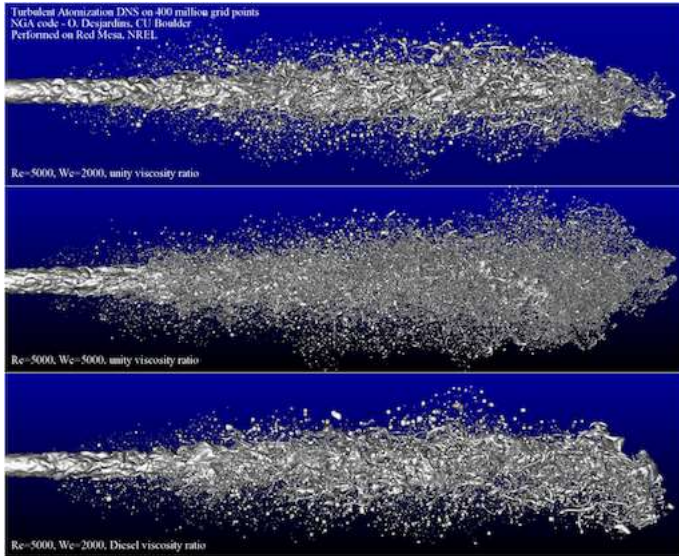
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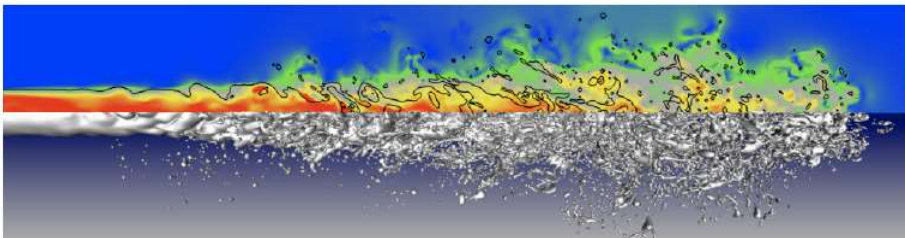
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Additional material

Intact liquid core ?

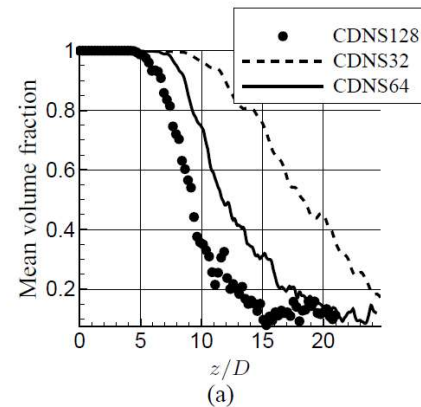
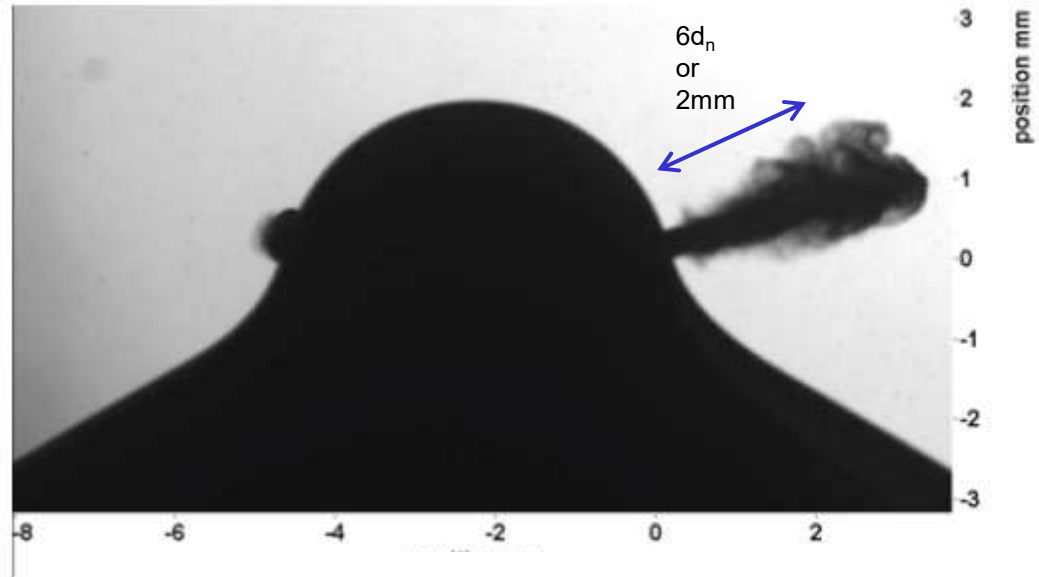


Desjardins et al, 2010



Reynolds number compared to real high-pressure sprays.
DNS done with $U_{exit} \sim 100\text{m/s}$

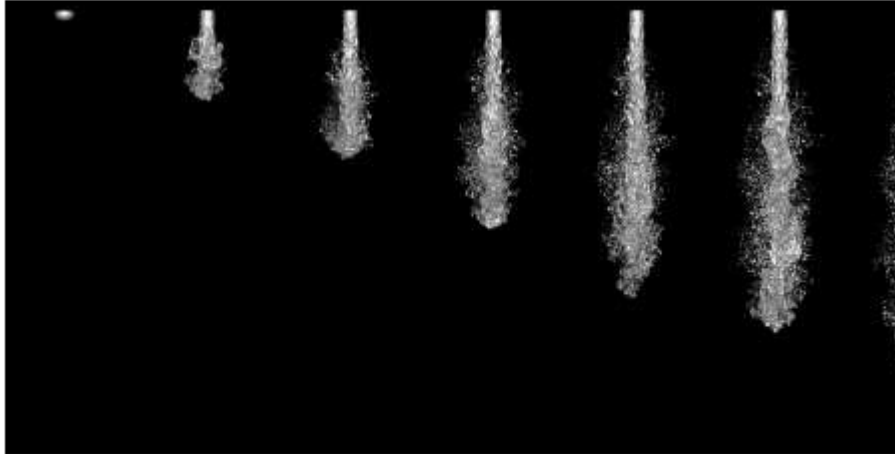
Hillamo et al. 2010, Atom. Sprays



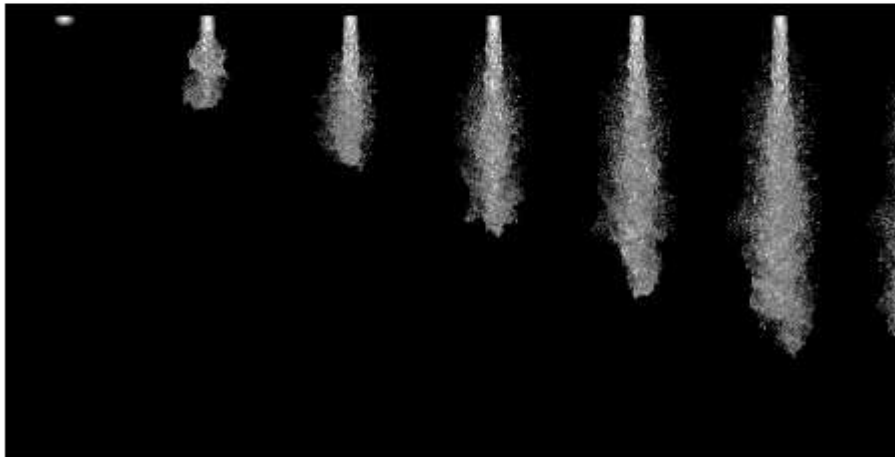
Chesnel et al. 2011

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Additional material Intact liquid core ?



(a) $Re = 5000$, $We = 2000$



(b) $Re = 5000$, $We = 5000$

- **Turbulence inside the nozzle has bigger effect than previously thought of.**
- **Taken the experimental and numerical evidence, it seems very likely that in modern high-pressure injection systems, the intact liquid core is either very short or even non-existent.**



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Spray Equations



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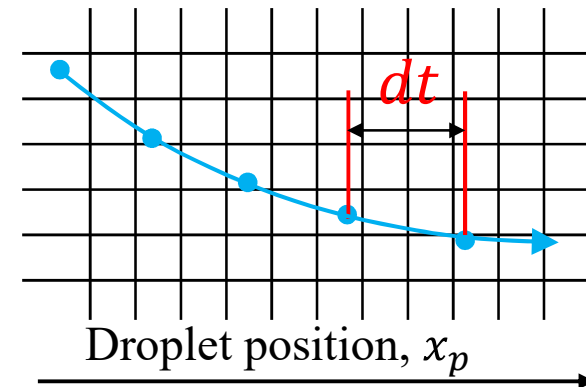
Lagrangian spray modeling

- **Discrete computational points are tracked**
- **Spray equations (Ordinary differential equation (ODE)):**

$$\frac{dx_p}{dt} = u_p$$

- **Parcel position**

$$\rightarrow x_{p,2} = x_{p,1} + u_p dt$$



(1)

- **Eq. (1) can be used to calculate the particle position at each time instant.**



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Lagrangian spray modeling

- **Parcel equation of motion** $F = ma$

$$ma = F_{drag} = \frac{1}{2} \rho_g C_d A U^2$$

$$\frac{1}{6} \rho_p \pi d^3 \frac{du_p}{dt} = \frac{1}{2} (u_g - u_p) |u_g - u_p| \rho_g C_d \frac{\pi d^2}{4}$$

$$\rightarrow \frac{du_p}{dt} = \frac{C_d Re_p}{\tau_p 24} (u_g - u_p)$$

$$\tau_p = \frac{\rho_l d^2}{18 \rho_g \nu_g}$$

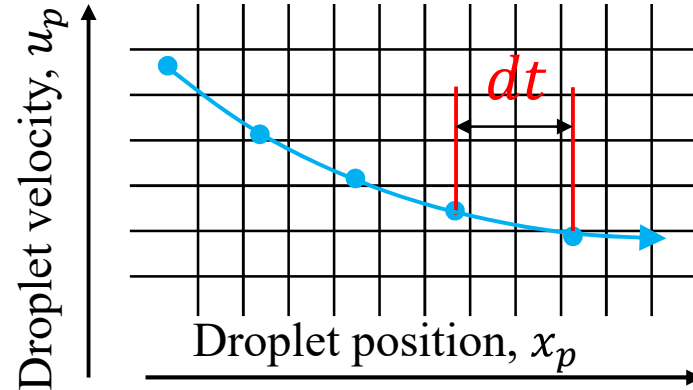


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Lagrangian spray modeling

or

$$u_{p,2} = u_{p,1} + \frac{C_d Re_p}{\tau_p 24} (u_g - u_{p,1}) dt \quad (2)$$

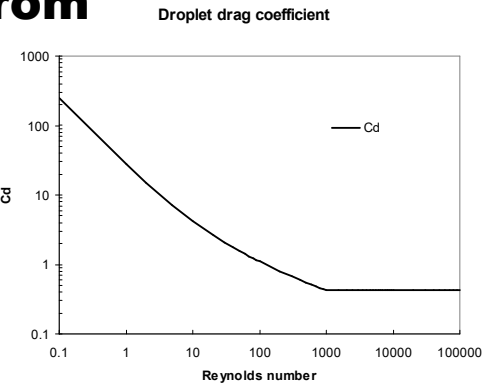
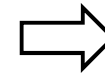


Eq. (2) can be used to update the particle velocity for each time instant, and then use Eq. (1) to update the position.

Droplet Drag

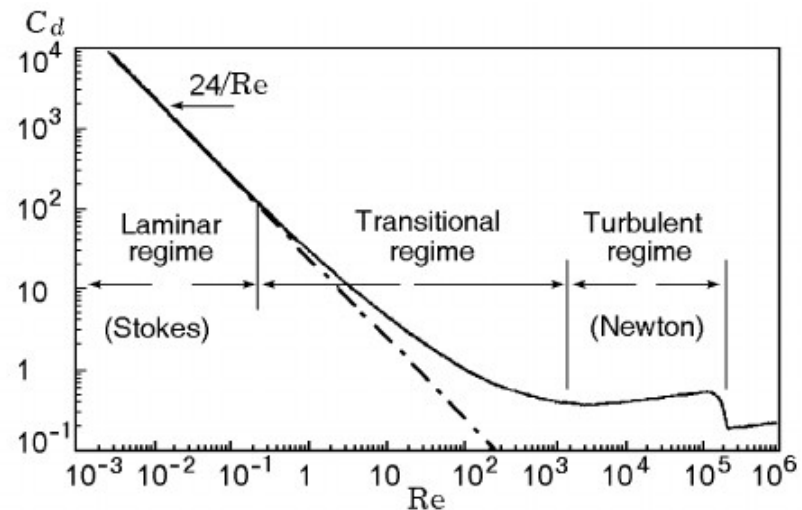
- **The resistance a droplet/particle encounters, is due to shear and form drag**
- **Droplet drag coefficient C_d is calculated from**

$$C_D = \begin{cases} \frac{24}{Re_p} \left(1 + \frac{1}{6} Re_p^{2/3} \right) & Re_p < 1000 \\ 0.424 & Re_p \geq 1000 \end{cases}$$

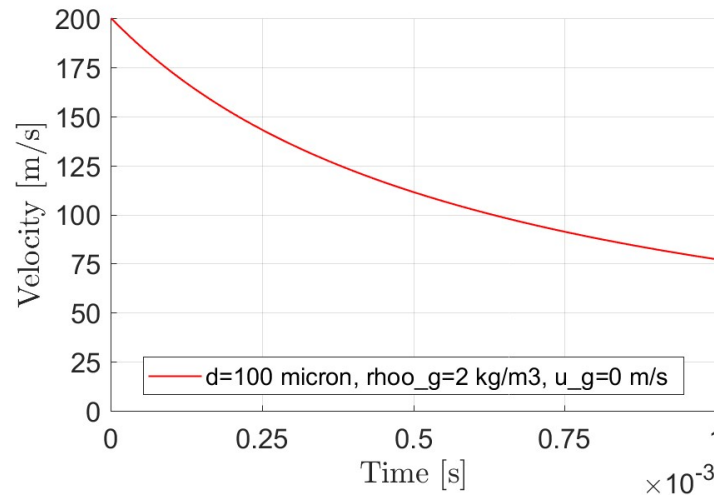


- **Droplet Reynolds number is**

$$Re_p = \frac{\rho_g |u_g - u_p| d}{\mu_g} = \frac{|u_g - u_p| d}{\nu_g}$$



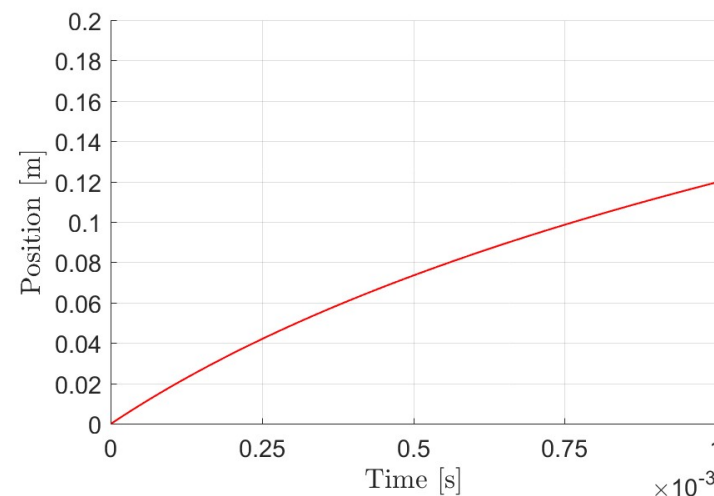
Lagrangian spray modeling (Matlab)



Constant Cd=0.424 and constant gas phase velocity (0 m/s)

Droplet size is constant d=100micro-m

Gas density is 2 kg/m³ and liquid density is 800kg/m³.



Constant gas phase velocity may be a strong assumption.



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Dispersed and continuous phase coupling



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Stokes number

- **Stokes number** $St = \frac{\tau_p}{\tau_f}$
- **Ratio of the characteristic particle time scale (or the momentum relaxation time scale)**

$$\tau_p = \frac{\rho_l d^2}{18 \rho_g v_g}$$

- **To the characteristic flow time scale**

$$\tau_f = \frac{D}{U_0}$$

d droplet diameter

D nozzle diameter

U_0 flow velocity

Stokes number & Turbulent dispersion

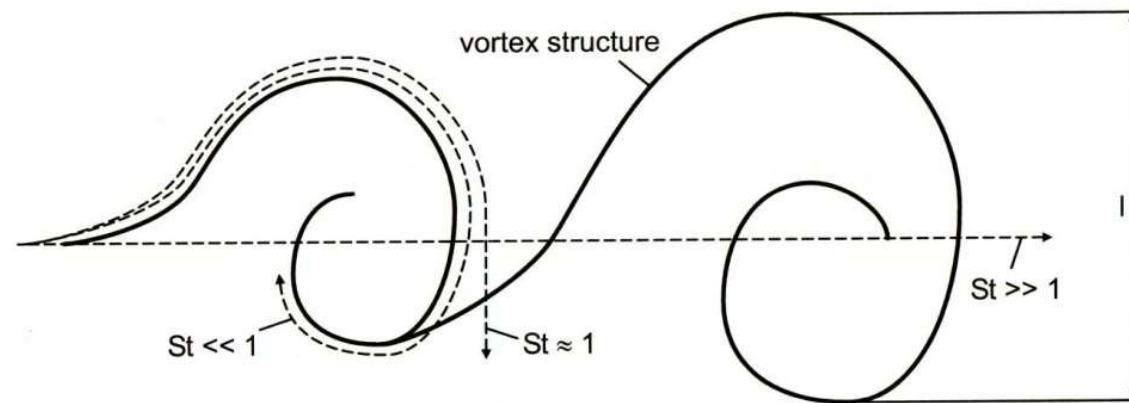
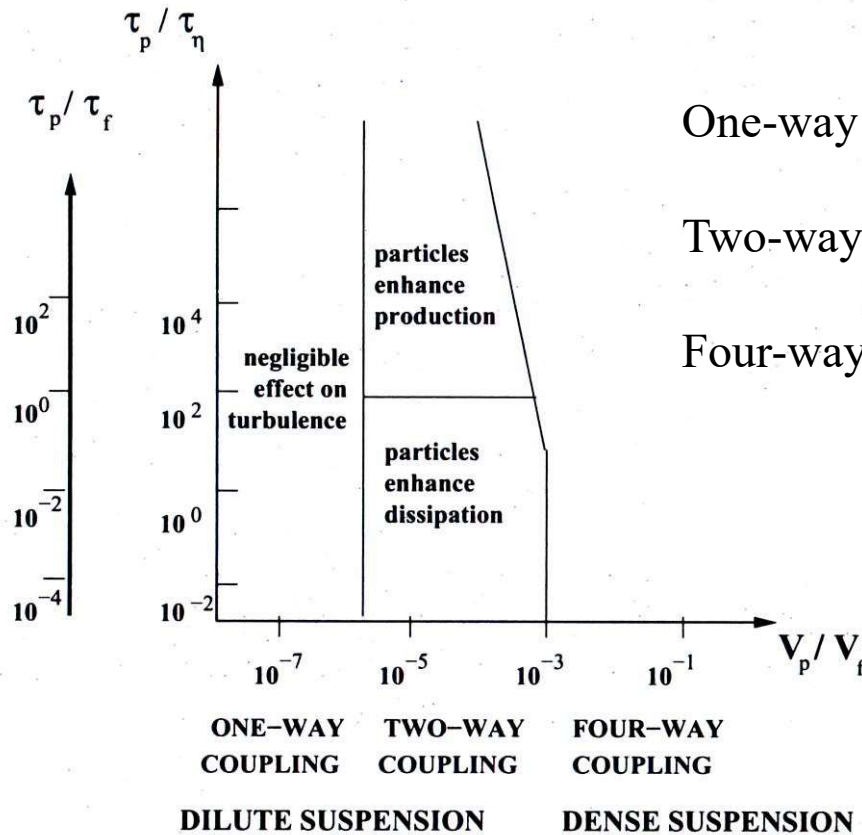


Figure by G. Stiesch

Coupling regimes



One-way coupling → Gas phase only interacts with particles

Two-way coupling → Gas phase and particles both interact with each other

Four-way coupling → Additionally, particles interact with each other

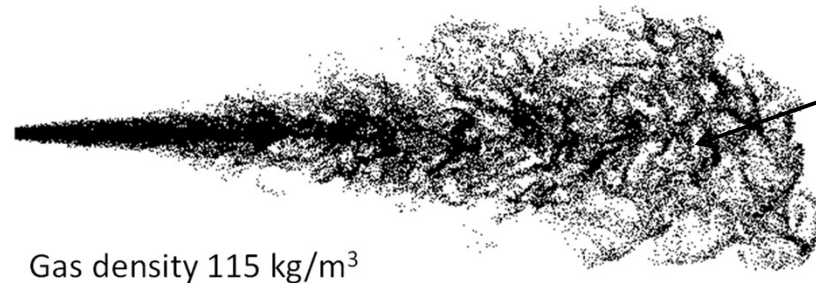
Figure by Elghobashi, 1994.



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Stokes number

- **If $St \ll 1$ Particles will have plenty of time to adjust to changes in flow field velocity. Particle and fluid velocities nearly equal.**
- **If $St \gg 1$ Particle velocity will be little affected by flow velocity.**



St number

effect $St \sim \frac{1}{\rho g}$

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Source term in N-S Eqs.

- **The flow field 'sees' the droplets via source terms in the N-S eqs. In the 2-way coupling regime, the source term M_d accounts for the coupling. Droplets 'see' the flow field effect from the velocity difference.**

$$\frac{\partial \rho}{\partial t} + \frac{\partial \rho u_j}{\partial x_j} = 0$$

$$\frac{\partial \rho u_i}{\partial t} + \frac{\partial \rho u_i u_j}{\partial x_j} = -\frac{\partial}{\partial x_j} (p \delta_{ij} - \tau_{ij}) + \underline{M_d}$$

$$\underline{M_d} = \frac{1}{2} \rho_g C_d A |u_g - u_p| (u_g - u_p)$$



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Group Work

Discuss in pairs for about 10 min:

- 1. What information is needed to calculate droplet/particle Stokes number ?**
- 2. What information is not needed ?**
- 3. What does a particle St number mean ?**
- 4. If you would be given a task to design a spray configuration that would give a homogeneous mixture (of droplets and air / vapor and air), what kind of a St number would you aim for and why ?**



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What is dense ?



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What is dense ?

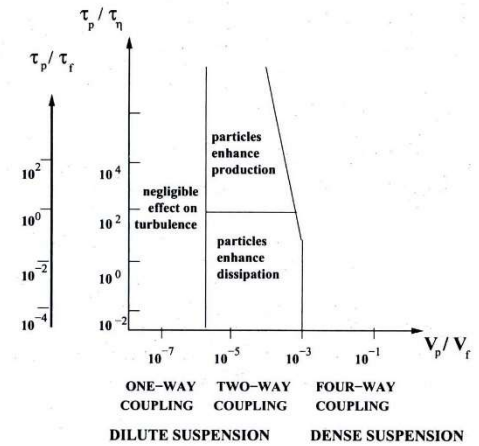
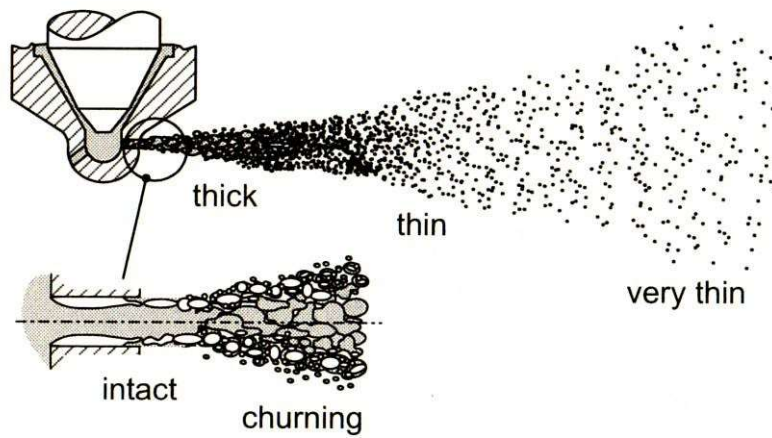
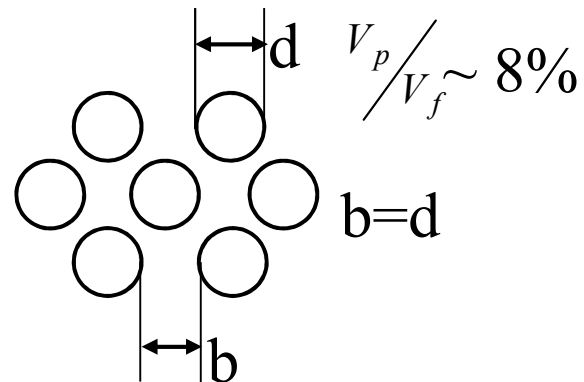
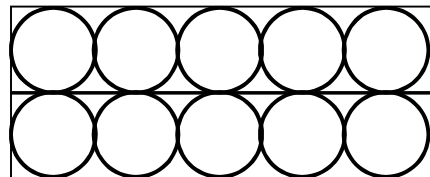


Fig. 5.2. Schematic illustration of different flow regimes

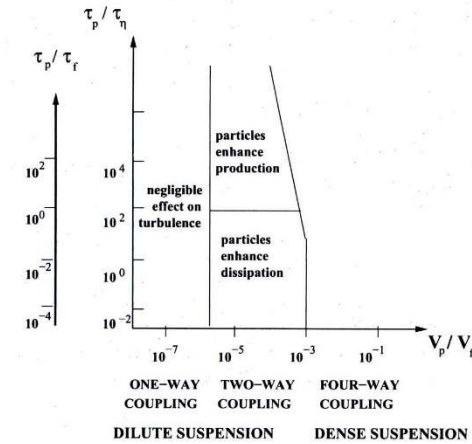
$$\frac{V_p}{V_f} \sim 52\%$$





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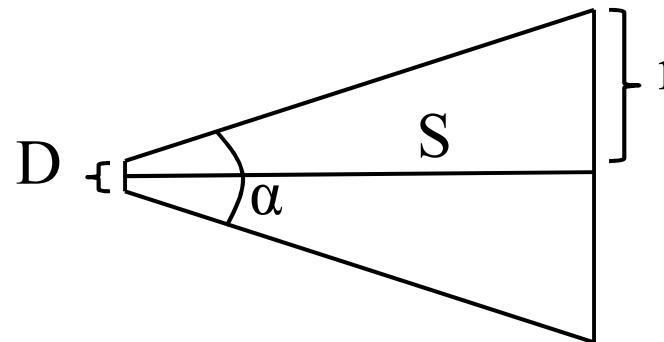
What is dense ?



- Typically assumed dense if $V_p/V_f > 10^{-3}$
- Assuming spray mass is divided homogenously in a cone-like volume
- From the schematic spray opening angle picture, the area ratio:

$$\left(\frac{D/2}{r}\right)^2 > \frac{1}{1000}$$

$$\left(\frac{D/2}{\tan(\alpha/2)S}\right)^2 > \frac{1}{1000}$$



$$\tan \frac{\alpha}{2} = \frac{r}{S}$$

$$r = \tan \frac{\alpha}{2} S$$

$S < 90D$ assuming $\alpha = 20^\circ$

If $D=90\mu m$, $S < 8.1\text{mm}$

D nozzle diameter

S spray penetration


α spray opening angle



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Additional material

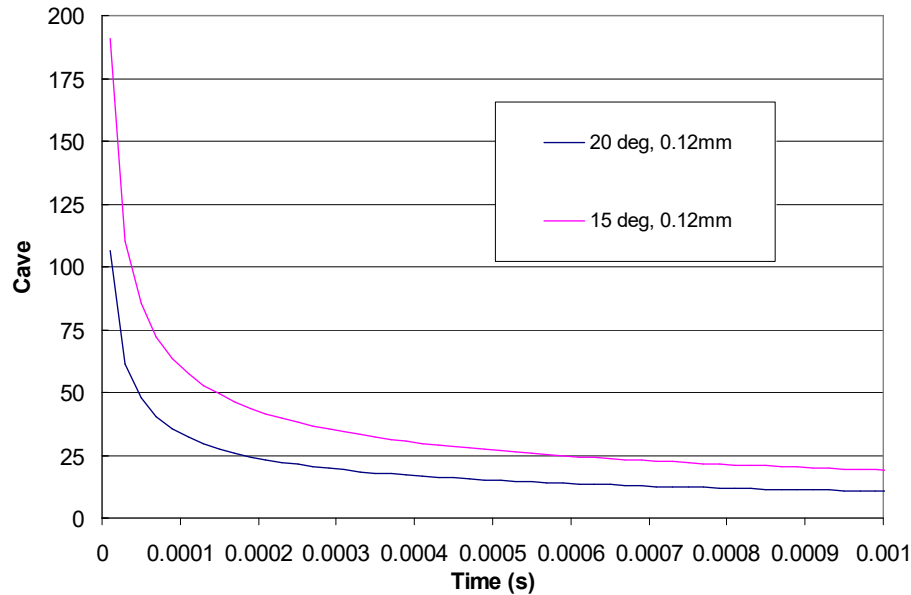
Nozzle Hole Size vs Average Fuel Concentration

- **Average concentration** $c_{ave} = \frac{m_s}{V_s}$
- **Assumed constant injection rate** $m_s \sim t D^2$
- **The spray volume evolves as a cone** $V = \frac{1}{3} A S$
- **Bottom area of the cone** $A(t) \sim \left(\tan\left(\frac{\alpha}{2}\right) S \right)^2$
- **Spray penetration assumes well known correlation** $S(t) \sim (t D)^{1/2}$
- **We obtain**
$$c_{ave} \sim \frac{t D^2}{\frac{1}{3} \pi \left(\tan\left(\frac{\alpha}{2}\right) S \right)^2 S} \sim \left(\frac{D}{t} \right)^{\frac{1}{2}}$$


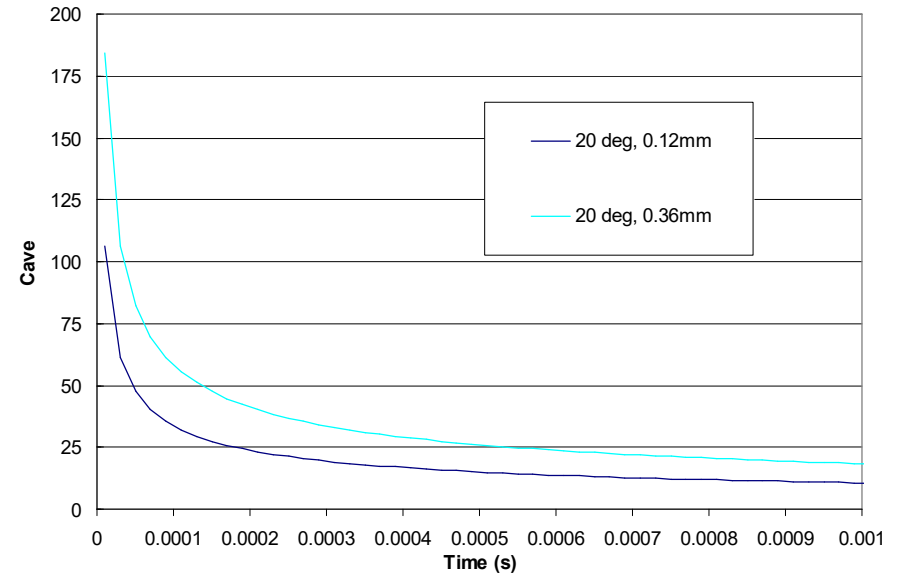
Additional material

Nozzle Hole Size vs Average Fuel Concentration

Calculated result from the previous slide final result



D=0.12mm, $\alpha=15^\circ$ or 20°



D=0.12mm or 0.36mm, $\alpha=20^\circ$

Turbulence in flow

- **Turbulent flow consists of vortices (eddies) of different size**
- **The biggest vortices are of the size of the flow geometry**
- **The eddies are breaking up into smaller eddies. At the same time kinetic energy is transported from the bigger eddies to smaller ones**
- **At the smallest turbulent scale (Kolmogorov scale) kinetic energy is dissipated into heat**
- **Fluctuations**
- **Convection**

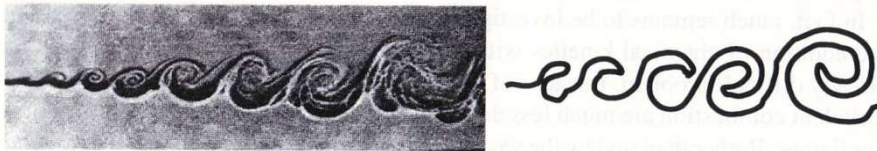
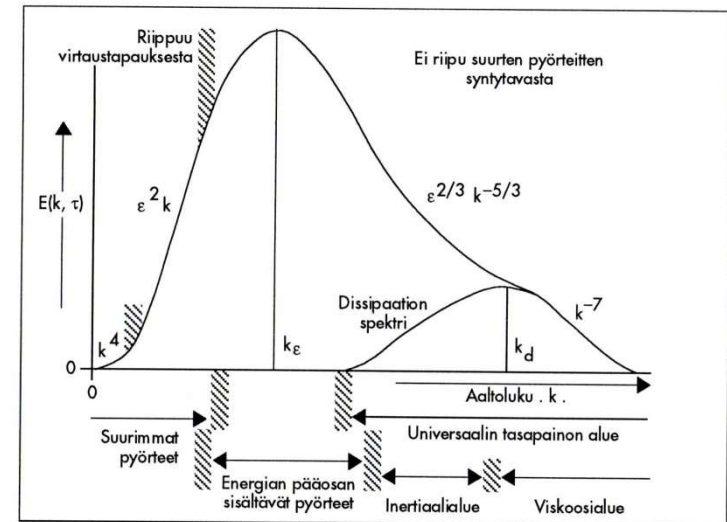


Fig. 12.1. Formation of a turbulent shear layer (Roshko 1975)



Kuva 12.4 Turbulenssin kineettisen energian jakauma aaltoluvun k funktiona. Aaltoluku on pyörteen halkaisijan käänteisluku.

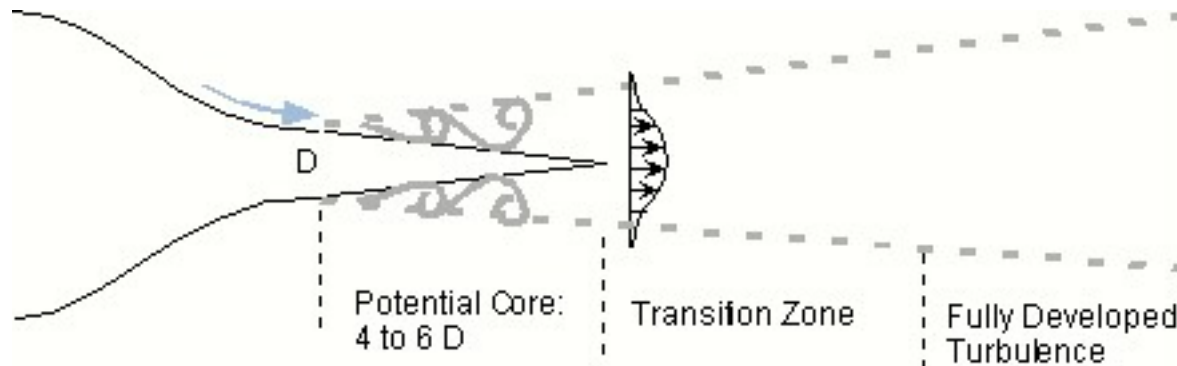


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Turbulence in flow

- **Turbulence \leftrightarrow Shear**
- **Turbulence is characterized by the following:**
 - **Three dimensional**
 - **Unsteady**
 - **Random**
 - **Strong vorticity**
 - **Dissipative**
 - **Strongly diffusive**

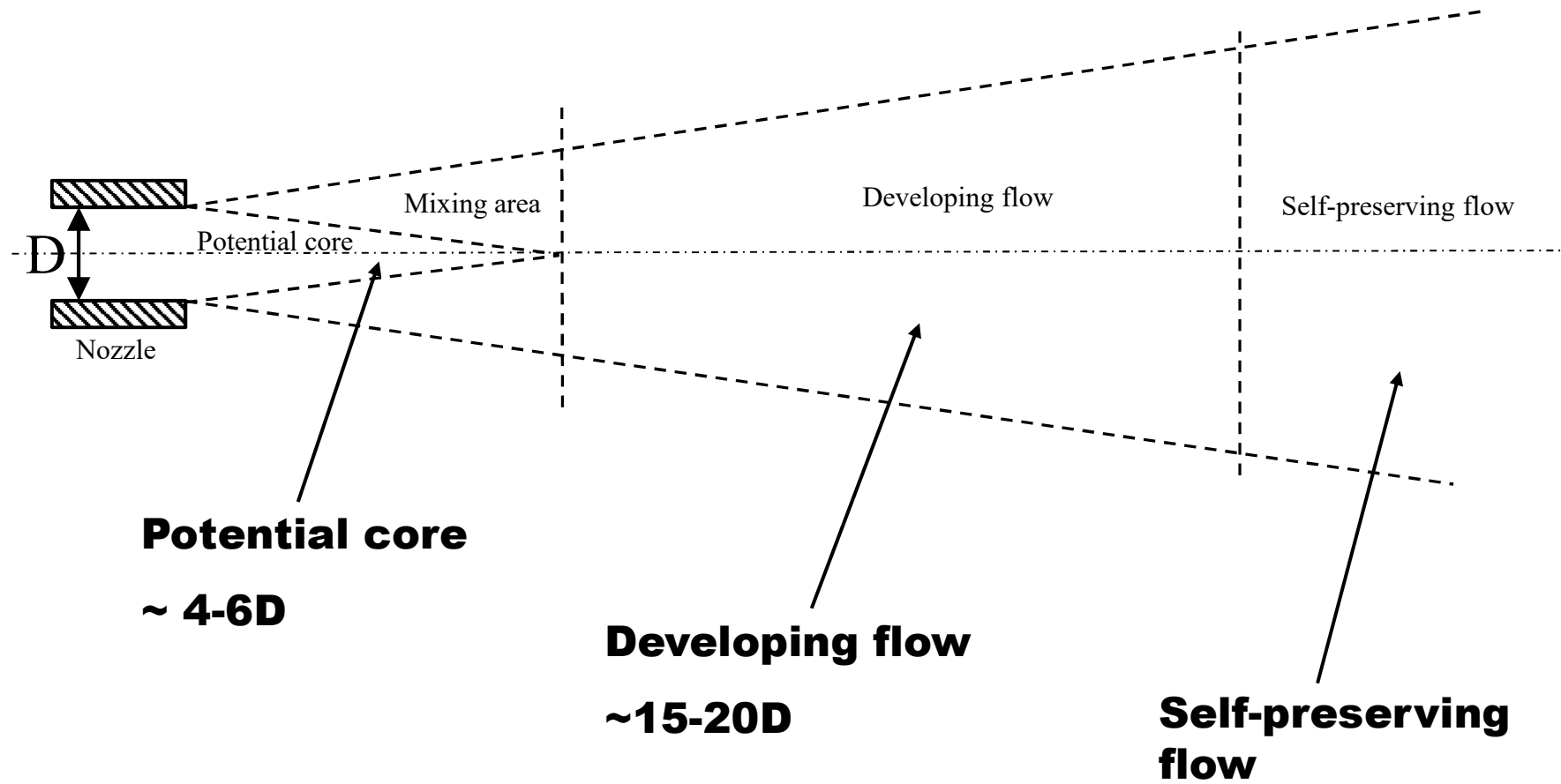
Turbulence = increased mixing !





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Jet development



Jet development: Kelvin Helmholtz Instability and Spray Mixing



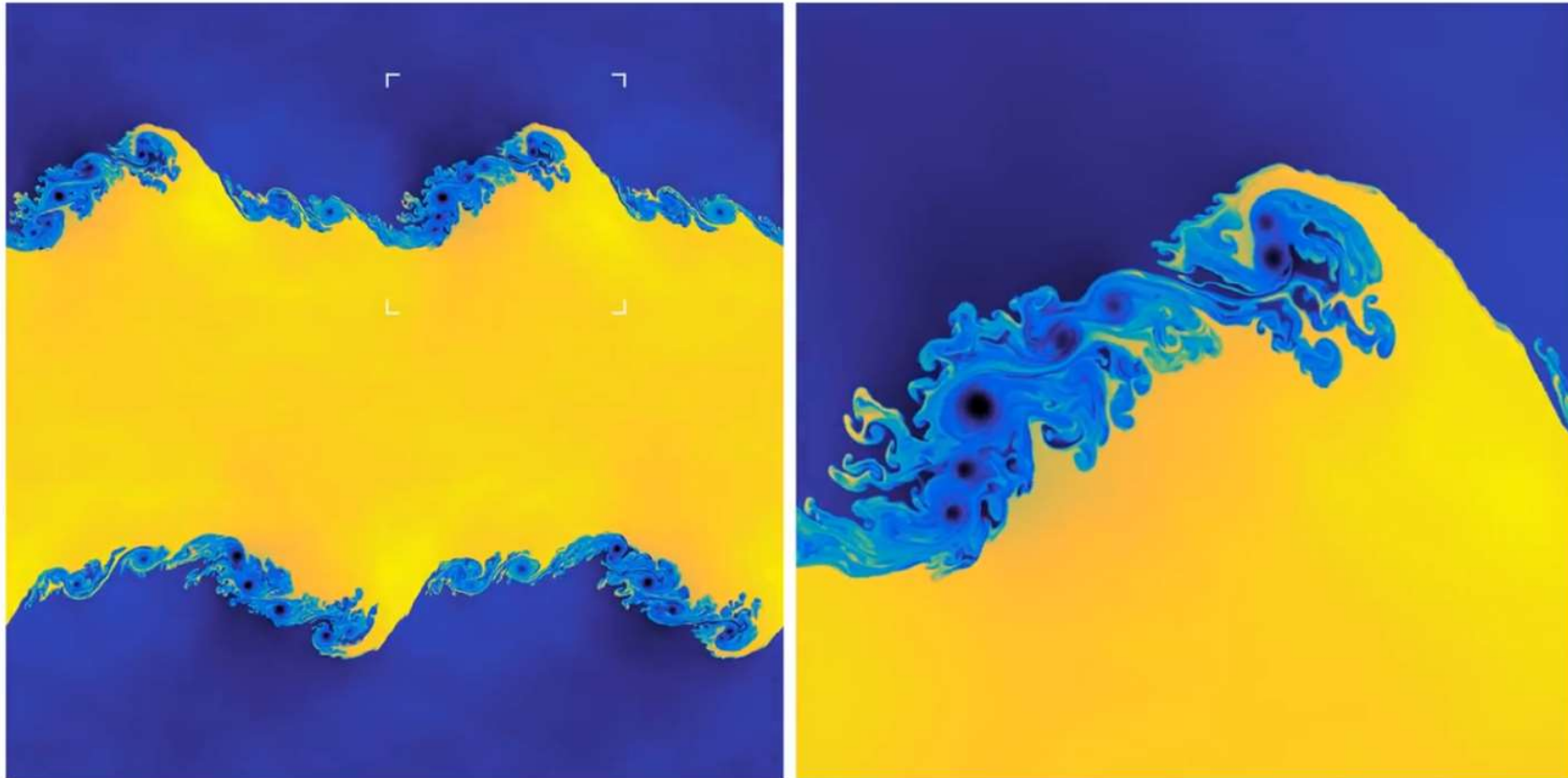
- **Spray and jet turbulence is initiated by a Kelvin Helmholtz instability at the near nozzle region**
- **Growth rate of KH instability is related to the velocity difference of the fluids and to the wavelength.**





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KH Instability



KH Instability and Mixing

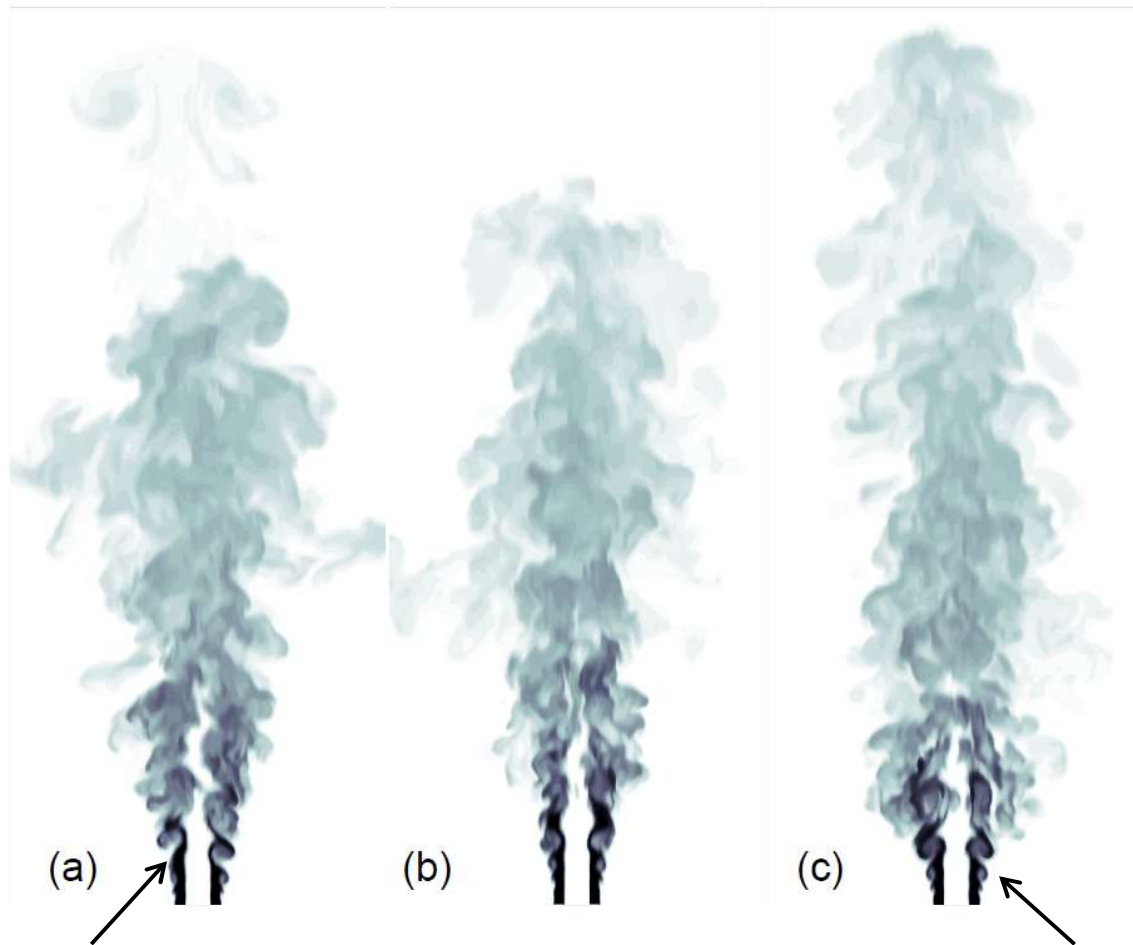


Figure V. Vuorinen



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Additional Reading

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